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**SIMULATION OF TRIPLE-SPOOL TURBOFAN  
ENGINE**

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**Air Force Aero Propulsion Laboratory  
Wright-Patterson Air Force Base, Ohio**

**April 1974**

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report describes a digital computer program entitled TRISPL. TRISPL is a computer program that simulates steady-state design and off-design performance of triple-spool turbofan engines. The program has been formulated for an engine type with two core spools and one fan spool but can easily be modified for other engine types (two fan spools and one core spool, for example). The program, written in Fortran IV language, uses performance maps (in Block Data format)		

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of the major engine components. Information on setting up the Block Data and input data is given in the report. Also included is a complete program listing with a description of each subroutine and sample results.

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## FOREWORD

This report was prepared in the Performance Branch (AFAPL/TBA), Turbine Engine Division, Air Force Aero Propulsion Laboratory, Wright-Patterson Air Force Base, Ohio, under Project 668A0215, with Edward K. Navaisis as Project Engineer.

This report covers work conducted within the Performance Branch in the time period between April 1973 and January 1974.

## ABSTRACT

This report describes a digital computer program entitled TRISPL. TRISPL is a computer program that simulates steady-state design and off-design performance of triple-spool turbofan engines. The program has been formulated for an engine type with two core spools and one fan spool but can easily be modified for other engine types (two fan spools and one core spool, for example). The program, written in Fortran IV language, uses performance maps (in Block Data format) of the major engine components. Information on setting up the Block Data and input data is given in the report. Also included is a complete program listing with a description of each subroutine and sample results.

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## SYMBOLS

### STATION NUMBERS

1	ambient
2	fan entrance
21	fan exit/intermediate compressor and duct entrance
22	intermediate compressor exit/high compressor entrance
3	high compressor exit/burner entrance
4	burner exit/hi pressure turbine entrance
45	hi press. turbine exit/int. press. turbine entrance
5	int. turbine exit/low press. turbine entrance
55	low press. turbine exit
6	afterburner entrance
7	afterburner exit
8	main nozzle throat
9	main nozzle exit
23	duct burner entrance
24	duct burner exit
25	duct exit if mixed flow engine
28	duct nozzle throat
29	duct nozzle exit

### OTHER THERMODYNAMIC PROPERTIES

T	total temperature (degrees R)
P	total pressure (atm)
H	total enthalpy (Btu/lbm)
S	total entropy (Btu/lbm °R)

TS	static temperature (degrees R)
PS	static pressure (atm)
FAR	fuel-air ratio
AM	Mach number
V	velocity (ft/sec)

#### COMPONENT SYMBOLS

F	fan
I	intermediate (middle) compressor
C	high pressure compressor
B, COM	combustor
D, DUC	fan duct
THP	high pressure turbine
TIP	intermediate turbine
TLP	low pressure turbine
A, AFT	afterburner
NOZ	nozzle
M	main nozzle
OB	overboard
T	total

#### ENGINE SYMBOLS

ETAR	ram recovery, $P_2/P_1$
CN	ratio of corrected speed to design corrected speed
PCN	percent of design shaft speed
PR	pressure ratio

In addition to all input symbols, the following are output:

A	areas (ft <sup>2</sup> )
AM	Mach numbers
BLDU	total bleed flow into duct (lbm/sec)
BLDUC	bleed flow into duct from high-pressure compressor (lbm/sec)
BLDUI	bleed flow into duct from intermediate compressor (lbm/sec)
BLC	high-pressure compressor bleed flow (lbm/sec)
BLF	fan bleed flow, overboard (lbm/sec)
BLHP	total bleed flow to high-pressure turbine (lbm/sec)
BLHPC	high-pressure turbine cooling bleed from high-pressure compressor (lbm/sec)
BLHPI	high-pressure turbine cooling bleed from intermediate compressor (lbm/sec)
BLI	intermediate compressor bleed flow (lbm/sec)
BLIP	total bleed flow to intermediate turbine (lbm/sec)
BLIPC	intermediate turbine cooling bleed from high-pressure compressor (lbm/sec)
BLIPI	intermediate turbine cooling bleed from intermediate compressor (lbm/sec)
BLLP	total bleed flow to low-pressure turbine (lbm/sec)
BLLPC	low-pressure turbine cooling bleed from high-pressure compressor (lbm/sec)
BLLPI	low-pressure turbine cooling bleed from intermediate compressor (lbm/sec)
BLOB	total overboard bleed flow (lbm/sec)
BLOBC	overboard bleed flow from high-pressure compressor (lbm/sec)
BLOBI	overboard bleed flow from intermediate compressor (lbm/sec)

FGP	pressure thrust (lbf)
FN	net thrust (lbf)
FRD	ram drag (lbf)
GU	initial or guessed values
ITRYS	number of loops thru engine before quitting
LOOPER	number of loops thru engine
SFC	specific fuel consumption (lbm/lbf/hr)
TOLALL	tolerance on convergence
VA	flight speed (ft/sec)
VJ	jet velocity (ft/sec)

#### INPUT SYMBOLS

ALTP	altitude (ft)
AM	flight Mach number
AM55	Mach number at low pressure turbine exit
AM6	afterburner entrance Mach number at design
A8	main nozzle throat area - can be changed at off design - (ft <sup>2</sup> )
A28	duct nozzle throat area - can be changed at off design - (ft <sup>2</sup> )
CNHPDS	design corrected speed - high pressure turbine
CNI7DS	design corrected speed - intermediate turbine
CNLPDS	design corrected speed - low pressure turbine
CVDNOZ	duct nozzle velocity coefficient
CVMNOZ	main nozzle velocity coefficient
DELFG, DELFN, DELSFC	delta degradation multiplier for gross thrust, net thrust, and specific fuel consumption, respectively. Usually input as 1.0
DPAFDS	afterburner design pressure drop, $\Delta P/P$
DPCODS	combustor design pressure drop, $\Delta P/P$

DTCODS	combustor design temperature rise, $\Delta T$ ( $^{\circ}R$ )
ETAA	afterburner efficiency
ETABDS	combustor efficiency at design
ETACDS	high-pressure compressor adiabatic efficiency at design
ETAD	duct burner efficiency
ETAFDS	fan adiabatic efficiency at design
ETAIDS	intermediate compressor adiabatic efficiency at design
ETAR	ram recovery, $P_2/P_1$
ETHPDS	high-pressure turbine adiabatic efficiency at design
ETIPDS	intermediate turbine adiabatic efficiency at design
ETLPDS	low-pressure turbine adiabatic efficiency at design
HPEXT	horsepower extraction
LAFTBN	index on afterburning desired
LAMTP	index on ram or inlet operation desired
IDBURN	index on duct burning desired
IDCD	duct nozzle will be convergent-divergent when $IDCD=1$
IDES	index for design point; must be set equal to 1 to design engine; zeroed automatically
IDUMP	index for dumping error matrix
IGASM	index for mixed flow or non-mixed flow turbofans
IMCD	main nozzle will be convergent-divergent when $IMCD=1$
ITRYS	index for maximum number of iterations
MODE	independent variable designator for engine operation
NOZFLT	index for floating main or duct nozzle
PCBLC	% of total high-pressure compressor airflow that is bled
PCBLDUC	% of high-pressure compressor bleed flow that is bled into duct



PCBLDUI	% of intermediate compressor bleed flow that is bled into duct
PCBLF	% of total fan airflow that is bled (overboard)
PCBLHPC	% of high-pressure compressor bleed flow that is bled to the high-pressure turbine
PCBLHPI	% of intermediate compressor bleed flow that is bled to the high pressure turbine
PCBLI	% of total intermediate compressor airflow that is bled
PCBLIPC	% of high-pressure compressor bleed flow that is bled to the intermediate turbine
PCBLIPI	% of intermediate compressor bleed flow that is bled to the intermediate turbine
PCBLLPC	% of high-pressure compressor bleed flow that is bled to the low-pressure turbine
PCBLLPI	% of intermediate compressor bleed flow that is bled to the low-pressure turbine
PCBLOBC	% of high pressure compressor bleed flow that is bled overboard
PCBLOBI	% of intermediate compressor bleed flow that is bled overboard
PCNC	high-pressure compressor shaft speed as a percent
PCNCDS	design high-pressure compressor shaft speed as a percent
PCNF	fan shaft speed as a percent
PCNFDS	design fan shaft speed as a percent
PCNI	intermediate compressor shaft speed as a percent
PCNIDS	design intermediate compressor shaft speed as a percent
PRCDS	high-pressure compressor pressure ratio at design
PRFDS	fan pressure ratio at design
PRIDS	intermediate compressor pressure ratio at design
PS55	static pressure at low-pressure turbine exit (atm)
P2	fan face total pressure, for nonstandard days (atm)

TFHPDS	high-pressure turbine flow function at design
TFIPDS	intermediate turbine flow function at design
TFLPDS	low-pressure turbine flow function at design
TOLALL	tolerance on convergence at error matrix
T2	fan face total temperature, for nonstandard days (°R)
T24	duct-burner exit temperature, when ductburning (°R)
T4	combustor exit/turbine inlet temperature (°R)
T4DS	combustor exit/turbine inlet temperature at design (°R)
T7	afterburner exit temperature, when afterburning (°R)
WFA	afterburner fuel flow rate, IAFBN=2 (lbm/sec)
WFB	main combustor fuel flow rate, MODE=2 (lbm/sec)
WFBDS	combustor fuel flow rate at design, MODE=2 (lbm/sec)
WFD	duct burner fuel flow rate, IDBURN=2 (lbm/sec)
ZCDS, ZFDS, ZIDS	design ratio of high pressure compressor, fan, and intermediate compressor respectively; equals pressure ratio at design on design speed line minus one divided by high(surge) value minus one on the design speed line

# OUTPUT SYMBOLS<sup>1</sup>

<sup>1</sup>

Some symbols, such as T, are followed by station numbers; see station number symbols

In addition to all input symbols, the following are output:

A	areas (ft <sup>2</sup> )
AM	Mach numbers
BLDU	total bleed flow into duct (lbm/sec)
BLDUC	bleed flow into duct from high-pressure compressor (lbm/sec)
BLDUI	bleed flow into duct from intermediate compressor (lbm/sec)
BLC	high-pressure compressor bleed flow (lbm/sec)
BLF	fan bleed flow, overboard (lbm/sec)
BLHP	total bleed flow to high-pressure turbine (lbm/sec)
BLHPC	high-pressure turbine cooling bleed from high-pressure compressor (lbm/sec)
BLHPI	high-pressure turbine cooling bleed from intermediate compressor (lbm/sec)
BLI	intermediate compressor bleed flow (lbm/sec)
BLIP	total bleed flow to intermediate turbine (lbm/sec)
BLIPC	intermediate turbine cooling bleed from high-pressure compressor (lbm/sec)
BLIPI	intermediate turbine cooling bleed from intermediate compressor (lbm/sec)
BLLP	total bleed flow to low-pressure turbine (lbm/sec)
BLLPC	low-pressure turbine cooling bleed from high-pressure compressor (lbm/sec)
BLLPI	low-pressure turbine cooling bleed from intermediate compressor (lbm/sec)
BLOB	total overboard bleed flow (lbm/sec)
BLOBC	overboard bleed flow from high-pressure compressor (lbm/sec)
BLOBI	overboard bleed flow from intermediate compressor (lbm/sec)

BYPASS	bypass ratio, duct airflow divided by intermediate compressor airflow
CNC	high pressure compressor corrected shaft speed as a percent
CNF	fan corrected shaft speed as a percent
CNHP	high pressure turbine corrected shaft speed as a percent, $PCNC/\sqrt{T_4}$
CNHPCF	high pressure turbine speed correction factor
CNI	intermediate compressor corrected speed as a percent
CNIP	intermediate turbine corrected shaft speed as a percent, $PCNI/\sqrt{T_4.5}$
CNIPCF	intermediate turbine speed correction factor
CNLP	low-pressure turbine corrected shaft speed as a percent, $PCNF/\sqrt{T_5}$
CNLPCF	low-pressure turbine speed correction factor
CS	ambient speed at sound (ft/sec)
DHHPCF	high-pressure turbine work correction factor
DHIPC	intermediate turbine work correction factor
DHLPCF	low-pressure turbine work correction factor
DHTC	high-pressure turbine work, $\Delta H$ , (Btu/lbm)
DHTCHP	high-pressure turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R)
DHTCIP	intermediate turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R)
DHTCLP	low-pressure turbine work, temperature corrected, $\Delta H/T$ (Btu/lbm °R)
DHTF	low-pressure turbine work, $\Delta H$ (Btu/lbm)
DHTI	intermediate turbine work, $\Delta H$ (Btu/lbm)
DPAFT	afterburner/tailpipe pressure, loss, $\Delta P/P$
DPCOM	combustor pressure loss, $\Delta P/P$
DPDUC	fan duct pressure loss, $\Delta P/P$

DTAFCF	afterburner temperature rise correction factor
DTAFDS	afterburner temperature rise at design, $\Delta T$ ( $^{\circ}R$ )
DTCOCF	Combustor temperature rise correction factor
DTDUCF	duct-burner temperature rise correction factor
DTDUDS	duct-burner temperature rise at design, $\Delta T$ ( $^{\circ}R$ )
ETAACF	afterburner efficiency correction factor
ETAADS	afterburner efficiency at design
ETAB	combustor efficiency
ETABCF	combustor efficiency correction factor
ETAC	high-pressure compressor adiabatic efficiency
ETACCF	high-pressure compressor efficiency correction factor
ETAD	ductburner efficiency
ETADCF	duct burner efficiency correction factor
ETADDS	ductburner efficiency at design
ETAF	fan adiabatic efficiency
ETAFCF	fan efficiency correction factor
ETAI	intermediate compressor adiabatic efficiency
ETAI CF	intermediate compressor efficiency correction factor
ETATHP	high-pressure turbine adiabatic efficiency
ETATIP	intermediate turbine adiabatic efficiency
ETATLP	low-pressure turbine adiabatic efficiency
ETHPCF	high-pressure turbine efficiency correction factor
ETIPCF	intermediate turbine efficiency correction factor
ETLPCF	low-pressure turbine efficiency correction factor
FAR	fuel-air ratios
FART	total fuel-air ratio
FG	gross thrust (lbf)

FGM	momentum thrust, total (lbf)
FGMD	momentum thrust from duct (lbf)
FGMM	momentum thrust from core (lbf)
FGP	total pressure thrust (lbf)
FGPD	pressure thrust from duct (lbf)
FGPM	pressure thrust from core (lbf)
FN	net thrust (lbf)
FRD	ram drag (lbf)
H	enthalpies (Btu/lbm)
P	total pressures (atm)
PCNC	high-pressure compressor shaft speed as a percent
PCNCGU	first guess of PCNC
PCNF	fan shaft speed as a percent
PCNFGU	first guess at PCNF
PCNI	intermediate compressor shaft speed as a percent
PCNIGU	first guess at PCNI
PRC	high pressure compressor pressure ratio
PRCCF	high pressure compressor pressure ratio correction factor
PRF	fan pressure ratio
PRFCT	fan pressure ratio correction factor
PRI	intermediate compressor pressure ratio
PRICF	intermediate compressor pressure ratio correction factor
PS	static pressures (atm)
S	entropies (Btu/lbm - °R)
SFC	specific fuel consumption (lbm/lbf/hr)
T	total temperatures (°R)

TS	static temperatures
T2DS	fan face total temperature at design ( $^{\circ}\text{R}$ )
T21DS	fan exit total temperature at design ( $^{\circ}\text{R}$ )
T22DS	intermediate compressor exit total temperature at design ( $^{\circ}\text{R}$ )
T24DS	ductburner exit temperature at design ( $^{\circ}\text{R}$ )
T7DS	afterburner exit temperature at design ( $^{\circ}\text{R}$ )
T4GU	first guess at $T_4$ ( $^{\circ}\text{R}$ )
TFFHP	high-pressure turbine flow function
TFFIP	intermediate turbine flow function
TFFLP	low-pressure turbine flow function
TFHPCF	high-pressure turbine flow function correction factor
TFIPCF	intermediate turbine flow function correction factor
TFLPCF	low-pressure turbine flow function correction factor
V	velocities (ft/sec)
VA	aircraft velocity (ft/sec)
VJD	fan duct exhaust velocity (ft/sec)
VJM	core exhaust velocity (ft/sec)
WA	airflows (lbm/sec)
WAC	high-pressure compressor airflow (lbm/sec)
WACC	high-pressure compressor corrected airflow (lbm/sec)
WACCF	high-pressure compressor airflow correction factor
WACDS	high-pressure compressor airflow at design (lbm/sec)
WAD	duct airflow (lbm/sec)
WAF	fan airflow (lbm/sec)
WAFc	fan corrected airflow (lbm/sec)
WAFCF	fan airflow correction factor

WAI	intermediate compressor airflow (lbm/sec)
WAIC	intermediate compressor corrected airflow (lbm/sec)
WAICF	intermediate compressor airflow correction factor
WA3CDS	high-pressure compressor exit corrected airflow at design (lbm/sec)
WA23CDS	ductburner entrance corrected airflow at design (lbm/sec)
WFADS	afterburner fuel flow rate at design (lbm/sec)
WFDDS	ductburner fuel flow rate at design (lbm/sec)
WFT	total fuel flow rate (lbm/sec)
WGT	total gas flow rate (lbm/sec)
WG6CDS	afterburner entrance corrected gas flow at design (lbm/sec)
ZC, ZF, ZI	similar to ZCDS, ZFDS, and ZIDS except these are off-design - see Input Symbols



## SECTION I

### INTRODUCTION

In 1967, a digital computer program for balancing cycle turbofan engines titled SMOTE was developed in the Air Force Aero Propulsion Laboratory (Reference 1). This program is capable of simulating design and off-design performance of two-spool turbofan engines. This report describes a similar technique, obtained by modifying the SMOTE program, that simulates design and off-design performance of three-spool turbofan engines. NASA, Pratt and Whitney, General Electric and other contractors have developed or are developing working models that simulate the performance of three spool engines. However, through the experience gained in developing an in-house computer program, more flexibility can be exercised in making modifications to satisfy varying future needs without being dependent on contractors or other organizations.

Three-spool turbofan engines will be competitive as candidates for powering future aircraft systems. This effort has provided the Air Force Aero Propulsion Laboratory with an in-house method for evaluating the performance of three-spool turbofan engines.

## SECTION II

### SUMMARY

A computer program titled TRISPL is described. This program is derived from SMOTE (Simulation of Turbofan Engine) which was developed by the Turbine Engine Division of the AF Aero Propulsion Laboratory, Wright-Patterson AFB, Ohio.

TRISPL calculates design and off-design performance of 3-spool turbofan engines. Component maps, input as block data, are scaled internally to simulate a specific engine. The program is formulated for two core spools and one fan spool operation. Options are included for mixed or separate flow engines and dry or afterburning operation. The program can be modified for different modes of operation and engine types, 2 fan spools and one core spool for example.

### SECTION III

#### METHOD OF ENGINE CALCULATIONS

The following discussion is very similar to that in the report describing the SMOTE program.

##### 1. COMPONENT MAPS

The performance of the major engine components is based on component maps. These maps are usually obtained from analytical methods or rig-testing and are then converted into Block Data subroutines for use by TRISPL. The maps presently included in TRISPL are very general and do not represent any particular engine or engine components.

The component maps are scaled at the engine design point by TRISPL in order to match their performance to a desired set of performance figures which are input as data. Scaling or correction factors are calculated and then applied to the maps at off-design points. The scaling process is linear; therefore, correction factors near unity result in the highest accuracy of component simulation. Conversely, however, not being close to 1.0 does not necessarily mean that the simulation is poor since many maps have been shown to be typical over quite large ranges in the variables.

TRISPL presently includes component maps for the fan, intermediate compressor, high pressure compressor, combustor, and the three turbines. Duct burning, duct losses, gas mixing, afterburning, tailpipe losses, and nozzle losses are all calculated or input, but these characteristics could also be included as Block Data if maps were available. Likewise, schedules for bleed air and variable area nozzles could be used.

a. Fan-Compressor Maps

The fan and compressor maps are very similar and are plots of pressure ratio vs. corrected airflow with constant corrected speed lines and constant efficiency islands (see Fig. 1). Entry to the map is through the corrected speed and  $Z$ , where  $Z$  is a ratio of pressure ratios, and is defined at a constant corrected speed as shown in Figure 1. It is advantageous to use  $Z$  instead of pressure ratio because  $Z$  is restrained between the limits of 0 and 1, whereas the limits on pressure ratio vary depending upon map location and the particular map. Also, as indication that the fan or a compressor is approaching surge is given as  $Z$  approaches 1.

b. Combustor Map

The combustor map is a plot of efficiency vs. temperature rise for constant input pressure (see Fig. 2). Entry to the map is through temperature rise and input pressure, with efficiency being output.

c. Turbine Maps

The turbine map is a plot of work function vs. corrected speed with constant turbine flow function lines and constant efficiency islands (see Fig. 3). The work function and flow function are defined as

$$\text{DHTC} = \frac{H_{\text{IN}} - H_{\text{OUT}}}{T_{\text{IN}}}$$

and

$$\frac{W G_{\text{IN}} \sqrt{T_{\text{IN}}}}{P_{\text{IN}}}$$

Entry to the map is through corrected speed and flow function, with work function and efficiency being output.

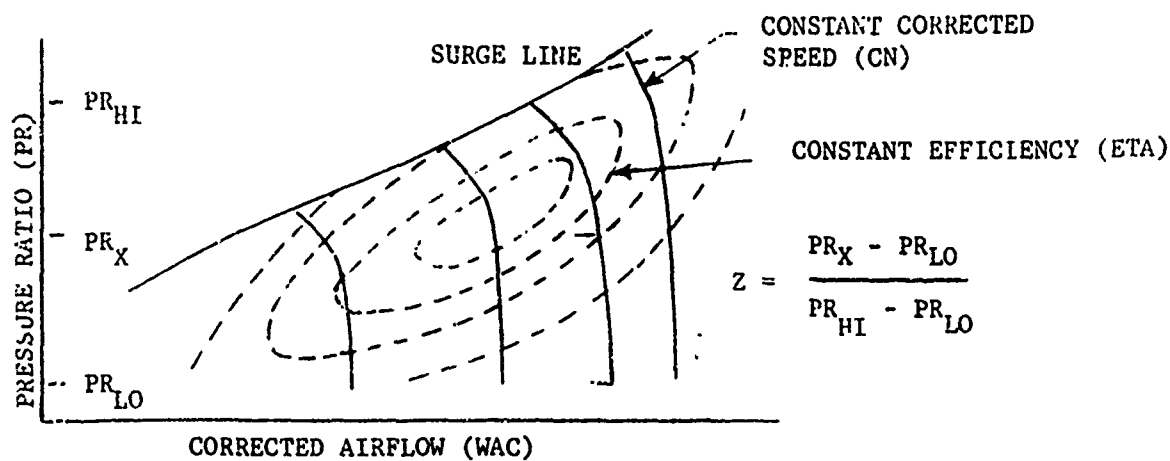


Figure 1. Example of Fan-Compressor Map

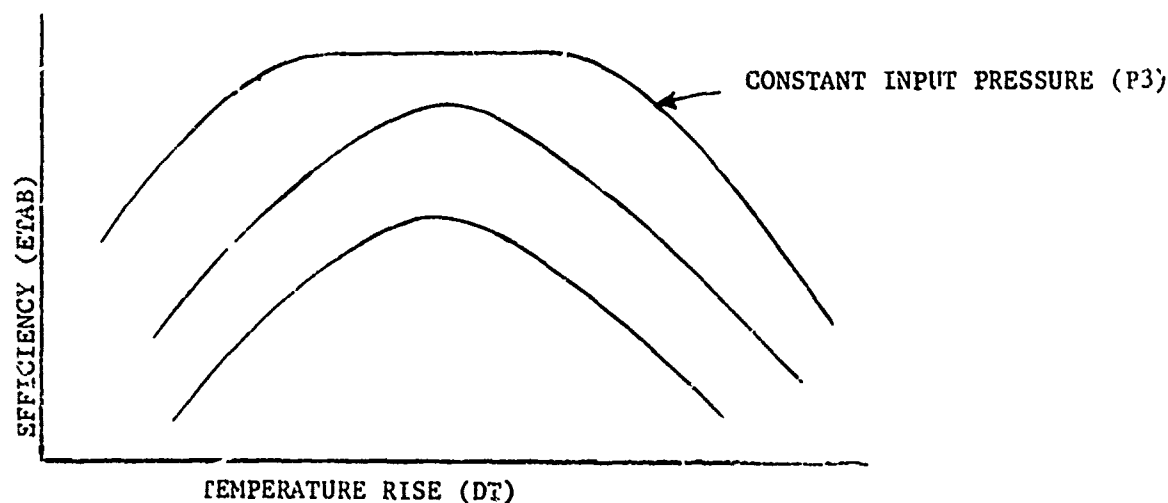


Figure 2. Example of Combustor Map

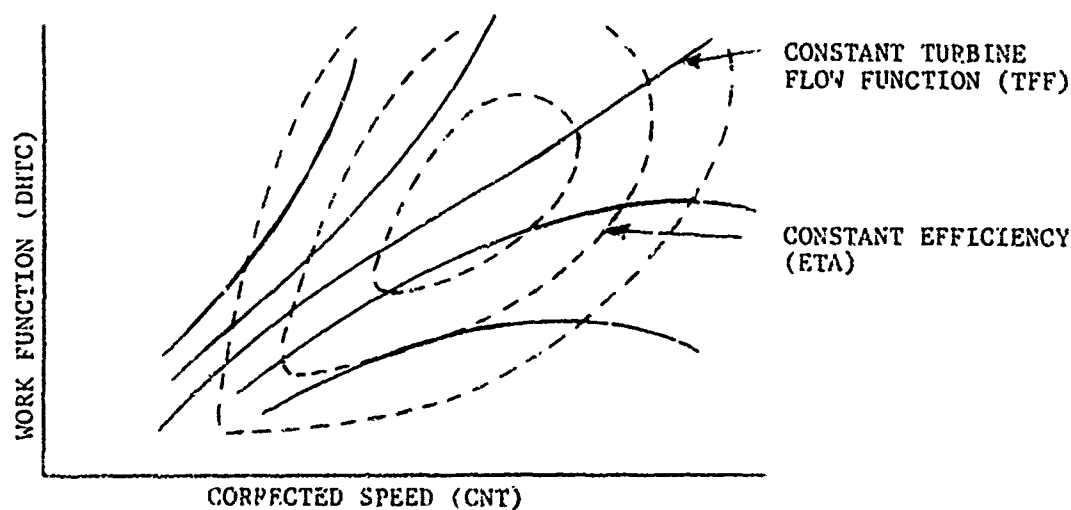


Figure 3. Example of Turbine Map

The work function could have been used as an entry in place of one of the present entries, but, because of the shape of the curves, this could lead to double entry points for one work function. However, if the turbine maps were plotted in a different format, this could be an acceptable method.

## 2. DESIGN POINT

Once the component maps have been reduced to Block Data form and placed in the program, it is necessary to run a design point. The design point is run at those conditions under which the real engine is designed or sized, often sea level static. Design parameters necessary to simulate the real engine (for example, airflow, bypass ratio, turbine inlet temperature, various pressure losses, pressure ratios, etc.) are input and a complete thermodynamic cycle calculation is performed. For more details on the cycle calculation see Section III 4, "Off-Design Points". Scale factors for the component maps are calculated to insure that the input design parameters are met. If the design parameters have been correctly input, the design point will be completed after one pass through the engine calculations (that is, no balancing will occur) because the maps are shifted to reduce the errors to zero.

Other parameters calculated and output at the design point include certain temperatures, airflows, gas mixing areas, and nozzle throat and exit areas.

## 3. SCALING FACTORS

Scaling or correction factors are calculated at the design point using the following equation:

$$P (\text{correction factor}) = P (\text{design}) / P (\text{map})$$

where P represents a general parameter. One exception to this equation

is the equation for calculating the fan and compressors pressure correction factors:

$$PR \text{ (correction factor)} = [PR \text{ (design)} - 1] / [PR \text{ (map)} - 1]$$

where PR represents a general pressure ratio.

Theoretically, if the component maps and the input design parameters are exact representations of a particular engine, the correction factors will equal 1. However, this will not be true due to map interpolations, certain assumptions such as ideal and isentropic flow, and tolerances in the thermodynamic calculations. If unmatched component maps are used, the correction factors can differ significantly from 1.

#### 4. OFF-DESIGN POINTS

The following discussion pertains particularly to off-design points, although the input and the general cycle calculations are the same for the design point. Throughout the following discussion, it should be remembered that scaling or correction factors (multipliers) are applied to all performance maps (Block Data parameters). A schematic diagram of the engine components and station designations is shown in Fig. 4.

##### a. Input

The program uses a controlled output; that is, the variables desired as output can be selected at the start of a run. This selection is obtained by placing the names of the variables in the first section of input cards. Controls, scaling factors and operating conditions make up the rest of the input.

The control inputs are used to determine the type of engine; mixed or separate flow, afterburning or duct burning, and convergent or convergent-divergent nozzle. The controls are also used to fix the

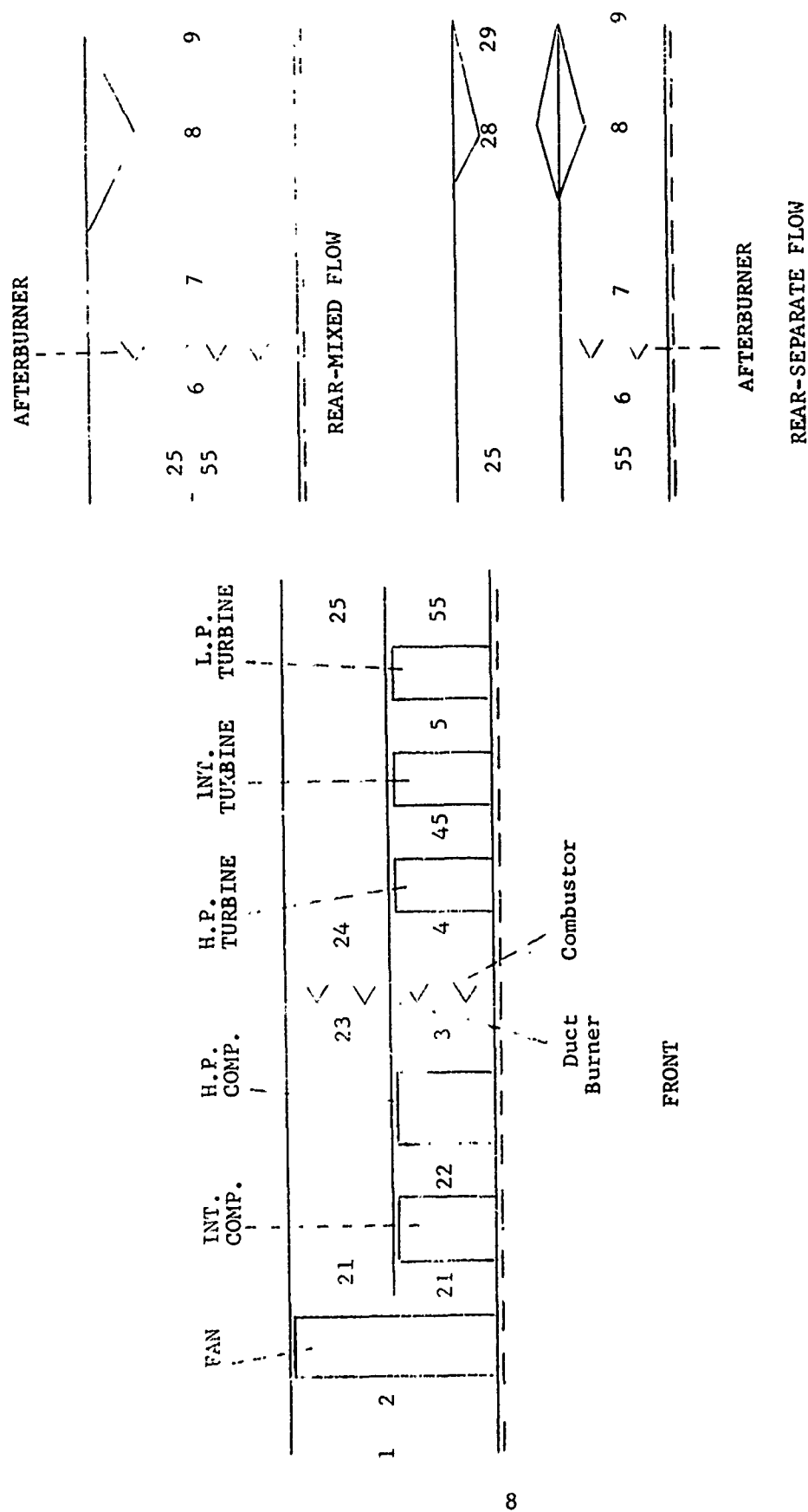


Figure 4. Schematic of Engine Components



mode of operation; constant PCNF, constant PCNC, constant T4, or constant WFB. Other controls determine inlet conditions, title printout, and cycle looping printouts. The correction factors can be input directly, or the design point can be run first and the calculated factors will be stored in common. The operating conditions include the flight Mach number, altitude, power setting (either PCNF, PCNC, T4, or WFB), duct burner and afterburner temperatures or fuel flows, bleed, and horsepower extraction.

b. Initial Values

The program uses six primary independent variables: ZF, PCNF, ZI, PCNI, ZC, PCNC (T4 may be substituted for PCNF or PCNC, depending upon the mode of operation). Three secondary independent variables (TFFHP, TFFIP, and TFFLP) are also used to ensure correct entry into the turbine maps. Initial values for these nine variables must be obtained to start the program at each point. A subroutine (GUESS) supplies these variables as a function of T2, T21, and some of the variables themselves. It is important to note that the closer the initial values are to the final values at a balanced point, the faster the program will run. Therefore, after a particular engine configuration has been run a few times, it is usually advisable to change the general initial value equations to suit the engine, using the knowledge gained from past runs to estimate more closely the final values of the variables.

c. Inlet

The thermodynamic properties of the atmosphere are found from a 1962 ARDC Atmosphere Tables subroutine. Using conservation of energy and isentropic flow, the conditions at the face of the fan can be found. A ram recovery can be input or, if not input, a ram recovery defined by MIL-E-5008E Specifications will be used. If desired, a T2-P2 direct input mode is available as are provisions for nonstandard day conditions.

d. Fan and Compressor

Block Data is used to determine the performance characteristics of the fan and compressors. When Z and PCN are known, the pressure ratio, corrected airflow and efficiency can be found by using a general Block Data interpolation routine named SEARCH. With the pressure ratio known and when the assumption of isentropic compression and the efficiency are used, the thermodynamic conditions at the exit of the fan and compressors can be calculated. Bleed for consumer use, leakage, or cooling is accounted for. Actual airflow leaving the fan and compressors is calculated from corrected airflow, temperature, pressure, and bleed.

It should be mentioned here that the present form of TRISPL calculates corrected speed (CN) in a manner slightly different from SMOTE. TRISPL uses an equation which forces corrected speed (CN) to equal physical shaft speed (PCN) at the design point. The equation is

$$CN = (PCN/\sqrt{\theta}) \cdot \sqrt{\theta_D}$$

where  $\theta$  is corrected temperature, and  $\theta_D$  is the corrected temperature at design. Thus, at the design point,  $CN = PCN$ . This equation is useful for studying theoretical engines where actual maps are not available but must be changed back to the SMOTE equation if real engines are to be simulated. SMOTE uses this equation:  $CN = PCN/\sqrt{\theta}$ .

e. Combustor

The pressure drop in the combustor is a function of a design pressure drop and ratio of corrected airflow to design corrected airflow.

Combustor efficiency is obtained from Block Data using SEARCH. The fuel used is assumed to be JP-4 (at 59°F), and, with the assumption of adiabatic and constant pressure combustion, a fuel heating value equation as a function of T4 has been derived. Thus the fuel/air ratio, fuel flow, and thermodynamic conditions at the combustor exit can be calculated. If WFB is known instead of T4, a small iteration is necessary.

f. Turbines

The turbine subroutines all use similar logic and obtain their performance characteristics from Block Data using subroutine SEARCH. All three turbine parameters (CN, TFF, DHTC) can be calculated before entering the turbine map, but only two are needed. Therefore, the third parameter obtained from the map is compared with the calculated third parameter, and a balancing error is generated if they are not equal. In this program, CN and TFF are used for map entries, and DHTC is used to generate the error. In addition, the efficiency is also obtained through SEARCH.

In addition, another error will be generated if TFF is not within map limits. The error will be the difference between TFF and the nearest map limit. This error becomes particularly important when the estimated initial values of the independent variables are far from the correct values, and the point is extremely unbalanced. When either TFF or CN is not within map limits, they are set to the nearest map limit, and one of the independent variables is changed in an attempt to rectify the situation. The operating point must appear on all maps before a complete cycle calculation can be accomplished.

Horsepower extraction is accounted for in calculating DHTC of the high pressure turbine. When the efficiency is used and the turbine process is assumed isentropic, the thermodynamic properties at the three turbine exits can be calculated. Any bleed airflow for cooling the turbines is treated as if it entered the main stream behind the turbine, and the thermodynamic properties at the turbine exit are recalculated to account for this.

g. Duct

The duct airflow and bypass ratio are calculated from the fan and intermediate compressor airflows. The pressure drop in the duct is treated as in the main combustor. For duct burning, the same fuel heating value equation that was used in the main combustor is again used, but the efficiency must be input. As in the combustor, either the temperature (T24) or the fuel flow (WFD) may be input.

If a separate flow engine is being simulated, the duct nozzle calculations are done in this routine, although they are accomplished in the same manner as for the main nozzle.

h. Mixer

The gas mixing areas (duct exit and turbine discharge for a mixed flow engine or just the turbine discharge area for a separate flow engine) are calculated at the design point using either an input core static pressure or Mach number. In the mixed flow mode, there is an option for calculating afterburner entrance area as a function of an input afterburner entrance Mach number at the design point. At an off-design point the areas are used to calculate static pressures and Mach numbers.

For a separate flow engine, the thermodynamic conditions entering the afterburner are now known, since they are identical to turbine discharge conditions.

For a mixed flow engine, a set of derived equations based on one-dimensional fluid flow theory and conservation of mass, energy, and momentum is used to determine the thermodynamic conditions after complete mixing of the two gas streams. These equations do not require that the static pressures of the two entering streams be equal. However, for a correct engine balance, the two static pressures must be equal, and a balancing error is generated if they are not equal.

#### i. Afterburner

The dry loss (cold loss) pressure drop in the afterburner is a function of a design pressure drop and the ratio of corrected gas flow to the design corrected gas flow.

For afterburning, the same equation for the fuel heating value that was used in the combustor is again used, but the efficiency must be input. As in the combustor, either temperature ( $T_2$ ) or the fuel flow (WFA) may be input. A momentum loss (hot loss) pressure drop is also calculated.

#### j. Nozzle

The main nozzle program uses fixed effective areas (except when afterburning or when different nozzle areas are directly input) calculated at the design point. Either a convergent or convergent-divergent subroutine may be used depending upon the input controls. If afterburning has been selected, the nozzle areas are allowed to float to obtain optimum performance; however, the areas are returned to their original design values after the afterburning point is completed. Nozzle

areas can also be changed by directly inputting different nozzle area values. The duct nozzle behaves identically to the main nozzle, including floating areas if duct-burning has been selected.

Because all thermodynamic properties of the gas stream are known, as well as the amount of flow, nozzle areas, and ambient pressure, there is a redundant parameter. For this program, the total pressure of the gas stream was chosen as the redundant parameter. The nozzle calculations are made without using the total pressure, and a required total pressure compatible with all other known parameters is calculated. This required pressure is compared with the actual pressure and a balancing error is generated if they are not equal.

#### k. Performance and Output

At this point, nine errors have been generated after one pass through the engine. Several more passes must be completed under control of the error matrix and engine balancing technique. See Section IV for a detailed description of the balancing technique. Eventually however, the errors will be reduced to zero, and engine performance will be calculated using standard equations. Gross thrust is obtained by summing the momentum term (a nozzle velocity coefficient may be input) and a pressure-area term, and the net thrust is in turn found by subtracting a ram drag (airflow momentum loss at inlet) term from the gross thrust. Specific fuel consumption (SFC) is total fuel flow divided by net thrust.

As previously mentioned, a controlled output is used, whereby only selected variables are printed. Each variable is labeled with its name and provisions have been made for changing the name of a variable. In addition, the values of all variables in common are printed in a close format so that variables other than those selected for a specific run are available later on.

## 5. QUADRATIC INTERPOLATION ROUTINE

Throughout the program there are many small loops (for example, thermodynamic iterations and table look-up) which require convergence. Trial-and-error methods and linear interpolations can be time consuming, especially when a tight tolerance is necessary; therefore, a general interpolation routine called AFQUIR (Air Force Quadratic Interpolation Routine) (Reference 1) is used.

This routine requires a dummy array dimensioned for nine locations. Also input into the routine through the calling argument are the independent and the dependent variables, the answer or value which the dependent variable is to converge upon, the number of tries at convergence, the tolerance, and a variable called DIR.

The DIR is either set or calculated in the calling program and is an initial guess at the direction and percentage change to apply to the first value of the independent variable. If not enough is known about the variable to calculate DIR, an arbitrary value may be set. This should not affect the final result, but may increase the number of tries of convergence.

The DIR thus establishes the second value at the independent variable. This value is used in the calling program to determine a corresponding second value of the dependent variable and AFQUIR is called a second time with two sets of values. A linear interpolation is made which results in a third value of the independent variable. AFQUIR is then called a third time with the third values of the independent and dependent variables and a quadratic interpolation is made. The values of these three sets of variables have been stored in the dummy array, and from hereon, quadratic

interpolations are made using the three sets which give values closest to the answer. Values farthest from the answer are lost.

Various safeguards are built into AFQUIR to return the interpolation method to DIR or linear if the roots of the quadratic become complex, if the quadratic does not intercept the answer, if the value of the independent variable differs radically from previous values, or if two sets of independent and dependent variables are identical.

Also it is possible to preload the dummy array directly at the linear or quadratic interpolations if desired.

In summary, AFQUIR is a completely flexible routine which performs quadratic interpolation for quick convergence of general functions.



## SECTION IV

### BALANCING TECHNIQUE

The balancing technique is virtually the same as that used in SMOTE. It is based on finding a solution for a set of partial differential equations. For this program, the set is composed of nine equations; however, using a set of only three equations will simplify the following discussion. This corresponds to a basic single spool turbojet simulation. SMOTE uses a set of six equations.

As discussed previously, nine independent variables (ZF, PCNF or T4, ZI, PCNI, ZC, PCNC or T4, TFFHP, TFFIP, and TFFLP) were selected. Once these variables have been given initial values, it is possible to proceed through an entire engine cycle calculation. Nine errors are generated as shown in Section III. These initial values of the nine variables and nine errors are referred to as base values.

In the following equations, V refers to a variable and E to an error. The basic set of differential equations based on  $E = f(V)$  is

$$dE_1 = \frac{\partial E_{11}}{\partial V_1} dV_1 + \frac{\partial E_{12}}{\partial V_2} dV_2 + \frac{\partial E_{13}}{\partial V_3} dV_3$$

$$dE_2 = \frac{\partial E_{21}}{\partial V_1} dV_1 + \frac{\partial E_{22}}{\partial V_2} dV_2 + \frac{\partial E_{23}}{\partial V_3} dV_3$$

$$dE_3 = \frac{\partial E_{31}}{\partial V_1} dV_1 + \frac{\partial E_{32}}{\partial V_2} dV_2 + \frac{\partial E_{33}}{\partial V_3} dV_3$$

where the single subscripts correspond to three variables and three errors and where the double subscripts indicate the change in a particular error (first subscript) due to a change in a particular variable (second subscript).

Assuming small changes result in the following approximations (where B refers to a base value):

$$dE = E - EB$$

$$dV = V - VB$$

$$\frac{\partial E}{\partial V} = \frac{\Delta E}{\Delta V}$$

With these approximations and the fact that E should be zero when the engine is balanced, the set of partial differential equations reduces to

$$E_1 - EB_1 = \frac{\Delta E_{11}}{\Delta V_1} dV_1 + \frac{\Delta E_{12}}{\Delta V_2} dV_2 + \frac{\Delta E_{13}}{\Delta V_3} dV_3 = -EB_1$$

$$E_2 - EB_2 = \frac{\Delta E_{21}}{\Delta V_1} dV_1 + \frac{\Delta E_{22}}{\Delta V_2} dV_2 + \frac{\Delta E_{23}}{\Delta V_3} dV_3 = -EB_2$$

$$E_3 - EB_3 = \frac{\Delta E_{31}}{\Delta V_1} dV_1 + \frac{\Delta E_{32}}{\Delta V_2} dV_2 + \frac{\Delta E_{33}}{\Delta V_3} dV_3 = -EB_3$$

Three more passes (nine for TRISPL) are now made through the engine cycle calculations, and one variable is changed by a small amount ( $\Delta V$ ) for each pass. The change in each error due to the small change in the variables ( $\Delta E/\Delta V$ ) can be calculated.

The above set of differential equations can now be solved for  $dV_1$ ,  $dV_2$  and  $dV_3$  and, in general, the new value of each independent variable would be given by

$$V = VB + dV$$

If the engine cycle calculations were linear functions, the engine would balance (errors equal zero) with these new values of the variables. However, this is usually not the case. The new errors become base errors (still keeping the old  $\Delta E/\Delta V$ 's) and another attempt at balance is performed. If several such attempts still fail, the entire process is repeated where the new errors and variables become base values and a new set of  $\Delta E/\Delta V$ 's are calculated.

A subroutine to determine the solution of a matrix is used to solve the set of differential equations. After each pass through the engine, a matrix array is loaded with the appropriate values; after ten passes (base value plus nine independent variables) the matrix subroutine is called to solve the matrix.

It was found that the "dV's" obtained from the solution of the differential equations were in many cases too large, thus causing the variables to exceed their limits, and to make it practically impossible to balance the cycle. The "dV's" are therefore multiplied by a suppression factor calculated in the program which limits the swing of the variables. Although this procedure may tend to increase the number of passes before balancing in some cases, it also balances points which previously would not balance. These points are most generally far from the design point, where oscillations of the dependent variables tend to build up.

## SECTION V

### INPUT/OUTPUT DESCRIPTIONS

#### 1. BLOCK DATA INPUT

The three compressor maps are entered into the program as BLOCK DATA subprograms FANDAT, INTDAT, and CMPDAT.

Using FANDAT as an example (refer to program listing), and referring to typical map (Fig. 1), the data are programmed as follows:

Card 1 identifies the program as BLOCK DATA. Card 2 is a comment card. Card 3 identifies the common block FAN into which data are to be stored and dimensions the program variables. Card 4 indicates that there are 10 speed lines N and the number of points NP on each line (6 on the lowest speed, 7 on the next 3 lines, etc). Card 5 assigns the value of speed to each of the 10 lines (low to high). The remaining cards indicate the values of pressure ratio (PR), corrected airflow (WAC), and efficiency (ETA) for the speed lines. For example, the card

DATA (PR (4,J), J = 1,7)/

denotes that the pressure ratios are for the 4th speed line (CN = 0.6) and that there are 7 points.

The combustor BLOCK DATA subprogram is CMBDAT. Referring to the program listing and a typical combustor map (Fig 2), the data are programmed as follows:

Card 1 identifies the program as BLOCK DATA. Card 2 identifies the common block COMB into which data are to be stored and dimensions the variables. Card 3 indicates that there are 15 lines of constant PSI(P3) by the value of N, and that there are 15 values of DELT (DT) and ETA(ETAB) along

each line of constant PSI(P3). Cards 4 and 5 assign values to each of the P3 lines from low to high pressure. Cards 6 to 8 assign values of  $\Delta T$  to each of the P3 lines starting at low  $\Delta T$ . The lowest value of  $\Delta T$  on each of  $\Delta T$  on the lowest value of P3. Next comes the second lowest value of  $\Delta T$  on each P3, etc. Cards 9 to 16 assign the value of ETAB in a one-to-one correspondence with the  $\Delta T$  values just assigned. The order is the same.

The turbine maps are the BLOCK DATA subprograms HTURB, ITURB, and LTURB. Taking HTURB as an example and referring to the program listing and a typical map (Fig.3) the data are programmed as follows: Card 1 identifies the subprogram as BLOCK DATA. Card 2 is a COMMENT card. Card 3 identifies the common block HTURB into which data are to be loaded and dimensions the program variables. Card 4 indicates the number of constant turbine flow function lines TFF as 11 (N) and the number of points on each line from low to high TFF. Cards 5 and 6 set values of TFF from low to high. The remaining cards set the values of corrected speed (CN), work function (DH), and efficiency (ETA) starting from low TFF. For example, the card,

DATA (DH(5,J), J = 1,15)/

denotes that the work functions are for the 5th flow function line (TFF = 49.175) and that there are 15 points.

## 2. CONTROLLED OUTPUT/NAMELIST INPUT

The input data is divided into two sections; data cards for the controlled output, and data cards in Namelist format for running each point. For the following discussion on setting up input data, refer to the listing of sample data immediately following the program listing.

a. Controlled Output

The variables that are to be output are selected by the first section of data cards. Any variable that is in one of the main commons (DESIGN, FRONT, SIDE, or BACK) may be selected for output by punching the name at the variable as it appears in the common (with trailing blanks if necessary) in Columns 1 through 7. Up to 150 variables (25 lines of 6 variables) may be chosen for a particular run. During the output phase, the name of the variable is printed out, with its value printed immediately below the name.

Another feature of the controlled output is the ability to change the name of a variable to be output; for example, it may be desired to change a station designation to one more common to a particular programmer. In this case, the variable name would be punched in Columns 1 through 7 as described above, but in addition, the desired name would be punched in Columns 15 through 22. Special symbols, such as /, may be used in the new name. The last card of the controlled output must be a card with THEEND punched in Columns 1 through 6.

In addition to the variables selected as controlled output, the values of all variables in common are printed in a close format so that variables other than those selected for a specific run are available later on.

b. NAMELIST Input

The normal data for running the desired points follows the controlled output data and is in a Namelist format, where the name of the Namelist is DATAIN. Usually the first set of data is the design point, as shown in the sample input data. When the design point is run (DDES = 1), all map scaling or correction factors are printed out, as well as being retained

in common. Therefore, it is possible to run off-design points immediately following the design point by making use of the values in common, or to begin running an off-design point immediately by inputting the scaling or correction factors. The first method is usually easier, but the second method may be desired if many points are to be run using the same engine parameters with no changes except for power setting, Mach number, and altitude.

The variables that must be input at the design point for the basic cycle (for example, no afterburning) are listed in Table I below:

TABLE I  
INPUTS REQUIRED FOR BASIC CYCLE AT DESIGN POINT

VARIABLE	DEFINITION	UNITS
PRFDS	Fan pressure ratio	
WAFDS	Fan face airflow	lb/sec
ETAFDS	Fan efficiency	
ZFDS	Design Z of fan	
PCNFDS	Fan shaft speed expressed as percent	
PRIDS	Intermediate compressor pressure ratio	
WAIDS	Intermediate compressor (core) airflow	lb/sec
ETAIDS	Intermediate compressor efficiency	
ZIDS	Design Z of intermediate compressor	
PCNIDS	Intermediate compressor shaft speed as a percent	
PRCDS	High pressure compressor pressure ratio	
ETACDS	High pressure compressor efficiency	
ZCDS	Design Z of high pressure compressor	
PCNCDS	High pressure compressor shaft speed as a percent	
ETABDS	Combustor efficiency	
DPCODS	Combustor pressure drop, $\Delta P/P$	
DTCODS	Combustor temperature rise	$^{\circ}\text{R}$
T40S	Turbine inlet temperature	$^{\circ}\text{R}$
TFHPDS	High pressure turbine flow function	$\text{lb}/\sqrt{\sigma R}/(\text{SFC})(P)$
CNHPDS	High pressure turbine corrected speed	
ETHPDS	High pressure turbine efficiency	
TFIPDS	Intermediate turbine flow function	$\text{lb}/\sqrt{\sigma R}/(\text{SFC})(P)$



VARIABLE	DEFINITION	UNITS
CNIPDS	Intermediate turbine corrected speed	
ETIPDS	Intermediate turbine efficiency	
TFLPDS	Low pressure turbine flow function	$\text{lb}\sqrt{\sigma R}/(\text{sec})(\text{psia})$
CNLPDS	Low pressure turbine corrected speed	
ETLPDS	Low pressure turbine efficiency	
DPDUDS	Fan duct pressure drop, $\Delta P/P$	
DPAFDS	Tailpipe pressure drop, $\Delta P/P$	
AM55 or PS55	Mach number at low pressure turbine exit Static pressure at low pressure turbine exit	atm
AM	Flight Mach number	
ALTP	Altitude	ft
HPEXT	Horsepower extraction	hp
CVMNOZ	Main nozzle velocity coefficient	
CVDNOZ	Duct nozzle velocity coefficient	

+

Various bleed flows (see Symbols)

+

Various control parameters (see below)

As mentioned in Table I, various control parameters which fix the engine type, mode of operation, method of calculating ram recovery, etc. must be input. These are listed below. Subroutine ZERO determines what values in common will be zeroed between points. None of the design values or correction factors are zeroed but some of the control parameters are. In the control parameter listing below, the superscripts (1) to (4) have the following meanings: (1) automatically returned to zero after

each point is calculated, must be re-input if option is again desired, (2) option can be used only at off-design points; (3) these input values remain as input unless changed by a new value; (4) a set-up case must be run where all the components are first matched before these  $\neq 0$  options are used, then the identical case may be repeated exercising these options.

	(1)	
IDES = 1		For calculating design point
	(3)	
MODE = 0		Specify T4
(2)	(3)	
MODE = 1		Specify PCNC
	(3)	
MODE = 2		Specify WFB
	(3)	
(2)MODE = 3		Specify PCNF
	(3)	
INIT = 0		Initializes point
	(3)	
INIT = 1		Will not initialize point
	(3)	
IDUMP = 0		No looping write-outs
	(3)	
IDUMP = 1		Will dump looping write-outs if error occurs
	(3)	
IDUMP = 2		Will dump looping write-outs after every point
	(3)	
IAMTP = 0		Will use AM and mil-spec ETAR
	(3)	
IAMTP = 1		Will use input AM and input ETAR
	(3)	
IAMTP = 2		Will use input T2 as T1=T1+T2 and standard P1 (T2 value needs to be input at every point or an error will occur whenever used)
	(3)	
IAMTP = 3		Will use input P2 and standard T1
	(3)	
IAMTP=4		Will use input T2 and input P2
	(3)	
IGASM=0		Separate flow, A6=A55

(3)		
IGASMX=1		Will mix duct and main streams, $A6=A25+A55$
(3)		
IGASMX=2		Will mix duct and main streams, input AM6
(4)	(1)	
IDBURN=1		For duct burning, input T24
(4)	(1)	
IDBURN=2		For duct burning input WFD
(4)	(1)	
IAFTBN=1		For afterburning (mixed or unmixed streams), input T7
(4)	(1)	
IAFTBN=2		For afterburning (mixed or unmixed streams), input WFA
(3)		
IDCD=1		FAn duct nozzle will be convergent-divergent
(3)		
IMCD=1		Main nozzle will be convergent-divergent
(4)	(3)	
NOZFLT=1		For floating main nozzle exit area
(4)	(3)	
NOZFLT=2		For floating duct nozzle exit area
(4)	(3)	
NOZFLT=3		For floating duct and main nozzle exit areas
ITRYS=N		Number of passes through engine before quitting
TOLALL=X		Tolerance which the errors must satisfy before engine is matched
DELFG, DELFN, DELSFC		Normally input as 1.0 unless a correction is desired
(1)		
ITITLE=1		A title card must follow after the input data for this point (see below)

A title card must be input immediately after the first point of the data pack and ITITLE must be set equal to 1 in the data for the first point. This is because a title is always printed for each point and must therefore be previously defined. The input format for the title is 12A6 and the resulting 72 spaces are centered on the page when printed out. The title may be changed by setting ITITLE=1 and inserting a new title card after the Namelist data for the point.

### Off-Design Operation

Shown in the sample input listing are methods of specifying off-design operation points. The user inputs the appropriate control parameters, Mach number, altitude and power setting other than design values (power setting being a value for T4, PCNC, PCNF, or WFB).

If the engine has all its nozzles fixed, then an input such as T4 or shaft speed will set the thrust level. Other means of changing engine operation can be accomplished by varying nozzle throat areas A8 and A28. For example, an off-design condition may exist where the operating point lies outside the limits of the block data input for a component map such as the fan. A nozzle throat area change could return the operating point back on the input map. It should be noted that an area remains changed until it is recalculated by a new design case or altered by a new input.

The nozzle exit area (A9 and/or A29) may be floated to obtain full expansion of using NOZFLT=1, 2 or 3 for non-afterburning cases.

To run duct burning (fan stream only), cases load ETAD, and either T24 or WFD. To run afterburning, cases load ETAA, T7 or WFB. When such a point is run, the exhaust nozzle areas are allowed to float to obtain optimum expansion. This means that there can be no balancing at the point, and it is necessary to prebalance the engine cycle in a nonaugmented mode. That is, an identical point, except that it is nonaugmented must be run before either afterburning or duct burning. When either IAFBN or IDBURN is greater than zero, the program will automatically set INIT=1 and use the balanced values from the preceding point. The nozzle areas are returned to their original values after completing an augmented point. Some examples of afterburning are given in the sample data listing.

SECTION VI  
SUBROUTINE DESCRIPTIONS

A flow chart of the computer program with the subroutines is shown in Figure 5. Listed here are brief descriptions of the subroutines.

TRISPL	Dummy main program to initiate the calculations and cause the input of the controlled output variables. Because of the looping between subroutines, control is never transferred back to this routine.
ENGBAL	Main subroutine. Controls all engine balancing loops, checks tolerances and number of loops and loads matrix.
MATRIX	Solves error matrix
INPUT	Reads Namelist data and title. Prints title,
ZERO	Zeroes common and certain controls
COINLT	Determines ram recovery and performs inlet calculations
ATMOS	1962 US Standard Atmosphere table
RAM	Calculates ram recovery defined by MIL-E-5008B Specifications
GUESS	Determines initial values of independent variables (PCNF, PCNI, PCNC, and T4) at each point. It may be desired to change these equations to suit a particular engine. The closer the initial values are to the final values, the faster the program will balance.
COFAN	Uses BLOCK DATA to perform fan calculations
COINTC	Uses BLOCK DATA to perform intermediate compressor calculations.

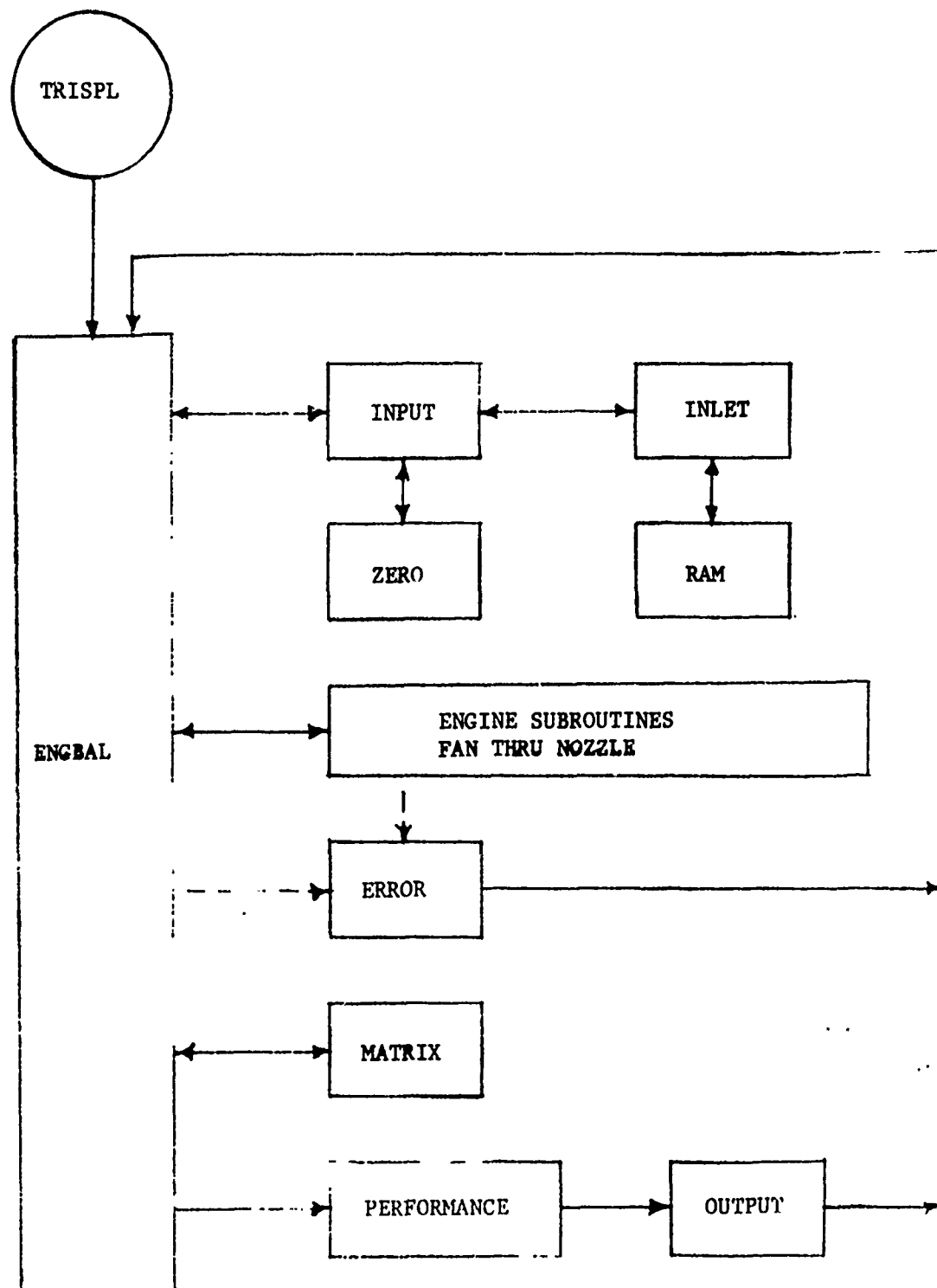


Figure 5. TRISPL Subroutine Flow Chart

COCOMP	Uses BLOCK DATA to perform high-pressure compressor calculations. Calculates ERR(7).
COCOMB	Uses BLOCK DATA to perform combustor calculations. May use either T4 or WFB as the main parameter.
COHPTB	Uses BLOCK DATA to perform high-pressure turbine calculations. Calculates ERR(1) and ERR(2).
COIPTB	Uses BLOCK DATA to perform intermediate turbine calculations. Calculates ERR(8) and ERR(9).
COLPTB	Uses BLOCK DATA to perform low-pressure turbine calculations. Calculates ERR(3) and ERR(4).
FRTOSD	Dummy routine to transfer values from common FRONT to common SIDE.
CODUCT	Performs duct and duct-burning calculations. May use T24 or WFD as the main parameter for duct burning. Controls the duct nozzle and calculates ERR(1) if in separate-flow mode.
FASTBK	Dummy routine to transfer values from common FRONT and SIDE to common BACK.
COMIX	Performs gas mixing calculations if in mixed flow mode. At design points it calculates areas from either an input static pressure (PS55) or an input Mach number (AM55) if PS55=0. Also, an option exists where afterburner entrance area A6 is calculated as a function of an input afterburner entrance Mach number AM6 at the design point. At off-design points it calculates static

pressures and Mach numbers from the design areas.  
Calculates ERR(5) if in mixed-flow mode.

COAFBN Performs afterburning calculations. May use either T7 or WFA as the main parameter.

COMNOZ Controls the main nozzle and calculates ERR(6).

PERF Calculates performance after engine is balanced.

OUTPUT Prints output except for controlled output. Prints the main commons in a close format after each point.

CONCUT Controls and prints the controlled output variables.

ERROR Controls all printouts if an error occurs. Prints name of subroutine where error occurred and also prints the values of all variables in the main commons.

SYG Controls printing from UNIT08. Throughout the program and particularly in ENGBAL, certain messages, variables, and matrix values are written on UNIT08 as an aid in determining why an error occurred or why a point did not balance. These values are printed out if subroutine ERROR is called and IDUMP is greater than zero, or after a good point if IDUMP=2.

TAPES Defines UNIT08, which is just a "scratch" disk and does not require a \$SETUP card. Normal input and output are on UNIT05 and UNIT06, respectively.

THCOMP Performs isentropic calculations for compressors.

THTURB Performs isentropic calculations for turbines.

THERMO Provides thermodynamic conditions using PROCOM.



PROCOM	Calculates thermodynamic gas properties for either air or a fuel-air mixture, based on JP-4.
SEARCH	General table lookup and interpolation routine to obtain data from Block Data routines.
MAPBAC	Used when calculations result in values not on the turbine maps. Changes the map value and an independent variable (PCNF, PCNC or T4) in an attempt to rectify the situation.
CONVRG	Performs nozzle calculations for a convergent nozzle.
CONDIV	Performs nozzle calculations for a convergent-divergent nozzle.
AFQUIR	General quadratic interpolation routine.
CMBDAT	Block Data for combustors.
FANDAT	Block Data for fan.
INTDAT	Block Data for intermediate compressor.
CMPDAT	Block Data for high-pressure compressor.
LTURB	Block Data for low-pressure turbine.
ITURB	Block Data for intermediate turbine.
HTURB	Block Data for high-pressure turbine.

SECTION VII  
PROGRAM LISTING

The following is a complete listing of all subroutines required  
to run TRISPL.

```
PROGRAM TRISPL(INPUT,OUTPUT,TAPES=INPUT,TAPE6=OUTPUT,TAPE8)  
COMMON /POINT/ICATPT  
IDATPT=0  
CALL CONOUT(1)  
CALL ENGBAL  
STOP  
END
```

```

SUBROUTINE FGBAL
COMMON / AGL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEDG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFCDU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3ZIOS ,PCNIOS ,PRIOS ,ETAIDS ,WAIDS ,FRICF ,ETAICF ,WAICF
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF
5T4DS ,WFBD ,DTCODS ,ETABDS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T20S
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T210S
8TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T220S
9T24DS ,WFODS ,OTDUDS ,ETADDS ,WA23DS ,OPDUDS ,OTDUCF ,ETADCF
AT7DS ,WFAOS ,OTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,OTAFCF ,ETAACF
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBG ,WA3
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC
EWFB ,TFFH ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLG ,PCBLDUI
FPCBLDUC ,PCBLOBI ,PCBLOBC ,PCBLHPI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLIPI
GPCBLFC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLDU ,XBLDUI ,XBLDUC
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6
6WAD ,WFD ,WG24 ,FAR24 ,ETAU ,DPOUC ,BYPASS ,DUMS7
7TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
1XT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25
2XWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9
8VA ,FRD ,VJD ,FGHD ,VJM ,FGHM ,FGPD ,FGPM
9VGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DIMENSION DELSAV(9)
DIMENSION VAR(9),DEL(9),ERRB(9),DELVAR(9),EHAT(9,9),VMAT(9),

```

```

1 AMAT(9)
  DATA DELSAV/9*.001/
  DATA AWORD/6HENGBAL/
  DATA VDELTA,VLIH,VCHNGE,NOMISX/
1 0.001,0.100,0.850,4/
  DATA DEL/9*0./
  CALL INPUT
  IF (INIT.EQ.1) GO TO 50
  TFFHP=TFHPDS
  TFFIP=TFIPDS
  TFFLP=TFLPDS
50 LOOPER=0
  NUMMAP=0
  NOMISS=0
1  LOOP=0
  MISMAT=0
  NOMAP=0
  IGO=2
  DO 2 I=1,9
    VMAT(I)=0.
    AMAT(I)=0.
    DELVAR(I)=0.
  DO 2 L=1,9
2  EMAT(I,L)=0.
3  LOOPER=LOOPER+1
  CALL COFAN
  WORD=AWORD
  IF (LOOPER.GT.ITRYS) GO TO 20
  IF (NOMAP.GT.0) GO TO 1
  NUMMAP=0
55 VAR(1)=ZF*100.
  IF (MODE.NE.3) VAR(2)=PCNF
  IF (MODE.EQ.3) VAR(2)=T4/10.
  VAR(3)=ZC*100.
  IF (MODE.NE.1) VAR(4)=PCNC
  IF (MODE.EQ.1) VAR(4)=T4/10.
  VAR(5)=TFFHP
  VAR(6)=TFFLP
  VAR(7)=ZI*100.
  VAR(8)=PCNI
  VAR(9)=TFFIP
  DO 4 I=1,9
    IF (ABS(ERR(I)).GT.TOLALL) GO TO 5
4  CONTINUE
  CALL PERF
  CALL ERROR
5  IF (LOOP.GT.0) GO TO 7
  MAPEOG=0
  MAPSET=0
  DO 6 I=1,9
    ERRB(I)=ERR(I)
6  DEL(I)=VDELTA*VAR(I)
  GO TO 9
7  IF (MISMAT.GT.0) GO TO 30
  IF (MAPEOG.EQ.0) GO TO 70

```

```

MAPEDG=0
MAPSET=1
VAR(LOOP)=VAR(LOOP)+2.*DEL(LOOP)
GO TO 10
70 IF (MAPSET.EQ.0) VAR(LOOP)=VAR(LOOP)+DEL(LOOP)
   IF (MAPSET.EQ.1) VAR(LOOP)=VAR(LOOP)-DEL(LOOP)
   MAPSET=0
   DO 8 I=1,9
   IF (DEL(LOOP).NE.0.) DELSAV(LOOP)=DEL(LOOP)
   IF (DEL(LOOP).EQ.0.) DEL(LOOP)=DELSAV(LOOP)
8   EMAT(I,LOOP)=(ERRB(I)-ERR(I))/DEL(LOOP)
9   LOOP=LOOP+1
   IF (LOOP.GT.9) GO TO 11
   VAR(LOOP)=VAR(LOOP)-DEL(LOOP)
10  ZF=VAR(1)/100.
   IF (MODE.NE.3) PCNF=VAR(2)
   IF (MODE.EQ.3) T4=VAR(2)*10.
   ZC=VAR(3)/100.
   IF (MODE.NE.1) PCNC=VAR(4)
   IF (MODE.EQ.1) T4=VAR(4)*10.
   TFFHP=VAR(5)
   TFFLP=VAR(6)
   ZI=VAR(7)/100.
   PCNI=VAR(8)
   TFFIP=VAR(9)
   IF (ZF.LT.0.) ZF=0.05
   IF (ZI.LT.0.) ZI=0.05
   IF (ZC.LT.0.) ZC=0.05
   GO TO (1,3), IGO
11  DO 12 I=1,9
12  AMAT(I)=-ERRB(I)
   DO 14 I=1,9
   IZERO=0
   DO 13 LOOP=1,9
13  IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1
   IF (IZERO.LT.9) GO TO 14
   WRITE(6,100) I
   LOOPER=ITRYS+100
   GO TO 20
14  CONTINUE
   DO 16 LOOP=1,9
   IZERO=0
   DO 15 I=1,9
15  IF (EMAT(I,LOOP).EQ.0.) IZERO=IZERO+1
   IF (IZERO.LT.9) GO TO 16
   WRITE(6,101) LOOP
   LOOPER=ITRYS+100
   GO TO 20
16  CONTINUE
17  CALL MATRIX(EMAT,VMAT,AMAT)
   LBIG=0
   VARBIG=0.0
   DO 18 L=1,9
   ABSVAR=ABS(VMAT(L))
   IF (ABSVAR.LE.VLIN*VAR(L)) GO TO 18

```

```

      IF (ABSVAR.LE.VARBIG) GO TO 18
      LBIG=L
      VARBIG=ABSVAR
18     CONTINUE
      VRATIO=1.0
      IF (LBIG.GT.0) VRATIO=VLIM*VAR(LBIG)/VARBIG
      ERRAVE=0.0
      VMTAVE=0.0
      DELAVE=0.0
      DO 19 L=1,9
      DELVAR(L)=VRATIO*VMAT(L)
      ERRAVE=ERRAVE+ABS(AMAT(L))
      VMTAVE=VMTAVE+ABS(VMAT(L))
      DELAVE=DELAVE+ABS(DELVAR(L))
19     VAR(L)=VAR(L)+DELVAR(L)
      ERRAVE=ERRAVE/9.
      VMTAVE=VMTAVE/9.
      DELAVE=DELAVE/9.
      IF (MISMAT.GT.0) GO TO 32
      IF (NOMISS.EQ.0) MISMAT=1
      IF (MISMAT.EQ.0) IGO=1
20     WRITE(8,102) LOOPER
      DO 21 I=1,9
21     WRITE(8,103) AMAT(I),(EMAT(I,L),L=1,9),VMAT(I),DELVAR(I),VAR(I)
      WRITE(8,104) ERRAVE,VMTAVE,DELAVE
22     IF (LOOPER.LT.ITRYS) GO TO 10
      CALL ERROR
      RETURN
30     VMTAVX=VMTAVE
      DO 31 I=1,9
31     AMAT(I)=-ERR(I)
      GO TO 17
32     WRITE(8,105) AMAT,ERRAVE,DELVAR,DELAVE,VMAT,VMTAVE,VAR
      MISMAT=MISMAT+1
      IF (VMTAVE.LT.VCHNGE*VMTAVX) GO TO 22
      WRITE(8,106)
      IF (MISMAT.LT.NOMISS) NOMISS=1
      MISMAT=0
      LOOP=0
      IGO=2
      GO TO 55
100    FORMAT(4H3ROW,I2,16H IS ZERO IN EMAT)
101    FORMAT(7HCOLUMN,I2,16H IS ZERO IN EMAT)
102    FORMAT(7H0   ERRB,31X23HERROR MATRIX AFTER LOOP,I4,31X4HVMAT,
16X6HDELVAR,7X10HVARIABLE$$)
103    FORMAT(1H0,F7.3,9F9.3,4XF9.3,3XF10.4,4XF11.4)
104    FORMAT(1H0,F8.4,32X14HAVERAGE VALUES,31X,2F11.4,5H$$$$$$)
105    FORMAT(12H0----- AMAT,10F11.6,6H$$$$$$,
1/,      12H -----DELVAR,10F11.6,6H$$$$$$,
2/,      12H ----- VMAT,10F11.6,6H$$$$$$,
3/,      12H -----  VAR,9F11.6,6H$$$$$$)
106    FORMAT(1H0,56X22HCHANGE TOO SMALL$$$$$$)
      END

```

```

SUBROUTINE INPUT
DIMENSION TITLE(12)
COMMON /POINT/IDATPT
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAHPT ,
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IOSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOMAP,NUMMAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIOS ,PCNIOS ,PRIOS ,ETAIOS ,WAIOS ,PRICF ,ETAICF ,WAICF ,
4ZCOS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFBDS ,DTCODS ,ETABDS ,WA3COS ,DPCODS ,DTCOCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2OS ,
7TFIPDS ,CNIPOS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21OS ,
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22OS ,
9T24DS ,WFODS ,DTCODS ,ETAODS ,WA23CS ,DPOUDS ,DTCOCF ,ETAOCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAAOS ,WG6COS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETA F ,WAF C ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,Z1 ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FFCBLDUC ,PCBL CBI ,PCBL OBC ,PCBLHPI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBL LPI ,
GPCBL LPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLOU ,XBLOUI ,XBLOUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6HAD ,WFC ,WG24 ,FAR24 ,ETAD ,DPOUC ,BYPASS ,DUMS7 ,
7TS28 ,FS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXHFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XFP1 ,DUMB ,
3Y6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,HFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGHD ,VJM ,FGHM ,FGPO ,FGFM ,
9FGH ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```



```

NAMELIST/DATAIN/ IEND,
AITITLE ,IDES ,MODE ,INIT ,ICUMP ,IAMTP ,IGASM ,
BIOBURN ,IAFTBN ,IDCO ,IMCO ,NCZFLT ,ITRYS ,TOLALL ,
CZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
DZIDS ,PCNIQS ,PRIQS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
EZCDS ,PCNCDS ,PRCDS ,ETACDS , ,PRCCF ,ETACCF ,WACCF ,
FT4DS ,WFEDS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
GTFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,OHHPCF ,T2DS ,
HTFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,OHIPCF ,T21DS ,
ITFLPCS ,CNLPCS ,ETLPCS ,TFLPCF ,CNLPCF ,ETLPCF ,OHLPCF ,T22DS ,
JT24DS ,WFUDS ,OTUDS ,ETAADS ,WA23DS ,OPUDS ,OTDUCF ,ETAACF ,
KT7DS ,WFAADS ,OTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,OTAFCF ,ETAACF ,
LA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
MT2 ,P2 ,T4 ,ZF ,PCNF ,ZI ,ZC ,PCNC ,
NWF8 ,TFFHP ,TFFIP ,TFFLP ,AM ,ALTP ,ETAR ,HPEXT ,
OPCBLF ,BLF ,PCBLI ,BLI ,PCBLC ,BLC ,PCBLDUI ,PCBLDUC ,
PBLDUI ,BLDUC ,PCBLOBI ,PCBLOBC ,BLOBI ,BLOBC ,PCBLHPI ,PCBLHPC ,
QPCBLIPI ,PCBLIPC ,PCBLIFI ,PCBLIPC ,BLHPI ,BLHPC ,BLIFI ,BLIPC ,
RBLIPI ,BLIPC ,PS55 ,AM55 ,AP6 ,T24 ,ETAD ,WFO ,
ST7 ,ETAA ,WFA ,CVDNOZ ,CVMNOZ ,DELFG ,DELFN ,DELSFC

```

```

DATA IEND/0/
C *** ITITLE=1 WILL READ IN TITLE
C *** IDES =1 FOR CALCULATING DESIGN POINT
C *** MODE =0 FOR CONSTANT T4
C *** MODE =1 FOR CONSTANT PCNC
C *** MODE =2 FOR CONSTANT WFB
C *** MODE =3 FOR CONSTANT PCNF
C *** INIT =1 WILL NOT INITIALIZE POINT
C *** IDUMP =1 WILL DUMP LOOPING WRITE-OUTS IF ERROR OCCURS
C *** IDUMP =2 WILL DUMP LOOPING WRITE-OUTS AFTER EVERY POINT
C *** IAMTP =0 WILL USE INPUT AM AND MIL SPEC ETAR
C *** IAMTP =1 WILL USE INPUT AM AND INPUT ETAR
C *** IAMTP =2 WILL USE T2 AS T1=T1+T2 AND STANDARD P1
C *** IAMTP =3 WILL USE P2 AND STANDARD T1
C *** IAMTP =4 WILL USE T2 AND P2
C *** IGASM=0 SEPARATE FLOW, A6=A55
C *** IGASM=1 WILL MIX DUCT AND MAIN STREAMS, A6=A25+A55
C *** IGASM=2 WILL MIX DUCT AND MAIN STREAMS, INPUT AM6
C *** IDURN=1 FOR DUCT BURNING, INPUT T24
C *** IDURN=2 FOR DUCT BURNING, INPUT WFO
C *** IAFTBN=1 FOR AFTERBURNING, INPUT T7
C *** IAFTBN=2 FOR AFTERBURNING, INPUT WFA
C *** IDCO =1 DUCT NOZZLE WILL BE C-0
C *** IMCO =1 MAIN NOZZLE WILL BE C-0
C *** NOZFLT=1 FOR FLOATING MAIN NOZZLE
C *** NOZFLT=2 FOR FLOATING DUCT NOZZLE
C *** NOZFLT=3 FOR FLOATING MAIN AND DUCT NOZZLES
C *** ITRYS =N NUMBER OF PASSES THRU ENGINE BEFORE QUITTING
DATA AWORD/6H INPUT/
IDATPT=ICATPT+1
CALL ZERO
WORD=AWORD
READ(5,DATAIN)
IF(IEND.NE.0) STOP
CALL REMARK(14HNEW DATA POINT)

```

```

      IF (IAFTBN.GT.0.OR.IDBURN.GT.0) INIT=1
      IF (ITITLE.EQ.1) READ(5,101) TITLE
      ITITLE=0
      WRITE(6,102) TITLE
      IF (MODE.EQ.0) WRITE(8,103) IDES,AM,ALTP,T4 ,T24,T7
      IF (MODE.EQ.1) WRITE(8,104) IDES,AM,ALTP,PCNC,T24,T7
      IF (MODE.EQ.2) WRITE(8,105) IDES,AM,ALTP,WFB ,T24,T7
      IF (MODE.EQ.3) WRITE(8,106) IDES,AM,ALTP,PCNF,T24,T7
      CALL COINLT
      RETURN
101  FORMAT(12A6)
102  FORMAT(1H1,30X12A6)
103  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
104  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H PCNC=,F8.3,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
105  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H WFB=,F8.4,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
106  FORMAT(1H0,7H IDES=,I3,10X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H PCNF=,F8.3,5X7H T24=,F8.2,5X7H T7=,F8.2,6H$$$$$$)
      END

```

```

SUBROUTINE ZERO
COMMON / ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2 IGASHX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3 ITRY5,LOOPER,NOMAP,NUMMAP,MAPEOG,TOLALL,ERR(9)
COMMON/ FRONT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7 CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8 CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4 ,
9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BL LPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC,PCBLCBI,PCBLOBC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBL LPI,
GFCBL LFC
COMMON/ SIDE/
1 XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLDU ,XBLDUI ,XBLDUC ,
2 XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3 T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4 T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5 T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6 WAD ,WFD ,WG24 ,FAR24 ,ETAO ,DPOUC ,BYPASS ,DUMS7 ,
7 TS28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWFB ,XWG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XP1 ,DUMB ,
3 T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4 T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5 WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6 PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7 TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8 VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9 FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DIMENSION Z1(94),Z2(56),Z3(72)
EQUIVALENCE (Z1(1),T1),(Z2(1),XP1),(Z3(1),XT55)
DATA ISTART/0/
ISTART=ISTART+1
IDES=0
JDES=0
INIT=0
IDBURN=0
IAFTBN=0
IDSHOC=3
IMSHOC=3
IF(ISTART.NE.1) GO TO 4
C*** ZERO FRONT,SIDE,AND BACK INITIALLY TO PREVENT CDC 6600 MODE ERRORS
DO 5 I=1,94
5 Z1(I)=0.

```

```

      DO 6 I=1,56
6     Z2(I)=0.
      DO 7 I=1,72
7     Z3(I)=0.
      CALL SYG(1)
      RETURN
4     CONTINUE
      T2Q=T2
      P2Q=P2
      T4Q=T4
      DO 1 I=1,94
1     Z1(I)=0.
      DO 2 I=1,56
2     Z2(I)=0.
      DO 3 I=1,72
3     Z3(I)=0.
      T2=T2Q
      P2=P2Q
      T4=T4Q
      CALL SYG(1)
      RETURN
      END

```

```

SUBROUTINE CCINLT
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX,IDBURN,IAFTBN,IOCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUMHAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFBS ,OTCDS ,ETABDS ,WA3CDS ,DPCDS ,DTCOCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIFDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFDS ,OTDUDS ,ETAADS ,WA23DS ,OPDUDS ,OTDUCF ,ETAOCF ,
AT7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVONoz ,CVHNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLOUI ,BLOUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
AGNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
8CNLP ,ETATLP ,DHTCLP ,DHTC ,BLLP ,BLLPI ,BL LPC ,CS ,
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNI ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC,PCBLOBI,PCBLOBC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBL LPI,
GPCBL LPI
DATA AWORD/6HCOINLT/
WORD=AWORD
AJ=778.26
G=32.174049
ALT=ALTP*2.0855531E+07/(2.0855531E+07-ALTP)
CALL ATMOS(ALT,T1,XX1,XX2,XX3,P1,CS,XX4,IIEP)
IF (IAMTP.EQ.2) T1=T1+T2
1 IF (IAMTP.NE.1) CALL RAM(AM,ETAR)
FAR=0.0
CALL FROCOM(FAR,T1,CS,XX2,XX3,R1,PHI1,H1)
S1=PHI1-R1*ALOG(P1)
H2=H1+(AM*CS)**2/(2.*AJ*G)
P2T=1.
DO 2 I=1,10
CALL THERMO(P2T,H2,T2T,S2T,AM,0,0.0,1)
IF (ABS(S2T-S1).LE.0.0001*S1) GO TO 3
2 P2T=P1*EXP((AM/1.986375)*((S2T-S1)+(1.986375/AM)*ALOG(P2T/P1)))
CALL ERROR
RETURN
3 IF (IAMTP.EQ.3.OR.IAMTP.EQ.4) ETAR=P2/P2T

```

```

P2=ETAR*P2T
IF (IAMTP.NE.4) CALL THERMO(P2,H2,T2,S2,XX5,C,0.0,1)
IF (IAMTP.EQ.4) CALL THERMO(P2,H2,T2,S2,XX5,0,0.0,0)
IF (INIT.EQ.1) GO TO 6
IF (IDES.EQ.1) GO TO 4
IF (MODE.EQ.3) GO TO 5
PCNF=GUESS (HCOE,T4,T4DS,PCNC,PCNCDS,WFB,WFBDS,T2,T2DS,PCNFDS)
PCNFGU=PCNF
GO TO 5
4 PCNF=PCNFDS
  PCNFGU=PCNF
  T2DS=T2
5 ZF=ZFDS
6 RETURN
END

```

```

C      SUBROUTINE ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
C      THIS IS A SUBROUTINE TO COMPUTE CERTAIN ELEMENTS OF THE 1962
C      U.S. STANDARD ATMOSPHERE UP TO 90 KILOMETERS.
C      CALLING SEQUENCE...

C      CALL ATMOS (ZFT, TM, SIGMA, RHO, THETA, DELTA, CA, AMU, K)
C      ZFT = GEOMETRIC ALTITUDE (FEET)
C      TM = MOLECULAR SCALE TEMPERATURE (DEGREES RANKINE)
C      SIGMA = RATIO OF DENSITY TO THAT AT SEA LEVEL
C      RHO = DENSITY (LB-SEC**2-FT**(-4) OR SLUGS-FT**3)
C      THETA = RATIO OF TEMPERATURE TO THAT AT SEA LEVEL
C      DELTA = RATIO OF PRESSURE TO THAT AT SEA LEVEL
C      CA = SPEED OF SOUND (FT/SEC)
C      AMU = VISCOSITY COEFFICIENT (LB-SEC/FT**2)

C      K = 1 NORMAL
C      = 2 ALTITUDE LESS THAN -5000 METERS OR GREATER THAN 90 KM
C      = 3 FLOATING POINT OVERFLOW

C      ALL DATA AND FUNDAMENTAL CONSTANTS ARE IN THE METRIC SYSTEM AS
C      THESE QUANTITIES ARE DEFINED AS EXACT IN THIS SYSTEM.

C      THE RADIUS OF THE EARTH (REFT59) IS THE VALUE ASSOCIATED WITH THE
C      1959 ARDC ATMOSPHERE SO THAT PROGRAMS CURRENTLY USING THE LIBRARY
C      ROUTINE WILL NOT REQUIRE ALTERATION TO USE THIS ROUTINE.
C      DIMENSION HB(10),TMB(10),DELTAB(10),ALM(10)
C      DATA (HB(I),I=1,10)/-5.,0.,11.,20.,32.,47.,52.,61.,79.,88.743/
C      DATA (TMB(I),I=1,10)/320.65,288.15,216.65,216.65,228.65,270.65,
C      1 270.65,252.55,180.65,180.65/
C      DATA (DELTAB(I),I=1,10)/1.75363,1.,2.23361E-01,5.40328E-02,
C      1 8.56663E-03,1.49455E-03,5.82289E-04,1.79718E-04,1.0241E-05,
C      2 1.6223E-06/
C      DATA (ALM(I),I=1,10)/-6.5,-6.5,0.,1.,2.8,2.,-2.,-4.,0.,0./
C      DATA REFT59/2.0855531E 07/, GZ /9.80665/,
C      A AMZ /28.9644 /, RSTAR /8.31432/,
C      B FITOKM/3.048E-04 /, S /110.4 /,
C      C AMUZ /1.2024E-05 /, CAZ /1116.45/,
C      D RHOZ /0.076474 /, GZENG /32.1741/

C      CONVERT GEOMETRIC ALTITUDE TO GEOPOTENTIAL ALTITUDE
C      HFT = (REFT59/(REFT59+ZFT))*ZFT
C      CONVERT HFT AND ZFT TO KILOMETERS
C      Z = FITOKM*ZFT
C      H = FITOKM*HFT
C      K = 1
C      TMZ = TMB(2)
C      IF (H.LT.-5.0.OR.Z.GT.90.0) GO TO 16
C      DO 10 M=1,10
C      IF (H-HB(M)) 11,12,10
C 10 CONTINUE
C      GO TO 16
C 11 M = M-1
C 12 DELH = H-HB(M)
C      IF (ALM(M).EQ.0.0) GO TO 13
C      TMK = TMB(M)+ALM(M)*DELH
C      GRADIENT IS NON ZERO, PAGE 10, EQUATION I.2.10-13

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      DELTA = DELTAB(M)*((TMB(M)/THK)**(GZ*AMZ/(RSTAR*ALH(M))))
      GO TO 14
13  THK = TMB(M)
C  GRADIENT IS ZERO, PAGE 10, EQUATION I.2.10-(4)
      DELTA = DELTAB(M)*EXP(-GZ*AMZ*DELH/(RSTAR*TMB(M)))
14  THETA = THK/THZ
      SIGMA = DELTA/THETA
      ALPHA = SQRT(THETA**3)*((TMZ+S)/(THK+S))
C  CONVERSION TO ENGLISH UNITS
      TH = 1.8*THK
      RHO = RHOZ*SIGMA/GZENG
      CA = CAZ*SQRT(THETA)
      AMU = AMUZ*ALPHA/GZENG
C  CALL OVERFL(J)
      J=2
      GO TO (15,17), J
15  K = K+2
      GO TO 17
16  K = 2
17  RETURN
      END

```



```

SUBROUTINE COFAN
COMMON / ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUPP ,IAHTP ,
2 IGASHX,IOURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3 ITRYS,LOOPER,NOHAP,NUMHAP,HAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,FCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3 ZIDS ,PCNICS ,FRIOS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAIICF ,
4 ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5 T4DS ,WF8DS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,OHHPCF ,T2DS ,
7 TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,OHIPCF ,T21DS ,
8 TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,OHLPFC ,T22DS ,
9 T24DS ,WFCDS ,OTDUDS ,ETADDS ,WA23DS ,OPDUDS ,OTOUFC ,ETADCF ,
A7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A6 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLDB ,
7 CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8 CNC ,FRC ,ETAC ,WACC ,WAC ,ETAB ,OPCCH ,WG4 ,
9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLLC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFR ,TFHP ,TFIP ,TFPL ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC,PCBLGBI,PCBLBGC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBLLLPI,
GPCBLLLPC
COMMON / FAN/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1 NCN,NPT(15)
DIMENSION WLH(2)
DATA AMORD,WLH/6H COFAN,6H (LO) ,6H (HI) /
WORD=AMORD
THETA=SQRT(12/518.658)
IF (IDES.NE.1) GO TO 10
THETAU=THETA
10 CNF=PCNF*THETAU/(100.*THETA)
IF (ZF.LT.0.) ZF=0.
IF (ZF.GT.1.) ZF=1.
CNFS=CNF
CALL SEARCH(ZF,CNF,PRF,WAF,ETAF,
1 CNX(1),NCN,PRX(1,1),WACX(1,1),ETAX(1,1),NPT(1),15,15,IGO)
IF ((1/CNF-CNFS).GT.0.0005*CNF) HAPECG=1
IF (IGO.EQ.1.OR.IGO.EQ.2) WRITE(8,1000) CNFS,WLH(IGO)
1000 FORMAT(19H0* * CNF OFF MAP,F10.4,2X A6,11H* * $$$$)
WAF=WAF*P2/THETA
IF (IDES.NE.1) GO TO 1
PRFCF=(PRFDS+1.)/(PRF-1.)

```

```

      ETAFCF=ETAFDS/ETAF
      WAFCF=WAFDS/WAF
      WRITE(6,100) PRFCF,ETAFCF,WAFCF,T2CS
100  FORMAT(11H0FAN DESIGN,13X6H PRFCF=,E15.8,8H ETAFCF=,E15.8,
18H WAFCF=,E15.8,8H T2CS=,E15.8)
1  PRF=PRFCF*(PRF-1.)+1.
   ETAF=ETAFCF*ETAF
   WAF=WAFCF*WAF
   CALL THCOMP(PRF,ETAF,T2,H2,S2,P2,T21,H21,S21,F21)
   IF(PCBLF.GT.0.) BLF=PCBLF*WAF
   IF(JDES.EQ.1) GO TO 7
   JDES=1
   IF(INIT.EQ.1) GO TO 6
   IF(IDES.EQ.1) GO TO 4
   IF(MODE.NE.2) GO TO 2
   T4=GUESS(3,Y1,Y2,PCNF,PCNFDS,WFB,WFBDS,Y7,Y8,T4DS)
   PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNIDS)
   PCNC=GUESS(4,Y1,Y2,PCNF,PCNFDS,WFB,WFBDS,Y7,Y8,PCNCDS)
   GO TO 5
2  IF(MODE.EQ.1) GO TO 3
   IF(MODE.EQ.0) GO TO 20
   T4=GUESS(7,Y1,Y2,PCNF,PCNFDS,Y5,Y6,T2,T2DS,T4DS)
20  PCNC=GUESS(5,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNCDS)
   PCNI=GUESS(9,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T21,T21DS,PCNIDS)
   GO TO 5
3  T4=GUESS(6,Y1,Y2,PCNC,PCNCDS,Y5,Y6,T21,T21DS,T4DS)
   PCNI=GUESS(8,T4,T4DS,Y3,Y4,Y5,Y6,T21,T21DS,PCNIDS)
   GO TO 5
4  PCNC=PCNCDS
   PCNI=PCNIDS
   T4=T4DS
   WFB=WFBDS
   T21DS=T21
5  ZC=ZCDS
   ZI=ZIDS
   PCNIGU=PCNI
   PCNCGU=PCNC
   T4GU=T4
6  INIT=0
7  IF(MODE.NE.3) GO TO 8
   IF(ABS(CNF-CNFS).LE.0.001*CNFS) GO TO 9
   WRITE(8,2000) CNFS,CNF
2000 FORMAT(10H0CNF WAS= ,E15.8,11H AND NOW= ,E15.8,
124H CHECK PCNF INPUTS$$$$)
   CALL ERROR
8  PCNF=100.*THETA*CNF/THETA0
9  CALL COINTC
   RETURN
   END

```

```

SUBROUTINE COINTC
COMMON/ ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
2 IGASMX ,IDURN ,IAFTBN ,IDCD ,IMCD ,IDSHGC ,IMSHGC ,NOZFLT ,
3 ITRY5 ,LOOPER ,NONAP ,NUMMAP ,NAPELG ,TOLALL ,ERR (9)
COMMON/ DESIGN/
1 PCNFGU ,PCNIGU ,PCNCCU ,T4GU ,DLMD1 ,DELFG ,DEIFN ,DELSFC
2 ZFDS ,PCMFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3 ZIDS ,PCNICS ,FRIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF
4 ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF
5 T4DS ,WFEOS ,DTGOS ,ETABOS ,WACOS ,DPCOS ,DTGOCF ,ETABCF
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHPHCF ,T2US
7 TFIPDS ,CNIPDS ,EYIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHPHCF ,T2IDS
8 TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHPHCF ,T2IDS
9 T24DS ,WFDOS ,DTGOS ,ETAOS ,WACOS ,DPCOS ,DTGOCF ,ETADCF
AT7DS ,WAFDS ,DTAFDS ,ETAADS ,WACOS ,DPAFDS ,DTAFCF ,ETAACF
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB
7 CNJ ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOB ,WA3
8 CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCCH ,WG4
9 CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BL LPC ,CS
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC
EWFB ,TFHFP ,TFHFP ,TFHFP ,PCBLF ,PCBLI ,PCBLC ,PCBLG
FPCBLDUC ,PCBLCBI ,PCBLBIC ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLLE
GPCBLLE
COMMON/ IPHF/WA22
COMMON/ INT/CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1 NCN,NPT(15)
DIMENSION WLH(2)
DATA AWORD,WLH/6HCOINTC,6H (LO) ,6H (HI) /
WORD=AWORD
THETA=SQRT(T21/518.668)
IF (IDES.NE.1) GO TO 10
THETAG=THETA
10 CN1=PCNI*THETAG/(100.*THETA)
IF (ZI.LT.0.) ZI=0.
IF (ZI.GT.1.) ZI=1.
CNIS=CN1
CALL SEARCH(ZI,CNI,PRI,WAIC,ETAX,
1 CNX(1),NCN,PRX(1,1),WACX(1,1),ETAX(1,1),NPT(1),15,15,IGO)
IF ((CNI-CNIS).GT..0005*CN1) NAPELG=1
IF (IGO.EQ.1.OR.IGO.EQ.2) WRITE(8,1000) CNIS,WLH(IGO)
1000 FORMAT(19H0* * * CNI OFF MAP,F10.4,2XA6,11H* * *$$$$$)
WAI=WAIC*P21/THETA
IF (IDES.NE.1) GO TO

```

```

PRICF=(PRIOS-1.)/(PRI-1.)
ETAICF=ETAIDS/ETAI
WAICF=WAIDS/WAI
WRITE(6,100)PRICF,ETAICF,WAICF,T21DS
100 FORMAT (23H01.P. COMPRESSOR DESIGN,1X8H PRICF=,E15.8,8H ETAICF=
1E15.8,8H WAICF=,E15.8,8H T21DS=,E15.8)
1 PRI=PRICF*(PRI-1.)+1.
ETAI=ETAICF*ETAI
WAI=WAICF*WAI
CALL THCOMP(PRI,ETAI,T21,H21,S21,P21,T22,H22,S22,P22)
IF(IDES.EQ.1) T22DS=T22
IF(PCBLI.GT.0) BLI=PCBLI*WAI
WA22=WAI-BLI
BLDUZ=PCBLCUI*BLI
BLHPI=PCBLHPI*BLI
BLIPI=PCBLIPI*BLI
BLOBI=PCBLCBI*BLI
PCNI=100.*THETA*CN1/THETA0
CALL COCOMP
RETURN
END

```

```

SUBROUTINE COCOMP
COMMON / ALL/
1 WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IATP ,
2 IGASMX,IOBURN,IAFTBN,IDCO ,IMCO ,IDSHOC,IMSHOC,HOZFLT,
3 IFRYS,LOOPER,MONAP,NUMMAP,MAPELG,TOLALL,ERR(9)
COMMON/DESIGN/
1 PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELF6 ,DELFN ,DELSFC .
2 ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3 ZIDS ,PCNIOS ,PRIOS ,ETAIOS ,WAIOS ,PRICF ,ETAI CF ,WAI CF ,
4 ZCDS ,PCNCCS ,PRCOS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5 T4DS ,WF8DS ,DTCCDS ,ETABDS ,WAZCDS ,DPCODS ,DTCCCF ,ETABCF ,
6 TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHFHCF ,T2DS ,
7 TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T2IDS ,
8 TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9 T24DS ,WF0DS ,DTDUOS ,ETAADS ,WAZ3DS ,OPDUOS ,DTDU CF ,ETAOCF ,
A T7DS ,WFAOS ,DTAFOS ,ETAADS ,W66CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29
CFS55 ,AM55 ,CYDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRGNT/
1 T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2
2 T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22
3 T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4
4 T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5
5 T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU
6 CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB
7 CNF ,PRI ,ETAI ,WAI ,WAI ,BLOB1 ,BLOB2 ,WA3
8 CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4
9 CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC
EWF8 ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDU
FPCBLDUC,PCBLCBI,PCBLORC,PCBLHPI,PCBLHPC,PCBLIFI,PCBLIPC,PCBLLP
GPCBLIPC
COMMON/ IPHP/ WA22
COMMON / COMP/ CNX(15),PRX(15,15),WACX(15,15),ETAX(15,15),
1 NCN,NPT(15)
DIMENSION WLH(2)
DATA AWORD,WLH/6HCOCOMP,6H (LO) ,6H (HI) /
WORD=AWORD
THETA=SQRT(T22/510.688)
IF (IDES.NE.1) GO TO 10
THETA0=THETA
10 CNC=PCNC*THETA0/(100.*THETA)
IF (ZC.LT.0.) ZC=0.
IF (ZC.GT.1.) ZC=1.
CNCS=CNC
CALL SEARCH(ZC,CNC,PRC,WACC,ETAC,
1 CNX(1),NCN,PRX(1,1),WACX(1,1),ETAX(1,1),NPT(1),15,15,IGO)
IF (MODE.EQ.1) GO TO 1
IF ((CNC-CNCS).GT.0.0005*CNC) MAPELG=1
1 IF (IGO.EQ.1.CR.IGO.EQ.2) WRITE(8,1000) CNCS,WLH(IGO)
1000 FORMAT(19H0* * * CNC OFF MAP,F10.4,2XA6,11H* * *$$$$$)
WAC=WACC*P22/THETA

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      IF (IDES.NE.1) GO TO 2
      WACDS=WA22
      PRCCF=(PRCDS-1.)/(PRC-1.)
      ETACCF=ETACDS/ETAC
      WACCF=WACDS/WAC
      WRITE(6,100)PRCCF,ETACCF,WACCF,T22DS
100  FORMAT(23H0H,P. COMPRESSOR DESIGN,1X8H PRCCF=,E15.8,8H ETACCF=,
1E15.8,8H WACCF=,E15.8,8H T22DS=,E15.8)
2    PRC=PRCCF*(PRC-1.)+1.
      ETAC=ETACCF*ETAC
      WAC=WACCF*WAC
      CALL THCOMP(PRC,ETAC,T22,H22,S22,F22,T3,H3,S3,P3)
      IF (PCBLC.GT.0.) BLC=PCBLC*WAC
      WA3=WAC-BLC
      BLDUC=PCBLDUC*BLC
      BLHPC=PCBLHPC*BLC
      BLIPC=PCBLIPC*BLC
      BLLPC=PCBLIPC*BLC
      BLOB=PCBLOB*BLC
      BLQI=BLQI+BLOUC
      BLHP=BLHPI+BLHPC
      BLIP=BLIPI+BLIPC
      BLLP=BLLPI+BLLPC
      BLOB=BLQI+BLCBC
      IF (MODE.NE.1) GO TO 3
      IF (ABS(CNC-CNCS).LE.0.001*CNCS) GO TO 4
      WRITE(6,2000)CNCS,CNC
2000 FORMAT(10H0CNC WAS= ,E15.8,11H AND NOW= ,E15.8,
124H CHECK PCNC INPUT$$$$$$)
      CALL ERROR
3    PCNC=100.*THETA*CNC/THETAD
4    ERR(7)=(WAC-WA22)/WAC
      CALL COCOMB
      RETURN
      END

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SUBROUTINE CCCOMB
COMMON / ALL/
1 WORD , IOES , JOES , KDES , MODE , INIT , IDUMP , IATP ,
2 IGASMX , ICBURN , IAFBND , IDCD , IMCC , IDSHOC , IMSHOC , NOZFLT ,
3 ITRY , LCOPE , NOHAF , NUMHAF , MAPEDG , TOLALL , ERR ( 9 )
COMMON / DESIGN /
1 PCNFGU , PCNIGU , PCNCGU , T4GU , OLMD1 , DELFG , DELFN , DELSFC ,
2 ZFDS , PCNFDS , PRFDS , ETAFDS , WAFDS , PRFCF , ETAFCF , WAFCF ,
3 ZIDS , PCNICS , PRIOS , ETAIDS , WAIDS , PRICF , ETAICF , WAICF ,
4 ZCOS , PCNCCS , PRCDS , ETACDS , WACDS , PRCCF , ETACCF , WACCF ,
5 T4DS , WFBD , DTCODS , ETABDS , WA3CDS , DPCODS , DTCOCF , ETABCF ,
6 TFHPDS , CNHPDS , ETHPDS , TFHPCF , CNHPCF , ETHPCF , DHHPCF , T2OS ,
7 TFIPDS , CNIPDS , ETIPDS , TFIPCF , CNIPCF , ETIPCF , DHIPCF , T21OS ,
8 TFLPDS , CNLPDS , ETLPDS , TFLPCF , CNLPCF , ETLPFC , DHLPCF , T22OS ,
9 T24DS , WFODS , CTODS , ETADDS , WA23DS , OPDUDS , DTODUCF , ETADCF ,
A7DS , WAFDS , DTAADS , ETAAADS , WG6CDS , DPAFDS , DTAFCF , ETAAFCF ,
BA55 , A25 , A6 , A7 , A8 , A9 , A28 , A29 ,
CPS55 , AM55 , CVDNOZ , CVMNOZ , A8SAV , A9SAV , A28SAV , A29SAV
COMMON / FRCNT /
1 T1 , P1 , H1 , S1 , T2 , P2 , H2 , S2 ,
2 T21 , P21 , H21 , S21 , T22 , P22 , H22 , S22 ,
3 T3 , P3 , H3 , S3 , T4 , P4 , H4 , S4 ,
4 T45 , P45 , H45 , S45 , T5 , P5 , H5 , S5 ,
5 T55 , P55 , H55 , S55 , BLF , BLI , BLC , BLJU ,
6 CNF , PRF , ETAF , WAF , WAF , BLDUI , BLDUC , BLOB ,
7 CN , PRI , ETAI , WAIC , WAI , BLOBI , BLOBC , WA3 ,
8 CNC , PRC , ETAC , WACC , WAC , ETAB , DPCOM , WG4 ,
9 CNHP , ETATHP , DHTCHP , DHTC , BLHP , BLHPI , BLHPC , FAR4 ,
ACNIP , ETATIP , DHTCIF , DHTI , BLIP , BLIPI , BLIFC , DUMF ,
BCNLP , ETATLP , DHTCLP , DHTF , BLLP , BLLPI , BLLPC , CS ,
CAG45 , FAR45 , WG5 , FAR5 , WG55 , FAR55 , HPEXT , AM ,
DALTP , ETAR , ZF , PCNF , ZI , PCNI , ZC , PCNC ,
EMFB , TFFHF , TFFIP , TFFLP , PCBLF , PCBLI , PCBLC , PCBLDUI ,
FPCBLDUC , PCBLCBI , PCBLBC , PCBLHPI , PCBLHPC , PCBLIFI , PCBLIFC , PCBLPI ,
GFCBLIFC
COMMON / COMB / PSI ( 15 ) , DELT ( 15 , 15 ) , ETA ( 15 , 15 ) , NPS , NPT ( 15 )
DIMENSION Q ( 9 ) , CUMBO ( 15 , 15 )
DATA AWORD / 6HCOCOMB /
WORD = AWORD
Q ( 2 ) = 0 .
Q ( 3 ) = 0 .
P3PSI = 14.696 * P3
WA3C = WA3 * SGRT ( T3 ) / P3PSI
IF ( IOES .EQ. 1 ) WA3CDS = WA3C
DPCOM = DPCODS * ( WA3C / WA3CDS )
IF ( DPCOM .GT. 1 . ) DPCOM = 1 .
P4 = P3 * ( 1 . - DPCOM )
1 IF ( T4 .GT. 3999 . ) T4 = 3999 .
IF ( T4 .GE. 1000 . ) GO TO 2
T4 = 1000 .
IF ( MODE .EQ. 1 ) MAPEDG = 1
2 CTCO = T4 - T3
IF ( IOES .NE. 1 ) GO TO 3
DTCOCF = DTCODS / DTCO
3 CTCO = CTCOCF * CTCO

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      F3PSIN=P3PSI
      CALL SEARCH(-1.,P3PSIN,DTCO,ETAB,CUMMY,
1 PSI(1),NPS,DELT(1,1),ETA(1,1),DUNE0(1,1),NPT(1),15,15,IGO)
      IF(IGO.EQ.7) CALL ERROR
4      IF(IDES.NE.1) GO TO 5
      ETABCF=ETABDS/ETAB
5      ETAB=ETABCF*ETAB
      HV=((((-0.4594317E-19*T4)-.2034116E-15)*T4+.2783643E-11)*T4
1+.2051501E-07)*T4-.2453116E-03)*T4-.9433296E-01)*T4+.1845537E+01
      CALL THERMO(P4,HA,T4,XX1,XX2,0,0.0,0)
      FAR4=(HA-H3)/(HV*ETAB)
      IF(FAR4.LT.0.) FAR4=0.
      WFBX=FAR4*WA3
      IF(MODE.NE.2) GO TO 8
      ERRW=(WFB-WFBX)/WFB
      DIR=SQRT(WFB/WFBX)
      CALL AFQUIR(Q(1),T4,ERRW,0.,20.,0.0001,DIR,T4T,IGO)
      GO TO (6,9,7),IGO
6      T4=T4T
      GO TO 1
7      CALL ERROR
8      WFB=WFBX
9      CALL THERMO(P4,H4,T4,S4,XX2,1,FAR4,0)
      WG4=WFB+WA3
      IF(IDES.EQ.1) WRITE(6,100) WA3COS,ETABCF,DTCOCF
100  FORMAT(17H0COMBUSTOR DESIGN,7X8H WA3COS=,E15.8,8H ETABCF=,E15.8,
18H DTCOCF=,E15.8)
      CALL COMPTB
      RETURN
      END

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SUBROUTINE COMPTB
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX,IOBURN,IAFTBN,IOCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOFAP,NUMMAP,MAPEOG,IOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFOS ,PCNFCS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNICS ,PRIOS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAIICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFEDS ,DTGDS ,ETABDS ,WA3CDS ,DPCDS ,DTCCF ,ETABCF ,
6TFHPOS ,CNHPDS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHPHCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPOS ,CNLPDS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,CTOUDS ,ETAADS ,WA23DS ,DPOUDS ,DTUOCF ,ETADCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVHNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIF ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLLPC ,CS ,
CHG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC ,PCBLGBI ,PCBLGBC ,PCBLHFI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBLLLPI ,
GPCBLLLFC
COMMON /HTLXB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLO,WHI/6HCOMPTB,6H (LC) ,6H (HI) /
WORD=AWORD
IF (IDES.EQ.0) GO TO 1
CNHPCF=CNHPDS*SQRT(T4)/PCNC
1 CNHP=CNHPCF*PCNC/SQRT(T4)
CNHPS=CNHP
TFFHPS=TFFHP
CALL SEARCH(-1.,TFFHP,CNHP,DHTCHP,ETATHP,
1TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,I.C
IF (IGO.EQ. 1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFHPS,WL
IF (IGO.EQ. 2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFHPS,WHI
IF (IGO.EQ.10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000) CNHPS,WLO
IF (IGO.EQ.20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000) CNHPS,WHI
1000 FORMAT(19H0****TFFHP OFF MAP,F10.4,2XA6,11H*****$)
2000 FORMAT(19H0**** CNHP OFF MAP,F10.4,2XA6,11H*****$)
IF (IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

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3   MAPGO=0
    IF (ABS(TFFHPS-TFFHP).LE.0.001*TFFHPS) GO TO 4
    MAPGO=1
    IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGO=3
    GO TO 5
4   IF (ABS(CNHPS-CNHP).GT.0.001*CNHPS) MAPGO=2
5   IF (MAPGO.GT.0) CALL MAPBAC(1,MAFGC,TFFHPS,TFFHP,CNHPS,CNHP,PCNC,
1  T4,MODE,NOMAP,NUMMAP)
    IF (NOMAP.GT.0) RETURN
    TFHCAL=WG4*SQRT(T4)/(14.696*P4)
    BTUEXT=0.706705*MPEXT
    DHTCC=(BTUEXT+WAC*(H3-H22))/(WG4*T4)
    IF (IDES.EQ.0) GO TO 6
    TFHPCF=TFHPDS/TFHCAL
    DHMPCF=DHTCC/DHTCHP
    ETHPCF=ETHPDS/ETATHP
    WRITE(6,102) CNHPCF,TFHPCF,ETHPCF,DHMPCF
102  FORMAT(20H0H.P. TURBINE DESIGN,5X7HCNHPCF=,E15.8,8H TFHPCF=,E15.8,
1  8H ETHPCF=,E15.8,8H DHMPCF=,E15.8)
6   TFHCAL=TFHPCF*TFHCAL
    DHTCHP=DHMPCF*DHTCHP
    ETATHP=ETHPCF*ETATHP
    DHTC=DHTCC*T4
    ERR(1)=(TFHCAL-TFFHP)/TFHCAL
    ERR(2)=(DHTCC-DHTCHP)/DHTCC
    CALL THTURB(DHTC,ETATHP,FAR4,H4,S4,P4,T45,H45,S45,P45)
    IF (BLHP.LE.0.) GO TO 7
    FAR45=WFB/(WA3+BLHP)
    WG45=WG4+BLHF
    H45=(BLHPI*H22+BLHPC*H3+WG4*H45)/WG45
    CALL THERMC(F45,H45,T45,S45,XX2,1,FAR45,1)
    GO TO 8
7   FAR45=FAR4
    WG45=WG4
8   CALL COIPTB
    RETURN
    END

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```

SUBROUTINE CCPTB
COMMON/ ALL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASHX,IOBURN,IAFTBN,ICD ,IMCD ,IOSHOC,IMSHOC,NOZFLT,
3ITRYS,LOCPER,NOMAP,NUMMAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WF8DS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
6TFHPOS ,CNHPOS ,ETHPOS ,TFHPCF ,CAHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPOS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,OTCJDS ,ETAADS ,WA23DS ,OPDUDS ,OTDUCF ,ETADCF ,
AT7DS ,WFAADS ,DTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCOM ,WG4 ,
9CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BL LPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLOUC,PCELCBI,PCBL0BC,PCBLHPI,PCBLHPC,PCBLIPI,PCBLIPC,PCBL LPI,
GPCBL LPC
COMMON/ITURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLO,WHI/6HCOIPTB,6H (LC) ,6H (HI) /
WORD=AWORD
IF (IDES.EQ.0) GO TO 1
CN_LPCF=CNIFDS*SQRT(T45)/PCNI
1 CNIP=CNIPCF*PCNI/SQRT(T45)
CNIPS=CNIP
TFFIPS=TFFIP
CALL SEARCH(-1.,TFFIP,CNIP,DHTCIP,ETATIP,
1TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGO
IF (IGO.EQ. 1.OR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFIPS,WLO
IF (IGO.EQ. 2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFIPS,WHI
IF (IGO.EQ.10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000) CNIPS,WLO
IF (IGO.EQ.20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000) CNIPS,WHI
1000 FORMAT(19H0****TFFIP OFF MAP,F10.4,2XA6,11H****$)
2000 FORMAT(19H0**** CNIP OFF MAP,F10.4,2XA6,11H****$)
IF (IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

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```

3   MAPGO=0
    IF (ABS(TFFIPS-TFFIP).LE.0.001*TFFIPS) GO TO 4
    MAPGO=1
    IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=3
    GO TO 5
4   IF (ABS(CNIPS-CNIP).GT.0.001*CNIPS) MAPGO=2
5   IF (MAPGO.GT.0) CALL MAPBAC(3,MAPGC,TFFIPS,TFFIP,CNIPS,CNIP,PCNI,
1T4,MODE,NOMAP,NUMMAP)
    IF (NOMAP.GT.0) RETURN
    TFICAL=WG45*SQRT(T45)/(14.696*P45)
    DHTIC=(WAI*(H22-H21))/(WG45*T45)
    IF (IDES.EG.0) GO TO 6
    TFIPCF=TFIPDS/TFICAL
    DHIPCF=DHTIC/DHTCIP
    ETIPCF=ETIPDS/ETATIP
    WRITE(6,102)CNIPCF,TFIPCF,ETIPCF,DHIPCF
102 FORMAT(20H01.P. TURBINE DESIGN,5X7HCNIPCF=,E15.8,8H TFIPCF=,E15.
10H ETIPCF=,E15.8,8H DHIPCF=,E15.8)
6   TFICAL=TFIPCF*TFICAL
    DHTCIP=DHIFCF*DHTCIP
    ETATIP=ETIPCF*ETATIP
    DHTI=DHTIC*T45
    ERR(8)=(TFICAL-TFFIP)/TFICAL
    ERR(9)=(DHTIC-DHTCIP)/DHTIC
    CALL THTURB (DHTI,ETATIP,FAR45,H45,S45,P45,T5,H5,S5,P5)
    IF (BLIP.LE.0) GO TO 7
    FAR5=WF8/(WA3+8LHP+BLIP)
    WG5=WG45+8LIP
    H5=(BLIPI*H22+8LIPC*H3+WG45*H5)/WG5
    CALL THERMO(P5,H5,T5,S5,XX2,1,FAR5,1)
    GO TO 8
7   FAR5=FAR45
    WG5=WG45
8   CALL COLPT2
    RETURN
    END

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```

SUBROUTINE COLPTB
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,IMSHOC ,NGZFLT ,
3ITRYS ,LOOPER ,HOMAP ,NUMMAP ,HAPEDG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFOS ,PCNFOS ,PRFOS ,ETAFOS ,NAFOS ,PRFCF ,ETAFCF ,NAFCF ,
3ZIDS ,PCNICS ,PRIOS ,ETAIDS ,NAIDS ,PRICF ,ETAICF ,NAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,NACDS ,PRCCF ,ETACCF ,NACCF ,
5T4DS ,WFEDS ,OTCDS ,ETABDS ,HA3CDS ,OPCDS ,OTCCCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,OHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,OHIPCF ,T2IDS ,
8TFLPDS ,CNLPDS ,ETLPS ,TFLPCF ,CNLPCF ,ETLPCF ,OHLPCF ,T2DS ,
9T24DS ,WFODS ,OTDUDS ,ETAADS ,HA23DS ,OPDUDS ,OTDUCF ,ETAACF ,
AT7DS ,WFAADS ,OTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,OTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDW02 ,CVMW02 ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,NAFC ,NAF ,BLOUI ,BLOUC ,BLOB ,
7CNI ,PRI ,ETAI ,NAIC ,NAI ,BLOGI ,BLOBC ,HA3 ,
8CNC ,PRC ,ETAC ,NACC ,NAC ,ETAB ,OPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPPI ,BLLPC ,CS ,
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EWF8 ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLG ,PCBLDUI ,
FPCBLDUC ,PCBLOBI ,PCBLOBC ,PCBLHPI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLLPPI ,
GFCBLLPC
COMMON /LTURB/TFFX(15),CNX(15,15),DHTCX(15,15),ETATX(15,15),
1NTFFS,NPTTFF(15)
DATA AWORD,WLO,WHI/6HCOLPTB,6H(LC),6H(HI)/
WORD=AWORD
IF (IDES.EQ.0) GO TO 1
CNLPCF=CNLPDS*SQRT(T5)/PCNF
1 CNLP=CNLPCF*PCNF/SQRT(T5)
CNLPS=CNLP
TFFLPS=TFFLP
CALL SEARCH(-1.,TFFLP,CNLP,DHTCLP,ETATLP,
1TFFX(1),NTFFS,CNX(1,1),DHTCX(1,1),ETATX(1,1),NPTTFF(1),15,15,IGO)
IF (IGO.EQ. 1.CR.IGO.EQ.11.OR.IGO.EQ.21) WRITE(8,1000)TFFLPS,WLO
IF (IGO.EQ. 2.OR.IGO.EQ.12.OR.IGO.EQ.22) WRITE(8,1000)TFFLPS,WHI
IF (IGO.EQ. 10.OR.IGO.EQ.11.OR.IGO.EQ.12) WRITE(8,2000) CNLPS,WLO
IF (IGO.EQ. 20.OR.IGO.EQ.21.OR.IGO.EQ.22) WRITE(8,2000) CNLPS,WHI
1000 FORMAT(19H0****TFFLP OFF MAP,F10.4,2XAG,11H*****$$$$)
2000 FORMAT(19H0**** CNLP OFF MAP,F10.4,2XAG,11H*****$$$$)
IF (IGO.NE.7) GO TO 3
2 CALL ERROR
RETURN

```

```

3     MAPGO=0
      IF (ABS(TFFLPS-TFFLP).LE.0.001*TFFLPS) GO TO 4
      MAPGO=1
      IF (ABS(CNLPS-CNLP).GT.0.001*CNLPS) MAPGO=3
      GO TO 5
4     IF (ABS(CNLPS+CNLP).GT.0.001*CNLPS) MAPGO=2
5     IF (MAPGO.GT.0) CALL MAPBAC(2,MAPGO,TFFLPS,TFFLP,CNLPS,CNLP,PCNF,
1    T4,MODE,NOMAP,NUMMAP)
      IF (NOMAP.GT.0) RETURN
      TFLCAL=NG5*SQRT(T5)/(14.696*P5)
      DHTCF=MAF*(H21-H2)/(NG5*T5)
      IF (IDES.EQ.0) GO TO 6
      TFLPCF=TFLPDS/TFLCAL
      CHLPCF=DHTCF/DHTCLP
      ETLPCL=ETLPDS/ETATLP
102    WRITE(6,102)CNLPCF,TFLPCF,ETLPCL,CHLPCF
      FORMAT(20HOL.P. TURBINE DESIGN,5X7HCNLPCF=,E15.8,8H TFLPCF=,E15.8,
1    8H ETLPCL=,E15.8,8H DHLPCF=,E15.8)
6     TFLCAL=TFLPCF*TFLCAL
      DHTCLP=DHLPCF*DHTCLP
      ETATLP=ETLPCL*ETATLP
      DHTF=DHTCF*T5
      ERR(3)=(TFLCAL-TFFLP)/TFLCAL
      ERR(4)=(DHTCF-DHTCLP)/DHTCF
      CALL THTURB(DHTF,ETATLP,FAR5,H5,S5,P5,T55,H55,S55,P55)
      IF (BLLP.LE.0.) GO TO 7
      FAR55=MF8/(MA3+R1*HP+BLLP*BLLP)
      NG55=NG5+BLLP
      H55=(BLLP1*H22+BLLP2*H3+NG5*H55)/NG55
      CALL THERMO(P55,H55,T55,S55,XX2,1,FAR55,1)
      GO TO 8
7     FAR55=FAR5
      NG55=NG5
8     CALL FRTOSC
      RETURN
      END

```

SUBROUTINE FATOSD  
COMMON/ FRONT/

1 T1	, F1	, H1	, S1	, T2	, P2	, H2	, S2	,
2 T21	, F21	, H21	, S21	, T22	, P22	, H22	, S22	,
3 T3	, P3	, H3	, S3	, T4	, P4	, H4	, S4	,
4 T45	, P45	, H45	, S45	, T5	, P5	, H5	, S5	,
5 T55	, P55	, H55	, S55	, BLF	, BLI	, BLC	, BLOU	,
6 CNF	, PRF	, E1AF	, WAF	, KAF	, BLDUI	, BLDUC	, BLOB	,
7 CNX	, PRI	, ETAI	, WAI	, BLOBI	, BLOB	, WAI3	,	,
8 CN	, PR	, ETAC	, WAC	, WAC	, ETAB	, OPCOM	, W4	,
9 CNHP	, ETATHP	, DHTCHP	, DHTC	, BLHP	, BLHPI	, BLHPC	, FAR4	,
ACNIP	, ETATIP	, DHTCIP	, DHTI	, BLIP	, BLIPI	, BLIPC	, DUMF	,
BCNLP	, ETATLP	, DHTCLP	, DHTF	, BLLP	, BLLPI	, BLLPC	, CS	,
CHG45	, FAR45	, W5	, FAR5	, W55	, FAR55	, HPEXT	, AM	,
DLTF	, ETAR	, ZF	, PCNF	, ZI	, PCNI	, ZC	, PCNC	,
EWFB	, TFFIP	, TFFIP	, TFFLP	, PCBLF	, PCBLI	, PCBLC	, PCBLDUI	,
FFCBLDUC, PCBL0BI, PCBL0BC, PCBLHPI, PCBLHPC, PCBLIPI, PCBLIPC, PCBLIPI,								
GPCBLIPC								

COMMON/ SIDE/

1 XP1	, XWAF	, XWAI	, XWAC	, XBLF	, XBLOU	, XBLOUI	, XBLOUC	,
2 XH22	, XH3	, XT21	, XP21	, XH21	, XS21	, DUMS1	, DUMS2	,
3 T23	, P23	, H23	, S23	, T24	, P24	, H24	, S24	,
4 T25	, P25	, H25	, S25	, T26	, P26	, H26	, S26	,
5 T29	, P29	, H29	, S29	, DUMS3	, DUMS4	, DUMS5	, DUMS6	,
6 XAD	, WFL	, W24	, FAR24	, ETAD	, DPOUC	, BYPASS	, DUMS7	,
7 TS29	, PS28	, V28	, AM28	, TS29	, PS29	, V29	, AM29	,

XP1=P1

XWAF=WAF

XWAI=WAI

XWAC=WAC

XBLF=ELF

XBLOU=BLOU

XBLOUI=BLOUI

XBLOUC=BLOUC

XH22=H22

XH3=H3

XT21=T21

XP21=P21

XH21=H21

XS21=S21

CALL CODUCT

RETURN

END

```

SUBROUTINE CODUCT
COMMON / ALL/
1WORD ,IDES ,JOES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASM ,IDBURN ,IAFTBN ,IQCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOMAP ,NUMMAP ,MAPEOG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,GUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIGS ,PRIOS ,ETAIDS ,WAIOS ,PRICF ,ETAICF ,WAIICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFEDS ,DTGDS ,ETABDS ,WA3CDS ,DPCDS ,DTCCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T20S ,
7TFIPDS ,CNIFDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIFCF ,T210S ,
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T220S ,
9T24DS ,WFCDS ,DTGDS ,ETADDS ,WA23DS ,DPDUS ,DTGUCF ,ETADCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAAOS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVHNOZ ,A9SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ SIDE/
1 P1 , WAF , WAI , WAC , ELF , BLDU , BLDUI , BLDUC ,
2 H22 , H3 , T21 , P21 , H21 , S21 , DUMS1 , DUMS2 ,
3 T23 , P23 , H23 , S23 , T24 , P24 , H24 , S24 ,
4 T25 , P25 , H25 , S25 , T28 , P28 , H28 , S28 ,
5 T29 , P29 , H29 , S29 , DUMS3 , DUMS4 , DUMS5 , DUMS6 ,
6 WAD , WFC , WG24 , FAR24 , ETAD , DPOUC , BYPASS , DUMS7 ,
7 TS28 , PS28 , V28 , AM28 , TS29 , PS29 , V29 , AM29
DIMENSION Q(9)
DATA AWORD1 , AWORD2 / 6HCODUCT , 6HONNOZZL /
WORD = AWORD1
Q(2) = 0.
Q(3) = 0.
WAX = WAF - WAI - ELF
WAD = WAX + BLDU
P23 = P21
H23 = (BLDUI * H22 + BLDUC * H3 + WAX * H21) / WAD
CALL THERMC(P23 , H23 , T23 , S23 , XX2 , 1 , 0.0 , 1)
BYPASS = (WAF - WAI) / WAI
WA23C = WAD * SORT(T23) / P23
IF (IDES.EQ.1) WA23DS = WA23C
DPOUC = DPDUS * (WA23C / WA23DS)
IF (DPOUC.GT.1.) DPOUC = 1.
P24 = P23 * (1. - DPOUC)
IF (IGASM.GT.0) IDBURN = 0
IF (IDBURN.NE.0) GO TO 2
T24 = T23
WFD = 0.
FAR24 = 0.
GO TO 1
2 IF (IDBURN.EQ.2) T24 = T23 + 2000.
3 IF (T24.GT.4000.) T24 = 4000.
IF (T24.LT.T23) T24 = T23
C *** IF DESIRED, ENTER CALCULATIONS FOR ETAD HERE
HV = (((((- .4594317E-19 * T24) - .2034116E-15) * T24 + .2783643E-11) * T24 +
1.2051501E-07) * T24 - .2453116E-03) * T24 - .9433296E-01) * T24 + .1845537E+
CALL THERMC(P24 , WA , T24 , XX1 , XX2 , 0 , 0.0 , 0)

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```

FAR24=(HA-H23)/(HV*ETA0)
IF (FAR24.LT.0.) FAR24=0.
WFOX=FAR24*WAD
IF (IDBURN.NE.2) GO TO 6
ERRW=(WFO-WFOX)/WFO
DIR=SQRT(WFO/WFOX)
CALL AFQUIR(Q(1),T24,ERRW,0.,20.,0.0001,DIR,T24T,IGO)
GO TO (4,7,5),IGO
4 T24=T24T
GO TO 3
5 CALL ERROR
6 WFO=WFOX
7 CALL THERMO(P24,H24,T24,S24,XX2,1,FAR24,0)
WG24=WFO+WAD
IF (IDES.EQ.1) WRITE(6,101) WA23DS
101 FORMAT(12HODUCT DESIGN,12X8H WA23DS=,E15.8)
*** IF DESIRED, ENTER OTHER LOSSES HERE
T25=T24
P25=P24
H25=H24
S25=S24
IF (IGASM.XGT.0) GO TO 11
WORD=AWORD2
A28SAV=A28
A29SAV=A29
NOZD=0
IDNOZ=0
IF (NOZFLT.EQ.2.OR.NOZFLT.EQ.3) NOZD=1
IF (IDES.EQ.1.OR.IDBURN.GT.0.OR.NOZD.EQ.1) IDNOZ=1
IF (IDCD.EQ.1) GO TO 8
CALL CONVRG(T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28, P25R,
1 T28,H28,P28,S28,TS28,PS28,V28,AM28,ICON)
GO TO (9,9,3,5),ICON
CALL CONDV(T25,H25,P25,S25,FAR24,WG24,P1,IDNOZ,A28,A29,P25R,
1 T28,H28,P28,S28,T29,H29,P29,S29,TS28,TS29,PS28,PS29,V28,V29,A
2 AM29,ICON)
IDSHOC=ICON
GO TO (10,10,10,5),ICON
9 T29=T28
H29=H28
P29=P28
S29=S28
TS29=TS28
PS29=PS28
V29=V28
AM29=AM28
A29=A28
IDSHOC=ICON+3
10 ERR(5)=(P25R-P25)/P25R
IF (IDNOZ.EQ.1) WRITE(6,100) A28,AM28,A29,AM29
100 FORMAT(19HODUCT NOZZLE DESIGN,5X8H A28=,E15.8,8H AM28=,E1
18H A29=,E15.8,8H AM29=,E15.8)
11 CALL FASTBK
RETURN
END

```

SUBROUTINE FASTBK  
COMMON/ FRCNT/

1T1	,P1	,H1	,S1	,T2	,P2	,H2	,S2	,
2T21	,P21	,H21	,S21	,T22	,P22	,H22	,S22	,
3T3	,P3	,H3	,S3	,T4	,P4	,H4	,S4	,
4T45	,P45	,H45	,S45	,T5	,P5	,H5	,S5	,
5T55	,P55	,H55	,S55	,BLF	,BLI	,BLC	,BLOU	,
6CNF	,PRF	,ETAF	,WAF	,WAF	,BLOUI	,BLOUC	,BLO8	,
7CNI	,PRI	,ETAI	,WAI	,WAI	,BLOBI	,BLOBC	,WA3	,
8CNC	,PRC	,ETAC	,WACC	,WAC	,ETAB	,DPCOM	,WG4	,
9GNHP	,ETATHP	,DHTCHP	,DHTC	,BLHP	,BLHPI	,BLHPC	,FAR4	,
ACNIP	,ETATIP	,DHTCIP	,DHTI	,BLIP	,BLIPI	,BLIPC	,DUMF	,
BCNLP	,ETATLP	,DHTCLP	,DHTF	,BLLP	,BLLPI	,BLLPC	,CS	,
CHG45	,FAR45	,WG5	,FAR5	,WG55	,FAR55	,HPEXT	,AM	,
DALTP	,ETAR	,ZF	,PCNF	,ZI	,PCNI	,ZC	,PCNC	,
EWFB	,TFFHF	,TFFIP	,TFFLP	,PCBLF	,PCBLI	,PCBLC	,PCBLDUI	,
FPCBLOUC	,PCBLOBI	,PCBLOBC	,PCBLHPI	,PCBLHPC	,PCBLIPI	,PCBLIPC	,PCBLLPI	,
GPCBLIFC								

COMMON/ SIDE/

1XP1	,XWAF	,XWAI	,XWAC	,XBLF	,XBLOU	,XBLOUI	,XBLOUC	,
2XH22	,XH3	,XT21	,XP21	,XH21	,XS21	,DUMS1	,DUMS2	,
3T23	,P23	,H23	,S23	,T24	,P24	,H24	,S24	,
4T25	,P25	,H25	,S25	,T28	,P28	,H28	,S28	,
5T29	,P29	,H29	,S29	,DLMS3	,DUMS4	,DUMS5	,DUMS6	,
6WAD	,WFD	,WG24	,FAR24	,ETAD	,DPOUC	,BYPASS	,DUMS7	,
7TS28	,PS28	,V28	,AM28	,TS29	,PS29	,V29	,AM29	,

COMMON / EACK/

XXT55	,XP55	,XH55	,XS55	,XT25	,XP25	,XH25	,XS25	,
XXWFB	,XWG55	,XFAR55	,XWFD	,XWG24	,XFAR24	,XXP1	,DUMB	,
3T6	,P6	,H6	,S6	,T7	,P7	,H7	,S7	,
4T8	,P8	,H8	,S8	,T9	,P9	,H9	,S9	,
5WG6	,WFA	,WG7	,FAR7	,ETAA	,OPAFT	,V55	,V25	,
6PS6	,V6	,AM6	,TS7	,PS7	,V7	,AM7	,AM25	,
7TS8	,PS8	,V8	,AM8	,TS9	,PS9	,V9	,AM9	,
8VA	,FRD	,VJD	,FGMD	,VJM	,FGMM	,FGPD	,FGPM	,
9FGM	,FGP	,WFT	,WGT	,FART	,FG	,FN	,SFC	,

XT55=T55

XP55=P55

XH55=H55

XS55=S55

XT25=T25

XP25=P25

XH25=H25

XS25=S25

XWFB=WFB

XWG55=WG55

XFAR55=FAR55

XWFD=WFD

XWG24=WG24

XFAR24=FAR24

XXP1=P1

CALL COMIX

RETURN

END

```

SUBROUTINE COMIX
DIMENSION QQ(9)
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IAMTP ,
2IGASMX,IDBURN,IAFTBN,IDCD ,IMCD ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOMAP,NUMMAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIDS ,ETAIDS ,WAFDS ,PRICF ,ETAICF ,WAFCF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFDS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
6TFHPOS ,CNHPOS ,EYHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,OHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,OHIPCF ,T2IDS ,
8TFLPOS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,OHLPCF ,T22DS ,
9T24DS ,WFCDS ,OTDUDS ,ETAADS ,WA23DS ,OPDUDS ,OTDUCF ,ETAOCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/FRONT/DUMF1(98),ZF,PCNF,DUMF2(21)
COMMON / BACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WF8 ,WG55 ,FAR55 ,WFO ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WG7 ,FART ,FG ,FN ,SFC
DATA AWORD/6H COMIX/
WORD=AWORD
AJ=778.26
CAPSF=2116.2170
G=32.174049
CALL PROCOM(FAR55,T55,XX1,XX2,XX3,XX4,PHI55,XX5)
CALL PROCOM(FAR24,T25,XX1,XX2,XX3,XX4,PHI25,XX5)
IF (IDES.EQ.0) GO TO 6
C *** CALCULATE A55 AND A25 WITH PS25=PS55
IF (PS55.EQ.0.) GO TO 50
TS55=T55*(PS55/P55)**0.286
DO 1 I=1,15
CALL PROCOM(FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)
PHIS=PHI55-REX55*ALOG(P55/PS55)
DELPHI=PHIS-PHIS55
IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 3
1 TS55=TS55*EXP(4.0*DELPHI)
2 CALL ERROR
RETURN
50 TS55=0.875*T55
DO 51 I=1,15
CALL PROCOM(FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)
V55=AM55*CS55
HSCAL=H55-V55**2/(2.*G*AJ)
DELHS=HSCAL-HS55

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51 IF (ABS(DELHS).LE.0.0055*HSCAL) GO TO 52
   TS55=TS55+DELHS/CP55
   GO TO 2
52 PS55=P55/EXP((PHI55-PHIS55)/REX55)
   3 IF (H55.GT.HS55) GO TO 53
   WRITE(8,101)P55,PS55,T55,TS55,H55,HS55
101 FORMAT(22H0SQRT OF H55-HS55 NEG ,6E15.6,6H$$$$$$)
   CALL ERROR
53 V55=SQRT(2.*G*AJ*(H55-HS55))
   RHO=CAPSF*PS55/(AJ*REX55*TS55)
   A55=HG55/(RHO*V55)
   AM55=V55/CS55
   IF (IGASHX.GT.0) GO TO 54
   WRITE(6,104)A55,AM55
104 FORMAT(20HOTURBINE AREA DESIGN,6X6H A55=,E15.8,8H AM55=,E15.8
   GO TO 34
54 PS25=PS55
   TS25=T25*(PS25/P25)**0.286
   DO 4 I=1,15
   CALL FROCOM(FAR24,T325,CS25,AK25,CP25,REX25,PHIS25,HS25)
   PHIS=PHI25-REX25*ALOG(P25/PS25)
   DELPHI=PHIS-PHIS25
   IF (ABS(DELPHI).LE.0.0001*PHIS) GO TO 5
   4 TS25=TS25*EXP(4.0*DELPHI)
   GO TO 2
   5 IF (H25.GT.HS25) GO TO 55
   WRITE(8,102)P25,PS25,T25,TS25,H25,HS25
102 FORMAT(22H0SQRT OF H25-HS25 NEG ,6E15.6,6H$$$$$$)
   CALL ERROR
55 V25=SQRT(2.*G*AJ*(H25-HS25))
   RHO=CAPSF*PS25/(AJ*REX25*TS25)
   A25=HG24/(RHO*V25)
   AM25=V25/CS25
   WRITE(6,100)A55,AM55,A25,AM25
100 FORMAT(25HOTURBINE/DUCT AREA DESIGN,7H A55=,E15.8,
10H AM55=,E15.8,8H A25=,E15.8,8H AM25=,E15.8)
   GO TO 20
C *** CALCULATE PS55 AND PS25
6 WQA=HG55/A55
   C1=P55*SQRT(G/(T55*AJ))*CAPSF
   MCON=0
   QO(2)=0.
   QO(3)=0.
   AM55=0.50
   TS55=0.875*T55
   7 DO 8 I=1,15
   CALL FROCOM(FAR55,TS55,CS55,AK55,CP55,REX55,PHIS55,HS55)
   V55=AM55*CS55
   HSCAL=H55-V55**2/(2.*G*AJ)
   DELHS=HSCAL-HS55
   IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 9
   8 TS55=TS55+DELHS/CP55
   GO TO 2
   9 WQAT=C1*SQRT(AK55/REX55)*AM55/(1.+(AK55-1.)*AM55**2/2.)*
1 ((AK55+1.)/(2.*(AK55-1.)))

```

```

      AMX=AM55
      IGOGO=0
10    DIR=WQA/WQAT
      EH=(WQA-WQAT)/WQA
      CALL AFQUIR(QQ(1),AMX,EH,0.,30.,0.0005,DIR,AMXT,ICON)
      GO TO (11,15,2),ICON
11    IF(AMXT.LE.1.0) GO TO 13
      AMXT=0.7
      MCON=MCON+1
      IF(MCON.LE.1) GO TO 13
      IF(MOCE.EQ.3) GO TO 120
      WRITE(8,103)FCNF,AMX,P55,PS55,P25,PS25
103   FORMAT(12H0COMIX PCNF=,F7.4,4H AM=,F8.6,5H P55=,F9.5,
16H PS55=,F9.5,5H P25=,F9.5,6H PS25=,F9.5,6H$$$$$$)
      PCNF=1.01*PCNF
12    NOMAP=7
      RETURN
120   WRITE(8,121)ZF,AMX,P55,PS55,P25,PS25
121   FORMAT(10H0CMIX ZF=,F8.5,4H AM=,F8.6,5H P55=,F9.5,
16H PS55=,F9.5,5H P25=,F9.5,6H PS25=,F9.5,6H$$$$$$)
      ZF=0.99*ZF
      GO TO 12
13    IF(IGOGO.EQ.1) GO TO 14
      AM55=AMXT
      GO TO 7
14    AM25=AMXT
      GO TO 16
15    IF(IGOGO.EQ.1) GO TO 19
      PS55=P55/EXP((PHI55-PHIS55)/REX55)
      IF(IGASHX.LE.0) GO TO 34
      WQA=WG24/A25
      C1=P25*SQRT(G/(T25*AJ))*CAPSF
      MCON=0
      QQ(2)=0.
      QQ(3)=0.
      AM25=0.25
      TS25=0.875*T25
16    DO 17 I=1,15
      CALL PROCOM(FAR24,TS25,CS25,AK25,CP25,REX25,PHIS25,HS25)
      V25=AM25*CS25
      HSCAL=H25-V25**2/(2.*G*AJ)
      DELHS=HSCAL-HS25
      IF(ABS(DELHS).LE.0.0005*HSCAL) GO TO 18
17    TS25=TS25+DELHS/CP25
      GO TO 2
18    WQAT=C1*SQRT(AK25/REX25)*AM25/(1.+(AK25-1.)*AM25**2/2.)*
1((AK25+1.)/(2.*(AK25-1.)))
      AMX=AM25
      IGOGO=1
      GO TO 10
19    PS25=P25/EXP((PHI25-PHIS25)/REX25)
2)   WG6=WG24+WG55
      ERR(5)=(PS25-PS55)/PS25
      WF6=WFD+WF0
      FAR6=WF6/(WG6-WF6)

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```

H6=(HG24*H25+HG55*H55)/H65
CALL THERMO(1.,H5,T6,PHI6,AP: 1,FARL 1)
C1=PS55*AS5*(1.+AK55*AM55**2)+PS25*A25*(1.+AK25*AM25**
TS6=0.833*TE
DO 25 I=1,15
CALL FROGOP(FAR6,TS6,CS6,AK6-CP6,REX6,PHIS6,HS6)
C2=HG6*SQRT(AJ*REX6*TE/(AK6*U))
C3=C2/(CAPSF*C1)
C4=(AK6-1.)/2.-(C3*AK6)**2
C5=1.-2.*AK6*C3**2
C6=C5**2+4.*C4*C3**2
IF (C6)21,22,23
21 CALL ERROR
RETURN
22 AM62G=-C5/(2.*C4)
GO TO 24
23 AM62G=(SQRT(C6)-C5)/(2.*C4)
24 IF (AM62G.LE.0) GO TO 21
AM6G=SQRT(AM62G)
V6=AM6G*CS6
HSCAL=H6-V6**2/(2.*G*AJ)
DELHS=HSCAL-HS6
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 26
2: TS6=TS6+DELHS/CP6
GO TO 21
26 IF (IGASHX,CT.0) A6G=A25+A55
C7=SQRT(1.+(AK6-1.)*AM62G/2.)
PS6=C2/(CAPSF*A6G*AM6G*C7)
F6=PS6*EXP((PHI6-PHIS6)/REX6)
CALL THERMO(P6,H6,T6,S6,XX1,1,FAR6,C)
S6AVE=(HG24*S25+HG55*S55)/HG6
IF (S6-GE.S6AVE) GO TO 27
S6=S6AVE
P6=EXP(AMX*(PHI6-S6)/1.986375)
27 IF (IGASN,EG.1) GO TO 35
IF (IDES,EG.0) GO TO 30
C *** CALCULATE A6 AS A FUNCTION OF INPUT AM6
TS6P=TE/(1.0+(((AK6-1.0)/2.0)*AM6**2.))
DO 28 JJ=1,15
CALL FROGOP(FAR6,TS6P,CS6,AK6-CP6,REX6,PHIS6,HS6)
V6=AM6*CS6
HSCAL=H6-V6**2/(2.*G*AJ)
DELHS=HSCAL-HS6
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 2
28 TS6=TS6P+DELHS/CP6
GO TO 28
29 PS6=P6/(1.0+(((AK6-1.0)/2.0)*AK6**2.))**((AK6/(AK6-1.0)).
AM6DS=AM6
RHO=CAPSF*PS6/(AJ*REX6*TS6)
AG=HG6/(RHO*V6)
WRITE(6,200) A6
200 FORMAT(1H, 3)AFTER:JUNNER ENTRANCE DESIGN AREA A6 ,F8.3)
GO TO 36
C *** CALCULATES AM6=F(A6DESIGN)
30 TS6P=TE/(1.0+(((AK6-1.0)/2.0)*AM6CS**2.))

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```

DO 32 I=1,15
CALL PROCOM (FAR6,TS6P,CS6,AK6,CP6,REX6,PHIS6,HS6)
PS6P=PS6*(TS6P/TS6)**(AK6/(AK6-1.0))
RHO=CAPSF*PS6P/(AJ*REX6*TS6P)
IF (H6.GT.HS6) GO TO 31
WRITE (8,201)P6,PS6P,T6,TS6P
201  FORMAT(20H0SORT OF H6-HS6 NEG ,6E15.6,6H$$$$$)
CALL ERROR
31  V6=SQRT(2.*G*AJ*(H6-HS6))
    A6P=HG6/(RHO*V6)
    DELA6=A6P-A6
    V6=HG6/(RHO*A6)
    AM6=V6/CS6
    AM62=AM6**2.
    IF (ABS(DELA6).LE.0.002*AF) GO TO 33
32  TS6P=T6/(1.0+((AK6-1.0)/2.0)*AM62)
    GO TO 21
33  TS6=TS6P
    PS6=PS6P
    GO TO 36
34  T6=T55
    P6=P55
    H6=H55
    S6=S55
    HG6=HG55
    PS6=PS55
    V6=V55
    AM6=AM55
    IF (IGASHX.EQ.0) A6=A55
    GO TO 36
35  AM62=AM62G
    AM6=AM6G
    A6=A25+A55
36  CALL COAFBN
    RETURN
    END

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SUBROUTINE COAFBN
COMMON / ALL/
1WORD ,IDES ,JOES ,KUES ,MODE ,INIT ,IDUMP ,IAHPT ,
2IGASHX,ICBURN,IAFTBN,IDCN ,IMCO ,IDSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUMHAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC
2ZFOS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF
3ZIOS ,PCNICS ,PRI , ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF
4ZCOS ,PCACDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF
5T4OS ,WF8OS ,OTCDS ,ETABOS ,WA3CDS ,OPCDS ,DTCOCF ,ETABCF
6TFHPDS ,CNHPDS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2OS
7TFIPDS ,CNIPCS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21OS
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22OS
9T24DS ,WF0DS ,OTODS ,ETAODS ,WA23DS ,OPDUDS ,DTODUCF ,ETAODCF
AT7DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A26 ,A29
CPS55 ,AM55 ,CVDNOZ ,CVHNOZ ,A8SAV ,A9SAV ,A26SAV ,A29SAV
COMMON / BACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6FS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DIMENSION C(9)
DATA AWORD/6HCOAFBN/
WORD=AWORD
C(2)=0.
C(3)=0.
AJ=778.26
CAPSF=2116.2170
G=32.174049
WF6=WFB
IF (IGASHX.GT.0) WF6=WF6+WFD
WA6=WG6-WF6
C *** CRY LOSS
WG6C=WG6*SCR1(T6)/P6
2 IF (IDES.EQ.1) WG6CDS=WG6C
DPAFT=DPAFDS*(WG6C/WG6CDS)
IF (DPAFT.GT.1.) DPAFT=1.
P7=P6*(1.-DPAFT)
A7=A6
FAR6=WFB/WAC
CALL PROCOM(FAR6,T6,XX1,XX2,XX3,XX4,PHI6,XX6)
WQA=WG6/A7
C1=P7*SQRT (G/(T6*AJ))*CAPSF
AM7=AM6
TS7=0.875*T6
20 DO 22 I=1,15
CALL PROCOM(FAR6,TS7,CS7,AK7,CP7,REX7,PHIS7,KS7)
V7=AM7*CS7

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HSCAL=H6-V7**2/(2.*G*AJ)
DELHS=HSCAL-HS7
IF (ABS(DELHS).LE.0.0005*HSCAL) GO TO 24
22 TS7=TS7+DELHS/CF7
GO TO 8
24 WQAT=C1*SQRT(AK7/REX7)*AM7/(1.+(AK7-1.)*AM7**2/2.)**
1 ((AK7+1.)/(2.*(AK7-1.)))
DIR=WQA/WQAT
EW=(WQA-WQAT)/WQA
CALL AFQUIR(Q(1),AM7,EW,0.,30.,0.0005,DIR,AM7T,IGO)
GO TO (26,28,8),IGO
26 AM7=AM7T
IF (AM7.GE.1.0) AM7=0.9
GO TO 20
28 PS7=P7/EXP((PHI6-PHI7)/REX7)
IF (IAFTBN.GT.0) GO TO 4
C *** NON-AFTERBURNING
3 T7=T6
WFA=0.0
FAR7=FAR6
WG7=WG6
GO TO 13
C *** AFTERBURNING
4 IF (IAFTBN.EQ.2) T7=T6+2000.
IF (T7.LE.T6) GO TO 3
RH065=CAPSF*PS7/(AJ*REX7*TS7)
PS65=PS7
V65=V7
Q(2)=0.
Q(3)=0.
5 IF (T7.GT.4000.) T7=4000.
C *** IF DESIRED, ENTER CALCULATIONS FOR ETAA HERE
HV=((((-0.4594317E-19*T7)-.2034116E-15)*T7+.2783643E-11)*T7
1+.2051501E-07)*T7-.2453116E-03)*T7-.9433296E-01)*T7+.1845537E+05
CALL THERMO(F7,HA,T7,XX1,XX2,0,0.0,0)
FAR7=(HA-H6)/(HV*ETAA)
IF (FAR7.GT.0.) GO TO 6
T7=T6
GO TO 5
6 WFAX=FAR7*WG6
IF (IAFTBN.EQ.1) GO TO 9
ERRW=(WFA-WFAX)/WFA
DIR=SQRT(WFA/WFAX)
CALL AFQUIR(Q(1),T7,ERRW,0.,20.,0.0001,DIR,T7T,IGO)
GO TO (7,10,8),IGO
7 T7=T7T
GO TO 5
8 CALL ERROR
9 WFA=WFAX
10 FAR7=(WF6+WFA)/WA6
WG7=WG6+WFA
C *** MOMENTUM LCSS
CALL PROCOF(FAR7,T7,XX1,XX2,XX3,REX7,PHI7,H7)
RH07=CAPSF*P7/(AJ*REX7*T7)
V7=WG7/(RHC7*A7)

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Q(2)=0.
Q(3)=0.
PS7=PS65-0.01
11  RH07=HG7/(V7*A7)
    HS7=H7-V7**2/(2.*G*AJ)
    CALL THERMO(1.0,HS7,TS7,PHIS7,XX2,1,FAR7,1)
    IF(TS7.GE.301.) GO TO 110
    CALL THERMC(1.0,HS7,400.,PHIS7,XX2,1,FAR7,0)
    V7=SQRT(2.*G*AJ*(H7-HS7))
    GO TO 11
110  PS7=RH07*AJ*REX7*TS7/CAPSF
    PS7A=PS65+(RH065*V65**2-RH07*V7**2)/(C    °SF)
    DIR=SQRT(ABS(PS7/PS7A))
    EP=(PS7-PS7A)/PS7
    CALL AFQUIR(Q(1),V7,EP,0.,50.,0.0005,DIR,V7T,IGO)
    V7=V7T
    IF(V7.LT.100.) V7=100.
    GO TO (11,12,8),IGO
12   P7=PS7*EXP((PHI7-PHIS7)/REX7)
    CALL FROCOM(FAR7,TS7,CS7,XX2,XX3,XX4,XX5,XX6)
    AM7=V7/CS7
13   CALL THERMO(P7,H7,T7,S7,XX2,1,FAR7,0)
    IF(IDES.EQ.1) WRITE(6,100) WG6CDS
100  FORMAT(19H0AFTERBURNER DESIGN,5X8H WG6CDS=,E15.8)
    CALL COMNOZ
    RETURN
    END

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SUBROUTINE COMNOZ
COMMON / ALL/
1WORD ,IDES ,JOES ,KOES ,MODE ,INIT ,IDUMP ,IATP ,
2IGASHY,IOBURN,IAFTBN,IDCD ,IMCD ,IOSHOC,INSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUHMAP,MAPEDG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DLMQ1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIOS ,PCNIOS ,PRIOS ,ETAIOS ,WAIOS ,PRICF ,ETAI CF ,WAI CF ,
4ZCOS ,PCNCOS ,PROCS ,ETACOS ,WACOS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFBOS ,DTCCDS ,ETABDS ,WAB3COS ,OPCCDS ,DTCCCF ,ETABCF ,
6TFHPOS ,CNHPOS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2OS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPOS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,DTCCDS ,ETACDS ,WAC3COS ,OPCCDS ,DTCCCF ,ETACCF ,
AT7DS ,WFAOS ,UTAFDS ,ETAADS ,WAG6COS ,DPAFDS ,DTAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVHNOZ ,AESAV ,A9SAV ,A28SAV ,A29SAV
COMMON / EACK/
1T55 ,P55 ,H55 ,S55 ,T25 ,P25 ,H25 ,S25 ,
2WFB ,WG55 ,FAR55 ,WFD ,WG24 ,FAR24 ,P1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6FS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DATA ANORD/6HMOZZL/
WORD=ANORD
AESAV=A8
A9SAV=A9
NOZM=0
IMNOZ=0
IF (NOZFLT.EQ.1.OR.NOZFLT.EQ.3) NOZM=1
IF (ICES.EQ.1.OR.IAFTBN.GT.0.OR.NOZM.EQ.1) IMNOZ=1
IF (IMCD.EQ.1) GO TO 1
CALL CONVRG(T7,H7,P7,S7,FAR7,WG7,F1,IMNOZ,A8, P7R,
1T8,H8,P8,S8,TS8,PS8,V8,AM8,ICON)
GO TO (3,3,3,2),ICON
1 CALL CONDIV(T7,H7,P7,S7,FAR7,WG7,F1,IMNOZ,A8,A9,P7R,
1T8,H8,P8,S8,T9,H9,P9,S9,TS8,TS9,PS8,PS9,V8,V9,AM8,AM9,ICON)
IMSHOC=ICON
GO TO (4,4,4,2),ICON
2 CALL ERROR
3 T9=T8
H9=H8
P9=P8
S9=S8
TS9=TS8
PS9=PS8
V9=V8
AM9=AM8
A9=A8
IMSHOC=ICON+3

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4      ERR(6)=(P7R-F7)/P7R
      IF (IMAGZ.EQ.1) WRITE(6,100) A8,AM8,A9,AM9
100    FORMAT(14HNOZZLE DESIGN,10X8H      A8=,E15.0,8H      AM8=,E15.8,
16H      A9=,E15.8,8H      AM9=,E15.8)
      RETURN
      END

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SUBROUTINE PERF
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
2IGASHX ,IDBURN ,IAFTBN ,IDCD ,IMCD ,IDSHOC ,INSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOHAP ,NUMMAP ,HAPEOG ,TOLALL ,ERR (9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,OLMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,FRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFEDS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
6TFHPDS ,CNHPCS ,ETHPCS ,TFHPCF ,CNHPCF ,ETHPCF ,DNHPCF ,T2DS ,
7YFIPDS ,CNIPCS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DNIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DNLPCF ,T22DS ,
9T24DS ,WFODS ,OTDUDS ,ETADDS ,WA23DS ,OPDUDS ,OTDUCF ,ETADCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,OPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AH55 ,CVNOZ ,CVHNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRCNT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLOUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOB ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,OPCCN ,WG4 ,
9CNHP ,ETATHF ,DHTCHP ,DHTC ,BLHP ,BLHFI ,BLHFC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIF ,DHT1 ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFHP ,TFIP ,TFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPC3LOUC ,PCBLCBI ,PCBLCBI ,PCBLHFI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLIPI ,
GPCBLIPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,X3LOU ,XBLDUI ,XBLDUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6WAD ,WFC ,WG24 ,FAR24 ,ETAO ,OPDUC ,BYPASS ,DUMS7 ,
7TS24 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON/ EACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWFB ,XWG55 ,XFAR55 ,XWFO ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
DATA AWCRO/6H FERF/
WORD = AWORD

```

G=32.174049  
 CAPSF=2116.2170  
 WFT=WFB+WFD+WFA  
 WAT=WAF-BLQB  
 WGT=WAT+WFT  
 FART=WFT/WAT  
 VA=AH\*CS  
 FRD=VA\*WAF/G  
 VJM=CVMNOZ\*V9  
 FGHH=VJM\*WG7/G  
 FGPH=CAPSF\*(PS9-P1)\*A29  
 IF (IGASHX.GT.0) GO TO 1  
 VJD=CVDNOZ\*V29  
 FGMD=VJD\*WG24/G  
 FGPD=CAPSF\*(FS29-P1)\*A29  
 1 FGH=FGHH+FGMD  
 FGP=FGPH+FGPD  
 FG=FGP+FGP  
 FN=FG-FRD  
 SFC=3600.\*WFT/FN  
 FG=DELF\*FG  
 FN=DELFN\*FN  
 SFC=DELSFC\*SFC  
 CALL CUTPUT  
 CALL ERROR  
 RETURN  
 END

```

SUBROUTINE CCNOUT(ICON)
  DIMENSION WORDY(345), IOUT(150), AOUT(6), WOUT(6), PARA(1(96),
1  PARAM2(121), PARAM3(56), PARAM4(72)
  COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IANTP ,
2IGASHX, IDURN, IAFBN, IDCO ,IMCO ,IDSHOC, IMSHOC, NOZFLT,
3ITRYS, LOOPER, NUMAP, NUMHAP, MAPEOG, TOLALL, ERR(9)
  COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNICS ,PRIOS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4DS ,WFBDOS ,OTCDS ,ETABDS ,WA3CDS ,OPCDS ,OTCCF ,ETABCF ,
6TFHPOS ,CNHPOS ,ETHPOS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T2IDS ,
8TFLPOS ,CNLPOS ,ETLPOS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFCDS ,OTCDS ,ETACDS ,WA2CDS ,OPCDS ,DYDUCF ,ETACCF ,
AT7DS ,WAFDS ,OTAFDS ,ETACDS ,WG6CDS ,OPAFDS ,ETAFCF ,ETAACF ,
BA55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
  COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLDU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLDB ,
7CNI ,PRI ,ETAI ,WAI ,WAI ,BLDBI ,BLDB ,WA3 ,
8CNC ,PRC ,ETAC ,WAC ,WAC ,ETAB ,OPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CWG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALIP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EWFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLC ,
FBLDUC ,PCBLBI ,PCBLBC ,PCBLPI ,PCBLHPC ,PCBLIPI ,PCBLIPC ,PCBLI ,
GPCBLPC
  COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,XBLDU ,XBLDUI ,XBLDUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6W20 ,WFC ,WG24 ,FAR24 ,ETAD ,OPDUC ,BYPASS ,DUMS7 ,
7T28 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
  COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWFB ,XHG55 ,XFAR55 ,XWFO ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,OPAF ,V55 ,V25 ,
6P6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7T28 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMH ,FGPD ,FGPH ,
9F6M ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC

```

EQUIVALENCE (PARAM1(1),PCNFGU),(PARAM2(1),T1),(PARAM3(1),XP1),  
1 (PARAM4(1),XT55)

DATA (WORDY(I),I=1,96)/

17HPCNFGU	,7HPCNIGU	,7HPCNCGU	,7HT4GU	,7HDUM01	,7HDELF	,7HDELF
27HDELFN	,7HDELSFC	,7HZFDS	,7HPCNFDS	,7HPRFDS	,7HETAFDS	,7HETAFDS
37HMAFDS	,7HPRFCF	,7HETAFCF	,7HMAFCF	,7HZIOS	,7HPCNIOS	,7HPCNIOS
47HPRIOS	,7HETAIQS	,7HMAIOS	,7HPRICF	,7HETAIQF	,7HMAICF	,7HMAICF
57HZCDS	,7HPCNCGDS	,7HPRCDS	,7HETACDS	,7HMACDS	,7HPRCCF	,7HPRCCF
67HETACCF	,7HMACCF	,7HT4DS	,7HMFBD	,7HDTCCDS	,7HETABDS	,7HETABDS
77HMA3CDS	,7HOPCQDS	,7HDTCCCF	,7HETABCF	,7HTFHPDS	,7HCNHPDS	,7HCNHPDS
87HETHFDS	,7HTFHPCF	,7HCNHPCF	,7HETHPCF	,7HDMHPCF	,7HT2DS	,7HT2DS
97HTFIPDS	,7HCNIPDS	,7HETIPDS	,7HTFIPCF	,7HCNIPCF	,7HETIPCF	,7HETIPCF
17HDMIPCF	,7HT2IDS	,7HTFLPDS	,7HCNLPDS	,7HETLPS	,7HTFLPCF	,7HTFLPCF
27HCNLPDS	,7HETLPCF	,7HDMHPCF	,7HT22DS	,7HT24DS	,7HMFQDS	,7HMFQDS
37HDTQDS	,7HETADDS	,7HMA23DS	,7HDPQDS	,7HDTQUCF	,7HETADCF	,7HETADCF
47HT7DS	,7HMAFDS	,7HDTAFDS	,7HETADDS	,7HMG6CDS	,7HOPAFDS	,7HOPAFDS
57HDTAFCF	,7HETACCF	,7HA55	,7HA25	,7HA6	,7HA7	,7HA7
67HA8	,7HA9	,7HA28	,7HA29	,7HPS55	,7HAM55	,7HAM55
77HCVDMOZ	,7HCVHMOZ	,7HA8SAV	,7HA9SAV	,7HA28SAV	,7HA29SAV	,7HA29SAV

DATA (WORDY(I),I=97,150)/

17HT1	,7HP1	,7HH1	,7HS1	,7HT2	,7HP2	,7HP2
27HH2	,7HS2	,7HT21	,7HP21	,7HH21	,7HS21	,7HS21
37HT22	,7HP22	,7HH22	,7HS22	,7HT3	,7HP3	,7HP3
47HH3	,7HS3	,7HT4	,7HP4	,7HH4	,7HS4	,7HS4
57HT45	,7HP45	,7HH45	,7HS45	,7HT5	,7HP5	,7HP5
67HH5	,7HS5	,7HT55	,7HP55	,7HH55	,7HS55	,7HS55
77HBLF	,7HBLI	,7HBLC	,7HBLDU	,7HCNF	,7HPRF	,7HPRF
87HETAF	,7HMAFC	,7HMAF	,7HBLDUI	,7HBLDUC	,7HBLDB	,7HBLDB
97HCNI	,7HPRI	,7HETAI	,7HMAIC	,7HMAI	,7HBLDBI	,7HBLDBI

DATA (WORDY(I),I=151,217)/

17HBLDB	,7HMA3	,7HCNC	,7HPRC	,7HETAC	,7HMACC	,7HMACC
27HMAC	,7HETAB	,7HOPCOM	,7HMG4	,7HCNHP	,7HETATHP	,7HETATHP
37HDTCHP	,7HCHTC	,7HBLHP	,7HBLHPI	,7HBLHPC	,7HFAR4	,7HFAR4
47HCNIP	,7HETATIF	,7HDTCHIP	,7HDTI	,7HBLIF	,7HBLIPI	,7HBLIPI
57HBLIPC	,7HCUMF	,7HCNLP	,7HETATLP	,7HDTCLP	,7HDTF	,7HDTF
67HBLLP	,7HBLIPI	,7HBLIPC	,7HCS	,7HMG45	,7HFAR45	,7HFAR45
77HMG5	,7HFAR5	,7HMG55	,7HFAR55	,7HHPEXT	,7HAM	,7HAM
87HALTP	,7HETAR	,7HZF	,7HPCNF	,7HZI	,7HPCNI	,7HPCNI
97HZC	,7HPCNC	,7HWF8	,7HTFFHP	,7HTFFIP	,7HTFFLP	,7HTFFLP
17HPCBLF	,7HPCBLI	,7HPCBLC	,7HPCBLDUI	,7HPCBLDUC	,7HPCBLDBI	,7HPCBLDBI
27HPCBLDB	,7HPCBLHPI	,7HPCBLHPC	,7HPCBLIPI	,7HPCBLIPC	,7HPCBLIPI	,7HPCBLIPI
37HPCBLIPC						

DATA (WORDY(I),I=218,273)/

17HXP1	,7HXNAF	,7HXHA1	,7HXMAC	,7HXBLF	,7HXBLDU	,7HXBLDU
27HXBLDUI	,7HXBLDUC	,7HXH22	,7HXH3	,7HXT21	,7HXP21	,7HXP21
37HXH21	,7HXS21	,7HQUHS1	,7HQUHS2	,7HT23	,7HP23	,7HP23
47HH23	,7HS23	,7HT24	,7HP24	,7HH24	,7HS24	,7HS24
57HT25	,7HP25	,7HH25	,7HS25	,7HT28	,7HP28	,7HP28
67HH28	,7HS28	,7HT29	,7HT29	,7HH29	,7HS29	,7HS29
77HQUHS3	,7HQUHS4	,7HQUHS5	,7HQUHS6	,7HWAU	,7HWF	,7HWF
87HMG24	,7HFAR24	,7HETAD	,7HOPDUC	,7HBYPASS	,7HQUHS7	,7HQUHS7
97HTS28	,7HPS28	,7HV28	,7HAM28	,7HTS29	,7HPS29	,7HPS29
17HV29	,7HAM29					

DATA (WORDY(I),I=274,345)/

17HXT55	,7HXP55	,7HXH55	,7HXS55	,7HXT25	,7HXP25	,7HXP25
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27HXH25	,7HXS25	,7HXXF8	,7HXXG55	,7HXFAR55	,7HXXWFD	,
37HXXG24	,7HXFAR24	,7HXXP1	,7HOLMB	,7HT6	,7HP6	,
47H46	,7HS6	,7HT7	,7HP7	,7HH7	,7HS7	,
57HT8	,7HF8	,7HH8	,7HS8	,7HT9	,7HP9	,
67H49	,7HS9	,7HWG6	,7HWFA	,7HWG7	,7HFAR7	,
77HETAA	,7HCPAFT	,7HV55	,7HV25	,7HPS6	,7HV6	,
87HAM6	,7HTS7	,7HPS7	,7HV7	,7HAM7	,7HAM25	,
97HTS8	,7HFS8	,7HV8	,7HAM8	,7HTS9	,7HPS9	,
17HV9	,7HAM9	,7HVA	,7HFRD	,7HVJD	,7HFGMD	,
27HJH	,7HFGMH	,7HFGPD	,7HFGPM	,7HFGM	,7HFGP	,
37HWFT	,7HWGT	,7HFART	,7HFG	,7HFN	,7HSFC	/

DATA THEEND, BLANK, LIMIT//HTHEEND ,7H ,345/

GO TO (1,12), ICCN

C \*\*\* INPUT SECTION

1 GO 4 N=1,150

NUM=N

READ(5,100)AIN,CHANGE

IF(AIN.EQ.THEEND) GO TO 5

GO 2 J=1,LIMIT

JJ=J

IF(AIN.EQ.WORDY(J)) GO TO 3

2 CONTINUE

WRITE(6,101)AIN

GO TO 4

3 IOUT(NUM)=JJ

IF(CHANGE.NE.BLANK) WORDY(JJ)=CHANGE

4 CONTINUE

WRITE(6,102)

5 NUM=NUM-1

RETURN

C \*\*\* OUTPUT SECTION

12 IF(NUM.EQ.1) GO TO 16

N=NUM

J=6

GO 15 I=1,NUM,6

IF(N.GT.6) GO TO 13

L=N

13 N=N-6

GO 14 K=1,J

L=L+K-1

M=IOUT(L)

WOUT(K)=WORDY(M)

IF(M.GT.96) GO TO 20

WOUT(K)=PARAM1(M)

GO TO 14

20 IF(M.GT.217) GO TO 21

MN=M-96

WOUT(K)=PARAM2(MN)

GO TO 14

21 IF(M.GT.273) GO TO 22

MN=M-217

WOUT(K)=PARAM3(MN)

GO TO 14

22 MN=M-273

WOUT(K)=PARAM4(MN)

```

14 CONTINUE
   WRITE(6,103) (WOUT(K),K=1,J)
   WRITE(6,104) (AOUT(K),K=1,J)
   IF(N.LE.0) GO TO 16
15 CONTINUE
16 RETURN
100 FORMAT(A7,7X,A7)
101 FORMAT('0H0THE WORD ,A7,26H NOT FOUND IN COMMON ARRAY)
102 FORMAT(22H0ERROR IN CONOUT INPUT)
103 FORMAT(1H ,25XA7,5(8XA7))
104 FORMAT(1H ,20X6E15.6)
END

```

```

SUBROUTINE ERROR
DIMENSION TRASH1(96),TRASH2(121),TRASH3(56),TRASH4(72)
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IDUMP ,IATP ,
2IGASHX,IOBURN,IAFTBN,IGCD ,IMCD ,IOSHOC,IMSHOC,NOZFLT,
3ITRYS,LOOPER,NOHAP,NUMHAP,MAPEOG,TOLALL,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2PFDS ,PCNFDS ,PRFDS ,ETAFDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIIDS ,ETAIIDS ,WAIIDS ,PRICF ,ETAICF ,WAIICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5W8DS ,WF8DS ,DTGDS ,ETABDS ,WA3CDS ,DPCDS ,DTGCCF ,ETABCF ,
6THPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T2IDS ,
8TLPDS ,CNLPDS ,ETLPDS ,TFLPCF ,CNLPCF ,ETLPCF ,DHLPCF ,T22DS ,
9T4DS ,WFDS ,DTGDS ,ETADDS ,WA23DS ,DPODS ,DTGUCF ,ETADCF ,
A17DS ,WFADS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
B455 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVDNOZ ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BLI ,BLC ,BLOU ,
6CNF ,PRF ,ETAF ,WAF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAI ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9C.HP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,DHTI ,BLIP ,BLIPI ,BLIPC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLLPI ,BLLLPC ,CS ,
C.H45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHF ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPBLOUC,PCBLOBI,PCBLOBC,PCBLHPI,PCBLHPC,PCBLIFI,PCBLIPC,PCBLLLPI,
GPBLLFC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XBLF ,XBLOU ,XBLOUI ,XBLOUC ,
2XH21 ,XH3 ,XT21 ,XP21 ,XF21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6H21 ,WFC ,WG24 ,FAR24 ,ETA0 ,DPOUC ,BYPASS ,DUMS7 ,
7T24 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON/ BACK/
XX55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXH55 ,XHG55 ,XFAR55,XWFD ,XWG24 ,XFAR24,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5HG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DFAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGHM ,FGPD ,FGFM ,
9FCM ,FGP ,HFT ,HGT ,FART ,FG ,FN ,SFC
EQUIVALENCE (TRASH1(1),PCNFGU), (TRASH2(1),T1), (TRASH3(1),XP1)

```

```

EQUIVALENCE (TRASH4(1),XT55)
DATA AWORD/6HCOMMON/
WRITE(6,100)WORD
WORD=AWORD
WRITE(6,102)WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE
WRITE(6,103)
WRITE(6,104)(TRASH1(I),I=1,96)
WRITE(6,105)
WRITE(6,104)(TRASH2(I),I=1,121)
WRITE(6,105)
WRITE(6,104)(TRASH3(I),I=1,56)
WRITE(6,103)
WRITE(6,104)(TRASH4(I),I=1,72)
WRITE(6,103)
WRITE(6,106)LOOPER
IF(IDUMP.EQ.0) GO TO 2
WRITE(6,105)
CALL SYG(2)
2   CALL ENGBAL
    RETURN
100  FORMAT(28H0AN ERROR HAS BEEN FOUND IN ,A6)
102  FORMAT(1H0,A6,9X,7E15.6,I4)
103  FORMAT(2H0 )
104  FORMAT(1H0,8E15.6)
105  FORMAT(1H1)
106  FORMAT(25H0FAILED TO CONVERGE AFTER,I4,6H LOOPS)
END

```



```

SUBROUTINE THCOMP(PR,ETA,T,H,S,P,TO,H0,SO,PO)
PO=P*FR
TP=T*PR*.28572
DO 1 I=1.25
CALL THERMC(PO,HP,TP,SP,X1,0,X2,0)
DELS=SP-S
IF (ABS(DELS).LE.0.00005*S) GO TO 2
1 TP=TP/EXP(4.*DELS)
CALL ERROR
2 HQ=H+((HF-H)/ETA)
CALL THERMC(PO,HQ,TO,SO,X1,0,X2,1)
RETURN
END

```

SUBROUTINE THTURB(DH,ETA,FAR,H,S,F,TC,HC,SO,PC)

HO=H-CH

HOP=H-DH/ETA

PT=P/2.

DO 1 I=1,25

CALL THERMC(PT,HOP,TT,ST,AMWT,1,FAR,1)

DELS=ST-S

IF (ABS(DELS).LE.0.00005\*S) GO TO 2

1 PT=P\*EXP(DELS\*AMWT/1.986375+ALOG(FT/P))

CALL ERROR

2 PO=PT

CALL THERMC(PO,HC,TO,SO,X1,1,FAR,1)

RETURN

END

```

SUBROUTINE PRCCOM(FARX,TCX,CSEX,TEX,CPEX,REX,PHI,HEX)
IF (FARX.LE.0.0) GO TO 1
FARX=0.067623
WRITE(8,101)
1 IF (TEX.GE.300.) GO TO 2
  TCX=300.
  WRITE(8,102)
2 IF (TEX.LE.4000.) GO TO 3
  TCX=4000.
  WRITE(8,103)
3 IF (FARX.GE.0.0) GO TO 4
  FARX=0.0
  WRITE(8,104)
C AIR PATH
4 CPA=(((((1.0115540E-25*TEX-1.4526770E-21)*TEX
1+7.621577E-18)*TEX-1.5122259E-14)*TEX-6.7178376E-12)
2*TEX+6.5519486E-08)*TEX-5.1536879E-05)*TEX+2.5020051E-01
  HEA=(((((1.2644425E-26*TEX-2.752522E-22)*TEX
1+1.2702630E-18)*TEX-3.0256518E-15)*TEX-1.6794594E-12)*TEX
2+2.1839826E-08)*TEX-2.5773440E-05)*TEX+2.5020051E-01)*TEX
3-1.755883E-01
  SEF=((2.120091E-01*ALOG(TEX)+((((1.4450767E-26*TEX
1-2.4211288E-22)*TEX+1.5243153E-18)*TEX-3.7820648E-15)*TEX
2-2.2392790E-12)*TEX+3.2759743E-08)*TEX-5.1576679E-05)*TEX
3+4.5432300E-02
  IF (FARX.LE.0.0) GO TO 5
C FUEL/AIR PATH
  CPF=(((((7.2678710E-25*TEX-1.3335668E-20)*TEX
1+1.1212913E-16)*TEX-4.2051104E-13)*TEX+3.9686793E-10)*TEX
2-1.3771901E-06)*TEX+1.2258630E-03)*TEX+7.3816638E-02
  HEF=(((((9.0848338E-26*TEX-1.9050949E-21)*TEX
1+1.7021525E-17)*TEX-8.4102208E-14)*TEX+2.4921698E-10)*TEX
2-4.5906332E-07)*TEX+6.1293150E-04)*TEX+7.3816638E-02)
3*TEX+3.0501530E+01
  SEF=((7.3816638E-02*ALOG(TEX)+((((1.0382670E-25*TEX
1-2.222611E-21)*TEX+2.0425326E-17)*TEX-1.0512776E-13)*TEX
2+3.3228928E-10)*TEX-6.885950E-07)*TEX+1.2258630E-03)*TEX
3+6.483399E-01
5 CPEX=(CPA+FARX)/(1.+FARX)
  HEX=(HEA+FARX*HEF)/(1.+FARX)
  PHI=(SEF+FARX*SEF)/(1.+FARX)
  AMW=28.97-.946186*FARX
  REX=1.98637E-01
  CSEX=SQRT(AMW*REX*TCX*25031.37)
  RETURN
101 FORMAT(1H,63HINPUT FUEL-AIR RATIO ABOVE LIMITS, IT HAS BEEN RESET
2TO 0.067623,CH$)
102 FORMAT(1H,64HPRCCOM INPUT TEMPERATURE BELOW 300.,6H$)
103 FORMAT(1H,64HPRCCOM INPUT TEMPERATURE ABOVE 4000.,6H$)
104 FORMAT(1H,38HPRCCOM INPUT FUEL-AIR RATIO BELOW ZERO,6H$)
END

```



```

SUBROUTINE SEARCH(P,A,B,C,D,AX,NA,BX,CX,DX,NO,NAH,NGH,NCODE)
DIMENSION AX(NAH),BX(NAH,NOH),CX(NAH,NOH),DX(NAH,NOH),NO(NAH),Q(9)
C *** NEEDS SUBROUTINE AFQUIR
C *** AX AND BX MUST BE STORED LO TO HI
C *** P= INPUT PROPORTION BETWEEN 0.0 AND 1.0
C *** IF NOT INPUT, P MUST EQUAL -1.
C *** NCODE=00 OK
C NCODE=01 A LO
C NCODE=02 A HI
C NCODE=07 ERROR
C NCODE=10 B LO
C NCODE=20 B HI
C NCODE=0
C=0.
D=0.
C *** FIND A
DO 1 I=1,NA
IH=I
IF (A.LT.AX(I)) GO TO 2
1 CONTINUE
IF (A.GT.AX(IH)) NCODE=2
A=AX(IH)
GO TO 3
2 IF (IH.GT.1) GO TO 3
NCODE=1
IH=2
A=AX(1)
3 IL=IH-1
LIH=NO(IH)
LIL=NO(IL)
C *** FIND B
FRM=(A-AX(IL))/(AX(IH)-AX(IL))
PP=0
IF (P.GE.0.) GO TO 6
BL=AX(IL,1)+FRM*(BX(IH,1)-BX(IL,1))
BH=AX(IL,LIL)+FRM*(BX(IH,LIH)-BX(IL,LIL))
IF (3.GE.BL) GO TO 4
NCODE=NCODE+10
B=BL
GO TO 5
4 IF (3.LE.BH) GO TO 5
NCODE=NCODE+20
B=B+
5 PP=0.5
C(2)=0.
Q(2)=0.
6 BH=PP*(BX(IH,LIH)-BX(IH,1))+BX(IH,1)
BL=PP*(BX(IL,LIL)-BX(IL,1))+BX(IL,1)
DO 7 J=2,LIL+
JH=J
IF (BH.LT.BX(IH,J)) GO TO 8
7 CONTINUE
8 JL=JH-1
GO 9 K=2,LIL
KH=K

```

```

          IF (BL.LT.BX(IL,K)) GO TO 10
9         CONTINUE
10        KL=KH-1

          PR=(BX(IH,JL)-BH)/(BX(IH,JH)-BX(IH,JL))
          CH= CX(IH,JL)-PR*(CX(IH,JH)-CX(IH,JL))
          OH= OX(IH,JL)-PR*(OX(IH,JH)-OX(IH,JL))

          PR=(BX(IL,KH)-BL)/(BX(IL,KH)-BX(IL,KL))
          CL= CX(IL,KH)-PR*(CX(IL,KH)-CX(IL,KL))
          DL= OX(IL,KH)-PR*(OX(IL,KH)-OX(IL,KL))

          BT=BL+PR*(CH-BL)
          CT=CL+PR*(OH-CL)
          DT=DL+PR*(OH-DL)

          IF (P.GT.1) GO TO 13
          DIR=SQRT(BT/DT)
          ERR=(C-B)/B
          CALL AFQUIR(Q(1),PP,ERR,0.,25.,0.001,DIR,PT,ICON)
          GO TO (11,12,12),ICON
          PP=PT
          IF (PP.LT.0.) PP=0.
          IF (PP.GT.1) PP=1.
          GO TO 6
12        NCODE=7
13        B=BT
          C=CT
          D=DT
          RETURN
          END

```

```

      SUBROUTINE CCNVRG(TI,HI,PI,SI,FAR,NG,PA,IDES,AO,PR,
1 TO,PO,SC,TSO,PSO,VO,AMO,ICON)
C      ICON=1      SUBSONIC, COMPARE PI WITH PR
C      ICON=2      SONIC, COMPARE PI WITH PR
C      ICON=4      ERROR
      AJ=778.26
      CAPSF=2116.217
      G=32.174049
      CALL PROCOM(FAR,TI,XX1,XX2,XX3,XX4,PHI,XX6)

C *** SONIC CALCULATIONS

      J=0
      TSS=0.833*TI
1      J=J+1
      CALL PROCOM(FAR,TSS,CSS,AKS,CP,REXS,PHISS,HSS)
      HSCAL=HI-CSS**2/(2.*G*AJ)
      DELHS=HSCAL-HSS
      IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2
2      TSS=TSS+DELHS/CP
      IF (J-15) 1,1,3
3      ICON=4
      RETURN
4      IF (IDES) 12,12,5

C *** ISENTROPIC EXPANSION CALCULATIONS

5      J=0
      TSI=TI*(PA/PI)**0.286
6      J=J+1
      CALL THERMO(PA,HSI,TSI,SSI,XX1,1,FAR,0)
      IF (ABS(SSI-SI)-0.0001*SI) 8,8,7
7      TSI=TSI/EXP(4.*(SSI-SI))
      IF (J-30) 6,6,3
8      VTS=SQRT(2.*G*AJ*(HI-HSI))
      IF (VIS-CSS) 9,11,11

C *** SUBSONIC DESIGN, CALCULATE AO

9      VO=VIS
      TSO=TSI
      PSO=PA
      CALL PROCOM(FAR,TSO,CSO,XX2,XX3,REX,PHISO,HSO)
      RHO=(CAPSF*FSC/(AJ*REX*TSO))
      AO=NG/(RHO*VO)
      AMO=VO/CSO
      PR=PI
      ICON=1
10     TO=TI
      HO=HI
      PO=PI
      SO=SI
      RETURN

C *** SONIC DESIGN, CALCULATE AO

```

```

11  VO=CSS
    TSO=TSS
    PSO=PI*(TSO/TSI)**(AKS/(AKS-1.))
    RHO=CAPSF*FSO/(AJ*REXS*TSO)
    AO=WG/(RHO*VO)
    AMO=1.0
    PR=PI
    ICON=2
    GO TO 10

```

C \*\*\* NON-DESIGN, CALCULATE CRITICAL CONDITIONS

```

12  VO=CSS
    TSO=TSS
    PSO=PA
    RHO=CAPSF*PSO/(AJ*REXS*TSO)
    AOCRT=WG/(RHO*VO)
    AMO=1.0
    PR=PSO*(TSI/TSO)**(AKS/(AKS-1.))
    IF (AO-AOCRT) 13,13,14

```

C \*\*\* NON-DESIGN, CRITICAL AND SUPERCRITICAL CONDITIONS

```

13  FSO=PSO*AOCRT/AO
    PR=PR*AOCRT/AO
    ICON=2
    GO TO 10

```

C \*\*\* NON-DESIGN, SUBSONIC CALCULATIONS

```

14  PSO=PA
    J=0
    TSO=0.833*TSO
15  J=J+1
    CALL PROCOM(FAR,TSO,CSO,AKO,CP,REX,PHISO,HSO)
    RHO=CAPSF*FSO/(AJ*REX*TSO)
    VO=WG/(RHO*AO)
    HSCAL=HI-VC**2/(2.*G*AJ)
    DELHS=HSCAL-HSO
    IF (ABS(DELHS)-0.0005*HSCAL) 17,17,16
16  TSO=TSO+DELHS/CP
    IF (J-15) 15,15,3
17  AMO=VO/CSO
    PR=PSO*(TSI/TSO)**(AKO/(AKO-1.))
    ICON=1
    GO TO 10
END

```

```

SUBROUTINE CCNDIV(TI,PI,PI,SI,FAR,WG,PA,IDES,AT,AO,PIR,
1 TI,HT,PT,ST,TC,HQ,PO,SO,TST,TSO,PST,PSO,VT,VO,AMT,AMO,ICON)
C   ICON=1  SUBSONIC, COMPARE PIR WITH PI
C   ICON=2  SONIC, SHOCK INSIDE NOZZLE, COMPARE PIR WITH PI
C   ICON=3  SONIC, SHOCK OUTSIDE NOZZLE, COMPARE PIR WITH PI
C   ICON=4  ERROR
      DIMENSION G(9)
      G(2)=0.
      G(3)=0.
      AJ=778.26
      CAPSF=2116.2170
      G=32.174049
      CALL PROCOM(FAR,TI,XX1,XX2,XX3,XX4,PHII,XX6)

C *** SONIC CALCULATIONS

      J=0
      TSS=0.833*TI
1     J=J+1
      CALL PROCOM(FAR,TSS,CSS,AK,CP,REXS,PHISS,HSS)
      HSCAL=HI-CSS**2/(2.*G*AJ)
      DELHS=HSCAL-TSS
      IF (ABS(DELHS)-0.0005*HSCAL) 4,4,2
2     TSS=TSS+DELHS/CP
      IF (J-15) 1,1,3
3     ICON=4
      RETURN
4     IF (IDES) 11,11,5

C *** SONIC DESIGN, CALCULATE AT

5     VT=CSS
      TST=TSS
      PST=PI*(TST/TI)**(AK/(AK-1.))
      RHO=CAPSF*PST/(AJ*REXS*TST)
      AT=WG/(RHO*VT)
      AMT=1.0

C *** IDEAL EXPANSION DESIGN, CALCULATE AO

      PSC=PA
      J=0
      TSO=TI*(PSC/PI)**.286
6     J=J+1
      CALL PROCOM(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
      PHICAL=PHII-REX*ALOG(PI/PSO)
      DELPHI=PHICAL-PHISO
      IF (ABS(DELPHI)-0.0001*PHICAL) 8,8,7
7     TSO=TSO*EXP(4.*DELPHI)
      IF (J-15) 6,6,3
8     VO=SQRT(2.*G*AJ*(HI-HSO))
      AMO=VO/CSO
      AO=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/(2.*
1 (AK-1.)))
      FIR=PI

```

```

      ICON=3
9      TO=TI
      HO=HI
      PO=PI
      SO=SI
10     TT=TI
      HT=HI
      PT=PI
      ST=SI
      RETURN

C *** ASSUME SONIC THROAT AND ISENTROPIC EXPANSION TO AO

11     VT=...
      AMT=1.0
      P=1SS
      RHO=HG/(AT*VT)
      PST=RHO*AJ*REXS*TST/CAPSF
      PIR=PST*(TI/TST)**(AK/(AK-1.))
      IF (PST-PA) 12,24,24
12     TSO=0.95*TI
      HAM=0
13     CALL FROCOF(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
      AMO=SQRT(2.*((TI/TSO)-1.)/(AK-1.))
      AOAL=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/
1 (2.*(AK-1.)))
      EA=(AO-AOAL)/AC
      DIR=SQRT(AO/AOAL)
      CALL AFQUIR(Q(1),TSO,EA,0.,100.,0.0001,DIR,TSOT,JCON)
      GO TO (14,15,3),JCON
14     TSO=TSOT
      IF (TSC-TI) 140,13,141
140    TSC=2.*TI/(AK+1.)
      IF (TSO.GT.TSC) GO TO 142
141    TSO=0.98*TI
      GO TO 13
142    IF (Q(2).LT.30.0.OR.AMO.LT.0.95.OR.HAM.EQ.1) GO TO 13
      TSO=2.*TI/(2.+0.98*(AK-1.))
      HAM=1
      GO TO 13
15     FSO=PIR*(TSO/TI)**(AK/(AK-1.))
      IF (FSO-PA) 17,16,24
C *** CRITICAL FLOW, ISENTROPIC EXPANSION TO PA

16     VO=AMO*CSO
      ICON=1
      GO TO 9

C *** SUBSONIC FLOW

17     FSO=PA
      Q(2)=0.
      Q(3)=0.
      J=0
      TSO=0.833*TI

```

```

18  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    RHO=CAPSF*PSC/(AJ*REX*TSO)
    VO=HG/(RHO*AO)
    HSCAL=HI-VC**2/(2.*G*AJ)
    DELHS=HSCAL-HSO
    IF (ABS(DELHS)-0.0005*HSCAL) 20,20,19
19  TSO=TSO+DELHS/CF
    IF (J-15) 18,18,3
20  AMO=VO/CSO
    FIR=PSO*(TI/TSO)**(AK/(AK-1.))
    TST=TSO
21  CALL PROCOP(FAR,TST,CST,AK,CP,REX,PHIST,HST)
    PST=PIR*(TST/TI)**(AK/(AK-1.))
    RHO=PST*CAPSF/(AJ*REX*TST)
    VT=HG/(RHO*AT)
    HSCAL=HI-VT**2/(2.*G*AJ)
    EH=(HSCAL-HST)/HSCAL
    DIR=1.+(HSCAL-HST)/(CP*TST)
    CALL AFQUIR(Q(1),TST,EH,0.,20.,0.0005,DIR,TSTT,JCON)
    GO TO (22,23,3),JCON
22  TST=TSTT
    GO TO 21
23  AMT=VT/CST
    ICON=1
    GO TO 9

```

C \*\*\* SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO PA

```

24  PSO=PA
    J=0
    TSO=TI*(PSC/FIR)**.286
25  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    PHICAL=PHII-REX*ALOG(PIR/PSO)
    DELPHI=PHICAL-PHISO
    IF (ABS(DELPHI)-0.0001*PHICAL) 27,27,26
26  TSO=TSO*EXP(4.0*DELPHI)
    IF (J-15) 25,25,3
27  VO=SQRT(2.*G*AJ*(HI-HSO))
    AMO=VC/CSO
    AOI)=(AT/AMO)*(2.*(1.+(AK-1.)*AMO**2/2.)/(AK+1.))**((AK+1.)/
1 (2.*(AK-1.)))
    ICON=3
    N=0
    IF (AO-AOI) 28,9,29

```

C \*\*\* SUPERCRITICAL FLOW, ISENTROPIC EXPANSION TO AO

```

28  N=1
29  TSO=0.833*TI
    J=0
30  J=J+1
    CALL PROCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
    AMO=SQRT(2.*(TI/TSO)-1.)/(AK-1.)

```

```

      AOCAL=(AT/AMC)*(2.*(1.+(AK-1.)*AMC**2/2.)/(AK+1.))**((AK+1.)/
1 (2.*(AK-1.)))
      CELA=AO-AOCAL
      IF (ABS(CELA)-0.0001*AO) 32,32,31
31      TSO=TSO*SQRT(AOCAL/AO)
      IF (J-50) 30,30,3
32      IF (N) 34,34,33

C *** UNDEREXPANDED, SHOCK OUTSIDE NOZZLE
33      PS0=PIR*(TSO/TI)**(AK/(AK-1.))

      VO=AMC*CSO
      GO TO 9

C *** OVEREXPANDED, FIND SHOCK POSITION
34      PSX=PIR*(TSO/TI)**(AK/(AK-1.))
      PSY=PSX*(2.*AK*AM0**2/(AK+1.)-(AK-1.)/(AK+1.))
      IF (PA-PSY) 35,36,36

C *** OVEREXPANDED, SHOCK OUTSIDE NOZZLE
35      PS0=PSX
      VO=AMC*CSO
      GO TO 9

C *** OVEREXPANDED, SHOCK INSIDE NOZZLE
36      PS0=PA
      J=0
      TSO=0.833*TI
37      J=J+1
      CALL PRUCOP(FAR,TSO,CSO,AK,CP,REX,PHISO,HSO)
      RHO=CAPSF*PSC/(AJ*REX*TSO)
      VO=WG/(RHO*AC)
      HSCAL=HI-VO**2/(2.*G*AJ)
      DELHS=HSCAL-HSO
      IF (ABS(DELHS)-0.0005*HSCAL) 39,39,38
38      TSO=TSO+DELHS/CF
      IF (J-15) 37,37,3
39      AM0=VC/CSO
      TO=TI
      HO=HI
      PO=PS0*(TO/TSO)**(AK/(AK-1.))
      SO=PHII-REX*ALOG(PO)
      ICON=2
      GO TO 10
      END

```



```

SUBROUTINE FQUIR(X,AIND,DEPEND,ANS,AJ,TOL,DIR,ANEH,ICON)
DIMENSION X(9)
C X(1)=NAME OF ARRAY TO USE
C AIND=INDEPENDANT VARIABLE
C DEPEND= DEPENDANT VARIABLE
C ANS=ANSWER UPON WHICH TO CONVERGE
C AJ=MAX NUMBER OF TRYs
C TOL=PERCENT TOLERANCE FOR CONVERGENCE
C DIR=DIRECTION AND PERCENTAGE FOR FIRST GUESS
C ANEH=CALCULATED VALUE OF NEXT TRY AT INDEPENDANT VARIABLE
C ICON=CONTROL  =1 GO THRU LOOP AGAIN
C               =2 YOU HAVE REACHED THE ANSWER
C               =3 COUNTER HAS HIT LIMITS
C X(2)=COUNTER STORAGE
C X(3)=CHOOSES METHOD OF CONVERGENCE
C X(4)=THIRD DEPEND VAR
C X(5)=THIRD IND VAR
C X(6)=SECOND DEPEND VAR
C X(7)=SECOND INC VAR
C X(8)=FIRST DEPEND VAR
C X(9)=FIRST IND VAR
C X(3) MUST BE ZERO UPON FIRST ENTRY TO ROUTINE

```

```

      Y=0.
      IF (ANS)1,2,1
1     DEP=DEPEND-ANS
      TOLANS=TOL*ANS
      GO TO 3
2     DEP=DEPEND
      TOLANS=TOL
3     IF (ABS(DEP)-TOLANS)5,5,4
4     IF (X(2)-AJ)6,8,7
5     ANEH=AIND
      X(2)=0.
      ICON=2
      RETURN
6     ANEH=Y
      X(2)=X(2)+1.
      ICON=1
      RETURN
7     ANEH=Y
      X(2)=0.
      ICON=3
      RETURN
8     IF (X(3))9,9,12
C *** FIRST GUESS USING DIR
9     X(3)=1.
      X(8)=DEP
      X(9)=AIND
      IF (AIND)10,11,10
10    Y=DIR*AIND
      GO TO 6
11    Y=DIR
      GO TO 6
12    IF (X(3)-1.)13,13,16

```

```

C *** LINEAR GUESS
13  X(3)=2.
    X(6)=DEP
    X(7)=AIND
    IF (X(8)-X(6))14,9,14
14  IF (X(9)-X(7))15,9,15
15  A=(X(9)-X(7))/(X(8)-X(6))
    Y=X(9)-A*X(8)
    IF (ABS(10.*X(9))-ABS(Y))9,9,6
C *** QUADRATIC GUESS
16  X(4)=DEP
    X(5)=AIND
    IF (X(7)-X(5))18,17,18
17  IF (X(6)-X(4))13,9,13
18  IF (X(6)-X(4))19,13,19
19  IF (X(9)-X(5))23,20,23
20  IF (X(8)-X(4))21,22,21
21  X(9)=X(7)
    X(8)=X(6)
    GO TO 13
22  X(9)=X(7)
    X(8)=X(6)
    X(3)=1.
    IF (X(9))10,11,10
23  IF (X(8)-X(4))24,21,24
24  F=(X(6)-X(4))/(X(7)-X(5))
    A=(X(8)-X(4)-F*(X(9)-X(5)))/((X(9)-X(7))*(X(9)-X(F)))
    B=F-A*(X(5)+X(7))
    C=X(4)+X(5)*A*(X(7)-X(5))
    IF (A)242,240,242
240 IF (B)241,7,241
241 Y=-C/B
    GO TO 37
242 IF (B)247,243,247
243 IF (C)245,244,245
244 Y=0.
    GO TO 37
245 G=-C/A
    IF (G)7,7,246
246 Y=SQRT(G)
    YY=-SQRT(G)
    GO TO 270
247 IF (C)249,248,249
248 Y=-B/A
    YY=0.
    GO TO 270
249 D=4.*A*C/B**2
    IF (1.-D)13,25,26
25  Y=-B/(2.*A)
    GO TO 37
26  E=SQRT(1.-D)
27  Y=(-B/(2.*A))*(1.+E)
    YY=(-B/(2.*A))*(1.-E)
270 J=4
    DEPMIN=ABS(X(4))

```

```

DO 29 I=6,8,2
IF (DEPMIN-ABS(X(I))) 29,29,28
28 J=I
DEPMIN=ABS(X(I))
29 CONTINUE
K=J+1
IF ((X(K)-Y)*(X(K)-YY)) 32,32,30
30 IF (ABS(X(K)-Y)-ABS(X(K)-YY)) 37,37,31
31 Y=YY
GO TO 37
32 IF (J-6) 33,34,34
33 JJ=J+2
KK=K+2
GO TO 35
34 JJ=J-2
KK=K-2
35 SLOPE=(X(KK)-X(K))/(X(JJ)-X(J))
IF (SLOPE*X(J)*(X(K)-Y)) 36,36,37
36 Y=YY
37 X(9)=X(7)
X(8)=X(6)
X(7)=X(5)
X(6)=X(4)
GO TO 6
END

```

```

SUBROUTINE MATRIX(E,V,A)
DIMENSION E(9,9),V(9),A(9),PIV(10),T(9,10)
DO 1 I=1,9
T(I,10)=A(I)
DO 1 J=1,9
1 T(I,J)=E(I,J)
DO 7 I=1,9
TEMP=0.
DO 2 J=I,9
IF(TEMP.GT.ABS(T(J,I))) GO TO 2
TEMP=ABS(T(J,I))
IPIV=J
2 CONTINUE
IP1=I+1
DO 3 J=IP1,10
3 PIV(J)=T(IPIV,J)/T(IPIV,I)
IFROM=9
ITO=9
4 IF(IFROM.EC.IPIV) GO TO 6
RM=-T(IFROM,I)
DO 5 J=IP1,10
5 T(ITO,J)=T(IFROM,J)+RM*PIV(J)
ITO=ITO-1
6 IFROM=IFROM-1
IF(IFROM.GE.I) GO TO 4
DO 7 J=IP1,10
7 T(I,J)=PIV(J)
DO 8 I=1,8
J=10-I
K=9-I
DO 8 L=J,9
8 T(K,10)=T(K,10)-T(K,L)*T(L,10)
DO 9 I=1,9
9 V(I)=T(I,10)
RETURN
END

```

```

SUBROUTINE OUTPUT
DIMENSION b(5,4),ANS1(96),ANS2(121),ANS3(56),ANS4(72)
COMMON / ALL/
1WORD ,IDES ,JDES ,KDES ,MODE ,INIT ,IGUMP ,IAMTP ,
2IGASM ,IOBURN ,IAFTBN ,IOCD ,IMCD ,IDSHOC ,IMSHOC ,NOZFLT ,
3ITRYS ,LOOPER ,NOHAP ,NUMHAP ,MAPEOG ,TOLALL ,ERR(9)
COMMON/DESIGN/
1PCNFGU ,PCNIGU ,PCNCGU ,T4GU ,DUMD1 ,DELFG ,DELFN ,DELSFC ,
2ZFDS ,PCNFDS ,PRFDS ,ETA FDS ,WAFDS ,PRFCF ,ETAFCF ,WAFCF ,
3ZIDS ,PCNIDS ,PRIIDS ,ETAIDS ,WAIDS ,PRICF ,ETAICF ,WAICF ,
4ZCDS ,PCNCDS ,PRCDS ,ETACDS ,WACDS ,PRCCF ,ETACCF ,WACCF ,
5T4JS ,WFBDs ,DTCODS ,ETA9DS ,WA3CDS ,DPCODS ,DTCOCF ,ETABCF ,
6TFHPDS ,CNHPDS ,ETHPDS ,TFHPCF ,CNHPCF ,ETHPCF ,DHHPCF ,T2DS ,
7TFIPDS ,CNIPDS ,ETIPDS ,TFIPCF ,CNIPCF ,ETIPCF ,DHIPCF ,T21DS ,
8TFLPDS ,CNLPDS ,ETL PDS ,TFLPCF ,CNLP CF ,ETLPCF ,DHLPCF ,T22DS ,
9T24DS ,WFUDS ,DTUDS ,ETAADS ,WA23DS ,DPUUDS ,DTUCF ,ETAOCF ,
AT7DS ,WFAOS ,DTAFDS ,ETAADS ,WG6CDS ,DPAFDS ,DTAFCF ,ETAACF ,
8A55 ,A25 ,A6 ,A7 ,A8 ,A9 ,A28 ,A29 ,
CPS55 ,AM55 ,CVON0Z ,CVMNOZ ,A8SAV ,A9SAV ,A28SAV ,A29SAV
COMMON/ FRONT/
1T1 ,P1 ,H1 ,S1 ,T2 ,P2 ,H2 ,S2 ,
2T21 ,P21 ,H21 ,S21 ,T22 ,P22 ,H22 ,S22 ,
3T3 ,P3 ,H3 ,S3 ,T4 ,P4 ,H4 ,S4 ,
4T45 ,P45 ,H45 ,S45 ,T5 ,P5 ,H5 ,S5 ,
5T55 ,P55 ,H55 ,S55 ,BLF ,BL1 ,BLC ,BLDU ,
6CNF ,PRF ,ETAF ,WAFCF ,WAF ,BLDUI ,BLDUC ,BLOB ,
7CNI ,PRI ,ETAI ,WAIC ,WAI ,BLOBI ,BLOBC ,WA3 ,
8CNC ,PRC ,ETAC ,WACC ,WAC ,ETAB ,DPCOM ,WG4 ,
9CNHP ,ETATHP ,DHTCHP ,DHTC ,BLHP ,BLHPI ,BLHPC ,FAR4 ,
ACNIP ,ETATIP ,DHTCIP ,GHTI ,BLIP ,BLIPI ,BLIFC ,DUMF ,
BCNLP ,ETATLP ,DHTCLP ,DHTF ,BLLP ,BLLPI ,BLLPC ,CS ,
CNG45 ,FAR45 ,WG5 ,FAR5 ,WG55 ,FAR55 ,HPEXT ,AM ,
DALTP ,ETAR ,ZF ,PCNF ,ZI ,PCNI ,ZC ,PCNC ,
EMFB ,TFFHP ,TFFIP ,TFFLP ,PCBLF ,PCBLI ,PCBLC ,PCBLDUI ,
FPCBLDUC ,PCBLCBI ,PCBL0BC ,PCBLHPI ,PCBLHPC ,PCBLIFI ,PCBLIPC ,PCBL LPI ,
GPCBL LPC
COMMON/ SIDE/
1XP1 ,XWAF ,XWAI ,XWAC ,XELF ,XBLOU ,XBLOUI ,XBLOUC ,
2XH22 ,XH3 ,XT21 ,XP21 ,XH21 ,XS21 ,DUMS1 ,DUMS2 ,
3T23 ,P23 ,H23 ,S23 ,T24 ,P24 ,H24 ,S24 ,
4T25 ,P25 ,H25 ,S25 ,T28 ,P28 ,H28 ,S28 ,
5T29 ,P29 ,H29 ,S29 ,DUMS3 ,DUMS4 ,DUMS5 ,DUMS6 ,
6WAO ,WFO ,WG24 ,FAR24 ,ETAD ,DPOUC ,BYPASS ,DUMS7 ,
7TS29 ,PS28 ,V28 ,AM28 ,TS29 ,PS29 ,V29 ,AM29
COMMON / BACK/
XXT55 ,XP55 ,XH55 ,XS55 ,XT25 ,XP25 ,XH25 ,XS25 ,
XXWFB ,XWG55 ,XFAR55 ,XWFD ,XWG24 ,XFAR24 ,XXP1 ,DUMB ,
3T6 ,P6 ,H6 ,S6 ,T7 ,P7 ,H7 ,S7 ,
4T8 ,P8 ,H8 ,S8 ,T9 ,P9 ,H9 ,S9 ,
5WG6 ,WFA ,WG7 ,FAR7 ,ETAA ,DPAFT ,V55 ,V25 ,
6PS6 ,V6 ,AM6 ,TS7 ,PS7 ,V7 ,AM7 ,AM25 ,
7TS8 ,PS8 ,V8 ,AM8 ,TS9 ,PS9 ,V9 ,AM9 ,
8VA ,FRD ,VJD ,FGMD ,VJM ,FGMM ,FGPD ,FGPM ,
9FGM ,FGP ,WFT ,WGT ,FART ,FG ,FN ,SFC
EQUIVALENCE (ANS1(1),PCNFGU),(ANS2(1),T1),(ANS3(1),XP1)

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EQUIVALENCE (ANS4(1),XT55)
DATA AWORD1,AWORD2/6HOUTPUT,6HCOMMON/
DATA (W(1,I),I=1,4)/6HSUBSON,6HIC C-D,6H NOZZL,6HE /
DATA (W(2,I),I=1,4)/6HSHOCK ,6HINSIDE,6H C-D N,6HOZZLE /
DATA (W(3,I),I=1,4)/6HSHOCK ,6HOUTSID,6HE C-D ,6HNOZZLE/
DATA (W(4,I),I=1,4)/6HSUBSON,6HIC CON,6HVERG. ,6HNOZZLE/
DATA (W(5,I),I=1,4)/6HSONIC ,6HCONVER,6HGENT N,6HOZZLE /
WORD=AWORD1
IF(IDES.EQ.1) GO TO 4
IF(IDBURN.GT.0) GO TO 2
IF(IAFTBN.GT.0) GO TO 1
WRITE(6,100) WORD,AM,ALTP,T4,ETAR
GO TO 3
1 WRITE(6,101) WORD,AM,ALTP,T4,T7,ETAR
GO TO 3
2 WRITE(6,102) WORD,AM,ALTP,T4,T24,ETAR
3 CALL CONOUT(2)
4 WRITE(6,104) (W(IMSHOC,I),I=1,4),FG,FN,SFC
IF(IGASHX.GT.0) GO TO 5
WRITE(6,105) (W(IDSHOC,I),I=1,4)
5 WRITE(6,106) LOOPER
WORD=AWORD2
WRITE(6,107) WORD,ZF,PCNF,ZI,PCNI,ZC,PCNC,T4,MODE
WRITE(6,108)
WRITE(6,109) (ANS1(I),I=1,96)
WRITE(6,108)
WRITE(6,109) (ANS2(I),I=1,121)
WRITE(6,108)
WRITE(6,109) (ANS3(I),I=1,56)
WRITE(6,108)
WRITE(6,109) (ANS4(I),I=1,72)
IF(IDES.EQ.1) GO TO 6
A8=A8SAV
A9=A9SAV
A28=A28SAV
A29=A29SAV
IF(IDUMP.NE.2) GO TO 6
WRITE(6,110)
CALL SYG(2)
6 CALL ENGBAL
RETURN
100 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,25X7H ETAR=,F7.4)
101 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,5X7H T7=,F8.2,5X7H ETAR=,F7.4)
102 FORMAT(1H0,A6,14X7H AM=,F7.3,6X7H ALTP=,F7.0,
16X7H T4=,F8.2,5X7H T24=,F8.2,5X7H ETAR=,F7.4)
104 FORMAT(6H0MAIN ,4A6,9X3HFG=,F9.2,18X3HFN=,F9.2,18X4HSFC=,F8.5)
105 FORMAT(6H0UCT ,4A6)
106 FORMAT(16H1CCNVERGED AFTER,I4,6H LOOPS,/,1H0)
107 FORMAT(1H ,A6,9X,7E15.6,I4)
108 FORMAT(1H )
109 FORMAT(1H ,8E15.6)
110 FORMAT(1H1)
END

```

```

SUBROUTINE MAPBAC(MAP,MAPGO,TFFS,TFF,CNS,CN,PCN,T,MODE,IGO,NUM)
DATA WH,WI,WL,WT,WS/6H H.P. ,6H I.P. ,6H L.P. ,6H TFF ,6HSPEED /
1  FORMAT(1H0,A6,12HTURBINE MAP ,A6,4HWAS=,E13.6,10H AND NOW=,E13.6,
16H$$$$$)
2  FORMAT(1H0,A6,A6,22HWAS ALSO CHANGED FROM ,E13.6,5H TO ,E13.6,
16H$$$$$)
  IF (NUM.GT.0) GO TO 3
  NUMH=0
  NUMI=0
  NUML=0
3  IGO=MAPGO+3*(MAP-1)
  GO TO (100,200,300,400,500,500.700,500.900),IGO
C *** HIGH PRESSURE TURBINE
100  TFF=TFF+0.1*(TFF-TFFS)
  WRITE (8,1) WH,WT,TFFS,TFF
  RETURN
200  CN=CN+0.05*(CN-CNS)
  IF (MODE.NE.1) PCN=PCN*(CN/CNS)
  IF (MODE.EQ.1) T =T *(CNS/CN)**2
  WRITE (8,1) WH,WS,CNS,CN
  IF (NUMH.GT.2) GO TO 210
  NUMH=1
  NUMH=NUMH+1
  RETURN
210  DELCN=CN-CNS
  IF (DELCN.GE.0.) RETURN
  TFF=TFF*(1.+DELCN/CN)
  WRITE (8,2) WH,WT,TFFS,TFF
  RETURN
300  TFF=TFF+0.1*(TFF-TFFS)
  WRITE (8,1) WH,WT,TFFS,TFF
  GO TO 200
C *** LOW PRESSURE TURBINE
400  TFF=TFF+0.1*(TFF-TFFS)
  WRITE (8,1) WL,WT,TFFS,TFF
  RETURN
500  CN=CN+0.05*(CN-CNS)
  IF (MODE.NE.3) PCN=PCN*(CN/CNS)
  IF (MODE.EQ.3) T =T *(CNS/CN)
  WRITE (8,1) WL,WS,CNS,CN
  IF (NUML.GT.2) GO TO 510
  NUML=1
  NUML=NUML+1
  RETURN
510  DELCN=CN-CNS
  IF (DELCN.GE.0.) RETURN
  TFF=TFF*(1.+DELCN/CN)
  WRITE (8,2) WL,WT,TFFS,TFF
  RETURN
600  TFF=TFF+0.1*(TFF-TFFS)
  WRITE (8,1) WL,WT,TFFS,TFF
  GO TO 500
C *** INTERMEDIATE SPOOL TURBINE
700  TFF=TFF+0.1*(TFF-TFFS)
  WRITE (8,1) WI,WT,TFFS,TFF

```

```

      RETURN
800  CN=CN+0.05*(CN-CNS)
      IF (MODE.NE.3) PCN=PCN*(CN/CNS)
      IF (MODE.EQ.3) T=T*(CNS/CN)
      WRITE(8,1) WI,WS,CNS,CN
      IF (NUMI.GT.2) GO TO 810
      NUM=1
      NUMI=NUMI+1
      RETURN
810  DELCN=CN-CNS
      IF (DELCN.GE.0.) RETURN
      TFF=TFF*(1.+DELCN/CN)
      WRITE(8,2) IT,WT,TFFS,TFF
      RETURN
900  TFF=TFF+0.1*(TFF-TFFS)
      WRITE(8,1) WI,WT,TFFS,TFF
      GO TO 800
      END

```



```

SUBROUTINE THERMO(PX,HX,TX,SX,AMX,L,FAR,K)
FX=0.
IF (L.EQ.1) FX=FAR
IF (K.EQ.1) GC TC 1
CALL PROCOM(FX,TX,CS,AK,CP,R,PHI,HX)
GO TO 3
1 TX=4.*HX
DO 2 I=1,15
CALL FROCOM(FX,TX,CS,AK,CP,R,PHI,t)
DELH=HX-H
IF (ABS(DELH).LE.0.00001*HX) GO TO 3
2 TX=TX+4.*DELH
WRITE(8,100)
100 FORMAT(31HONG CONVERGENCE IN THERMO$$$)
3 SX=PHI-R*ALOG(PX)
AMX=1.986375/R
RETURN
END

```

```

SUBROUTINE RAM(AM,ETAR)
IF (AM.GT.1.) GO TO 2
ETAR=1.
1  RETURN
2  IF (AM.GT.5.) GO TO 3
   ETAR=1.-0.075*((AM-1.)**1.35)
   GO TO 1
3  ETAR=800./((AM**4)+935.)
   GO TO 1
END

```

```

FUNCTION GUESS(M,T,TD,P,PD,N,WD,C,DD,VD)
IF (M.EQ.0) GUESS=VD*((T/TD)**1.60)*((DD/D)**0.50)
IF (M.EQ.1) GUESS=VD*((P/PD)**1.80)*((DD/D)**0.33)
IF (M.EQ.2) GUESS=VD*((W/WD)**0.33)*((DD/D)**1.00)
IF (M.EQ.3) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)
IF (M.EQ.4) GUESS=VD*((W/WD)**0.00)*((P/PD)**0.50)
IF (M.EQ.5) GUESS=VD*((T/TD)**1.10)*((DD/D)**0.60)
IF (M.EQ.6) GUESS=VD*((P/PD)**1.00)*((D/DD)**0.25)
IF (M.EQ.7) GUESS=VD*((P/PD)**0.62)*((D/DD)**0.31)
IF (M.EQ.8) GUESS=VD*((T/TD)**1.2)*DD/D
IF (M.EQ.9) GUESS=VD*P/PD*((D/DD)**1.5)
RETURN
END

```

```

BLOCK DATA CMEDAT
COMMON / COMB/PSI(15),DELT(15,15),ETA(15,15),N,NP(15)
DATA N,NP/15,15*15/
DATA PSI/4.9116,9.8232,14.735,19.646,24.558,29.470,34.381,
139.293,44.207,73.674,100.,200.,300.,400.,500./
DATA DELT/15*200.,15*300.,15*400.,15*500.,15*600.,15*700.,15*800.,
115*900.,15*1000.,15*1100.,15*1200.,15*1300.,15*1400.,15*1500.,
215*1600./
DATA ETA/
1.600,.726,.777,.806,.826,.843,.855,.865,7*.870,
2.758,.825,.858,.875,.888,.898,.906,.912,.914,6*.915,
3.868,.893,.911,.925,.935,.942,.947,.951,7*.953,
4.925,.936,.946,.955,.963,.969,.974,.977,.978,6*.979,
5.960,.966,.972,.977,.982,.985,.990,.992,.993,6*.995,
6.988,.991,.992,.994,.995,.997,.998,8*.999,
78*1.00,7*.999,120*1.00/
END

```

BLOCK DATA FANDAT  
 C THIS IS A GENERALIZED FAN MAP FOR UNREALISTIC SUPERSONIC ENGINE  
 COMMON/ FAN/CN(15),PR(15,15),HAC(15,15),ETA(15,15),N,NP(15)  
 DATA N,NP/10,6,3\*7,5\*10,8,5\*0/  
 DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5\*0./  
 DATA (PR(1,J) ,J=1,6 )/  
 C1.0000 ,1.0120 ,1.0280 ,1.0384 ,1.0448 ,1.0480 /  
 DATA (HAC(1,J) ,J=1,6 )/  
 C243.60 ,229.80 ,199.80 ,166.80 ,133.20 ,86.400 /  
 DATA (ETA(1,J) ,J=1,6 )/  
 C0.7559 ,0.7612 ,0.7665 ,0.7559 ,0.7251 ,0.6415 /  
 DATA (PR(2,J) ,J=1,7 )/  
 C1.0000 ,1.0200 ,1.0400 ,1.0584 ,1.0752 ,1.0920 ,1.1000 /  
 DATA (HAC(2,J) ,J=1,7 )/  
 C286.80 ,270.00 ,253.20 ,233.40 ,209.40 ,183.60 ,156.60 /  
 DATA (ETA(2,J) ,J=1,7 )/  
 C0.7559 ,0.7762 ,0.7920 ,0.7973 ,0.8026 ,0.7762 ,0.7401 /  
 DATA (PR(3,J) ,J=1,7 )/  
 C1.0000 ,1.0256 ,1.0512 ,1.0800 ,1.1160 ,1.1320 ,1.1480 /  
 DATA (HAC(3,J) ,J=1,7 )/  
 C333.60 ,322.80 ,310.20 ,291.60 ,259.80 ,240.00 ,213.60 /  
 DATA (ETA(3,J) ,J=1,7 )/  
 C0.7506 ,0.7762 ,0.8026 ,0.8281 ,0.8439 ,0.8281 ,0.7662 /  
 DATA (PR(4,J) ,J=1,7 )/  
 C1.0000 ,1.0368 ,1.0880 ,1.1240 ,1.1600 ,1.1896 ,1.1952 /  
 DATA (HAC(4,J) ,J=1,7 )/  
 C383.40 ,376.20 ,358.20 ,340.20 ,313.20 ,276.60 ,266.40 /  
 DATA (ETA(4,J) ,J=1,7 )/  
 C0.7454 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.8281 ,0.8078 /  
 DATA (PR(5,J) ,J=1,10)/  
 C1.0000 ,1.0640 ,1.1184 ,1.1480 ,1.1840 ,1.2096 ,1.2176 ,  
 C1.2240 ,1.2440 ,1.2672 /  
 DATA (HAC(5,J) ,J=1,10)/  
 C439.80 ,436.80 ,428.40 ,420.60 ,406.80 ,393.60 ,388.20 ,  
 C383.40 ,368.40 ,342.60 /  
 DATA (ETA(5,J) ,J=1,10)/  
 C0.7251 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9038 ,  
 C0.9011 ,0.8800 ,0.8281 /  
 DATA (PR(6,J) ,J=1,10)/  
 C1.0000 ,1.1000 ,1.1600 ,1.2000 ,1.2280 ,1.2552 ,1.2720 ,  
 C1.2864 ,1.3024 ,1.3320 /  
 DATA (HAC(6,J) ,J=1,10)/  
 C499.81 ,499.80 ,493.20 ,485.40 ,476.40 ,466.80 ,456.60 ,  
 C448.20 ,433.20 ,406.80 /  
 DATA (ETA(6,J) ,J=1,10)/  
 C0.6882 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9108 ,  
 C0.9011 ,0.8800 ,0.8272 /  
 DATA (PR(7,J) ,J=1,10)/  
 C1.0000 ,1.0760 ,1.1520 ,1.2192 ,1.2600 ,1.2896 ,1.3312 ,  
 C1.3616 ,1.3912 ,1.4000 /  
 DATA (HAC(7,J) ,J=1,10)/  
 C566.41 ,566.40 ,566.39 ,559.80 ,553.20 ,544.80 ,528.60 ,  
 C509.40 ,483.60 ,474.00 /  
 DATA (ETA(7,J) ,J=1,10)/  
 C0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,0.9011 ,

```

C0.8800 ,0.8281 ,0.8175 /
DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.0440 ,1.1352 ,1.2208 ,1.2944 ,1.3400 ,1.4000 ,
C1.4280 ,1.4480 ,1.4800 /
DATA (WAC(8,J) ,J=1,10)/
C633.61 ,633.60 ,633.59 ,633.00 ,625.80 ,616.80 ,600.00 ,
C586.80 ,576.60 ,553.20 /
DATA (ETA(8,J) ,J=1,10)/
C0.6002 ,0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,
C0.8589 ,0.8281 ,0.7867 /
DATA (PR(9,J) ,J=1,10)/
C1.0000 ,1.1040 ,1.2200 ,1.3240 ,1.4000 ,1.4480 ,1.5000 ,
C1.5336 ,1.5680 ,1.5840 /
DATA (WAC(9,J) ,J=1,10)/
C700.22 ,700.21 ,700.20 ,700.19 ,700.18 ,698.40 ,693.60 ,
C683.40 ,666.60 ,656.40 /
DATA (ETA(9,J) ,J=1,10)/
C0.5694 ,0.6415 ,0.7251 ,0.7762 ,0.8026 ,0.8078 ,0.8026 ,
C0.7762 ,0.7454 ,0.7251 /
DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,1.1632 ,1.3120 ,1.4000 ,1.4800 ,1.5400 ,1.5800 ,
C1.6600 /
DATA (WAC(10,J) ,J=1,8 )/
C750.03 ,750.02 ,750.01 ,750.00 ,749.99 ,749.98 ,749.40 ,
C736.80 /
DATA (ETA(10,J) ,J=1,8 )/
C0.5174 ,0.6415 ,0.7251 ,0.7559 ,0.7612 ,0.7506 ,0.7251 ,
C0.6415 /
END

```

BLOCK DATA INTDAT  
 THIS IS A GENERALIZED INT. COMP. MAP FOR UNREALISTIC SUPERSONIC EN  
 COMMON/ INT/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)  
 DATA N,NP/10,6,3\*7,5\*10,8,5\*0/  
 DATA CN/0.3,0.4,0.5,0.6,0.7,0.8,0.9,1.0,1.1,1.2,5\*0./  
 DATA (PR(1,J) ,J=1,6 )/  
 C1.0000 ,1.0180 ,1.0420 ,1.0576 ,1.0672 ,1.0720 /  
 DATA (WAC(1,J) ,J=1,6 )/  
 C121.80 ,114.90 ,99.900 ,83.400 ,66.600 ,43.200 /  
 DATA (ETA(1,J) ,J=1,6 )/  
 C0.7559 ,0.7612 ,0.7665 ,0.7559 ,0.7251 ,0.6415 /  
 DATA (PR(2,J) ,J=1,7 )/  
 C1.0000 ,1.0300 ,1.0600 ,1.0876 ,1.1128 ,1.1380 ,1.1500 /  
 DATA (WAC(2,J) ,J=1,7 )/  
 C143.40 ,135.00 ,126.60 ,116.70 ,104.70 ,91.800 ,78.300 /  
 DATA (ETA(2,J) ,J=1,7 )/  
 C0.7559 ,0.7762 ,0.7420 ,0.7973 ,0.8026 ,0.7762 ,0.7401 /  
 DATA (PR(3,J) ,J=1,7 )/  
 C1.0000 ,1.0384 ,1.0768 ,1.1200 ,1.1740 ,1.1980 ,1.2220 /  
 DATA (WAC(3,J) ,J=1,7 )/  
 C166.80 ,161.40 ,155.10 ,145.80 ,129.90 ,120.00 ,106.80 /  
 DATA (ETA(3,J) ,J=1,7 )/  
 C0.7506 ,0.7762 ,0.8026 ,0.8281 ,0.8439 ,0.8281 ,0.7762 /  
 DATA (PR(4,J) ,J=1,7 )/  
 C1.0000 ,1.0552 ,1.1320 ,1.1860 ,1.2400 ,1.2844 ,1.2928 /  
 DATA (WAC(4,J) ,J=1,7 )/  
 C191.70 ,188.10 ,179.10 ,170.10 ,156.60 ,138.30 ,133.20 /  
 DATA (ETA(4,J) ,J=1,7 )/  
 C0.7454 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.8281 ,0.8078 /  
 DATA (PR(5,J) ,J=1,10)/  
 C1.0000 ,1.0960 ,1.1776 ,1.2220 ,1.2760 ,1.3144 ,1.3264 ,  
 C1.3360 ,1.3660 ,1.4008 /  
 DATA (WAC(5,J) ,J=1,10)/  
 C219.90 ,218.40 ,214.20 ,210.30 ,203.40 ,196.80 ,194.10 ,  
 C191.70 ,184.20 ,171.30 /  
 DATA (ETA(5,J) ,J=1,10)/  
 C0.7251 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9038 ,  
 C0.9011 ,0.8800 ,0.8281 /  
 DATA (PR(6,J) ,J=1,10)/  
 C1.0000 ,1.1500 ,1.2400 ,1.3000 ,1.3420 ,1.3828 ,1.4080 ,  
 C1.4296 ,1.4536 ,1.4980 /  
 DATA (WAC(6,J) ,J=1,10)/  
 C249.91 ,249.90 ,246.60 ,242.70 ,238.20 ,233.40 ,228.30 ,  
 C224.10 ,216.60 ,203.40 /  
 DATA (ETA(6,J) ,J=1,10)/  
 C0.6882 ,0.7762 ,0.8281 ,0.8545 ,0.8800 ,0.9011 ,0.9108 ,  
 C0.9011 ,0.8800 ,0.8272 /  
 DATA (PR(7,J) ,J=1,10)/  
 C1.0000 ,1.1140 ,1.2280 ,1.3288 ,1.3900 ,1.4344 ,1.4968 ,  
 C1.5424 ,1.5868 ,1.6000 /  
 DATA (WAC(7,J) ,J=1,10)/  
 C283.21 ,283.20 ,283.19 ,279.90 ,276.60 ,272.40 ,264.30 ,  
 C254.70 ,241.80 ,237.00 /  
 DATA (ETA(7,J) ,J=1,10)/  
 C0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,0.9011 ,

```

C0.8800 ,0.8281 ,0.8175 /
  DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.0660 ,1.2028 ,1.3312 ,1.4416 ,1.5100 ,1.6000 ,
C1.6420 ,1.6720 ,1.7200 /
  DATA (WAC(8,J) ,J=1,10)/
C316.81 ,316.80 ,316.79 ,316.50 ,312.90 ,308.40 ,300.00 ,
C293.40 ,288.30 ,276.60 /
  DATA (ETA(8,J) ,J=1,10)/
C0.6002 ,0.6415 ,0.7251 ,0.7762 ,0.8281 ,0.8589 ,0.8800 ,
C0.8589 ,0.8281 ,0.7867 /
  DATA (PR(9,J) ,J=1,10)/
C1.0000 ,1.1560 ,1.3300 ,1.4860 ,1.6000 ,1.6720 ,1.7500 ,
C1.8004 ,1.8520 ,1.8760 /
  DATA (WAC(9,J) ,J=1,10)/
C350.12 ,350.11 ,350.10 ,350.09 ,350.08 ,349.20 ,346.80 ,
C341.70 ,333.30 ,328.20 /
  DATA (ETA(9,J) ,J=1,10)/
C0.5694 ,0.6415 ,0.7251 ,0.7762 ,0.8026 ,0.8078 ,0.8026 ,
C0.7762 ,0.7454 ,0.7251 /
  DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,1.2448 ,1.4680 ,1.6000 ,1.7200 ,1.8100 ,1.8700 ,
C1.9900 /
  DATA (WAC(10,J) ,J=1,8 )/
C375.03 ,375.02 ,375.01 ,375.00 ,374.99 ,374.98 ,374.70 ,
C368.40 /
  DATA (ETA(10,J) ,J=1,8 )/
C0.5174 ,0.6415 ,0.7251 ,0.7559 ,0.7612 ,0.7506 ,0.7251 ,
C0.6415 /
END

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C

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BLOCK DATA CMFOAT
THIS IS A GENERALIZED H.P. COMP. MAP FOR UNREALISTIC SUPERSONIC EN
COMMON/ COMP/CN(15),PR(15,15),WAC(15,15),ETA(15,15),N,NP(15)
DATA N,NP/10,2*6,2*8,4*10,2*8,5*0/
DATA CN/.562,.674,.787,.899,1.,1.034,1.067,1.124,1.236,1.292,5*0./
DATA (PR(1,J) ,J=1,6 )/
C1.0000 ,1.8400 ,2.4280 ,2.8690 ,3.8350 ,4.5490 /
DATA (WAC(1,J) ,J=1,6 )/
C51.000 ,50.200 ,49.500 ,48.800 ,46.700 ,44.500 /
DATA (ETA(1,J) ,J=1,6 )/
C0.5908 ,0.6218 ,0.6424 ,0.6527 ,0.6734 ,0.6424 /
DATA (PR(2,J) ,J=1,6 )/
C1.0000 ,1.9660 ,3.0930 ,3.9330 ,4.6890 ,5.5290 /
DATA (WAC(2,J) ,J=1,6 )/
C59.300 ,59.299 ,58.800 ,57.900 ,56.700 ,55.000 /
DATA (ETA(2,J) ,J=1,5 )/
C0.5908 ,0.6424 ,0.6940 ,0.7250 ,0.7456 ,0.7250 /
DATA (PR(3,J) ,J=1,8 )/
C1.0000 ,1.8400 ,2.6800 ,3.4080 ,4.5210 ,5.4450 ,6.3130 ,
C6.5230 /
DATA (WAC(3,J) ,J=1,8 )/
C70.000 ,70.000 ,69.999 ,69.500 ,68.800 ,67.900 ,66.400 ,
C65.700 /
DATA (ETA(3,J) ,J=1,8 )/
C0.5857 ,0.6424 ,0.6837 ,0.7250 ,0.7774 ,0.7929 ,0.7774 ,
C0.7697 /
DATA (PR(4,J) ,J=1,8 )/
C1.0000 ,2.0080 ,3.4290 ,4.6050 ,5.6970 ,6.6140 ,7.5380 ,
C7.9580 /
DATA (WAC(4,J) ,J=1,8 )/
C84.802 ,84.801 ,84.800 ,84.799 ,84.000 ,83.300 ,81.700 ,
C80.500 /
DATA (ETA(4,J) ,J=1,8 )/
C0.5805 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8290 ,0.8084 ,
C0.7929 /
DATA (PR(5,J) ,J=1,10)/
C1.0000 ,2.5190 ,3.9820 ,5.2770 ,6.4880 ,7.2020 ,8.0000 ,
C8.5670 ,9.3860 ,9.5960 /
DATA (WAC(5,J) ,J=1,10)/
C101.72 ,101.71 ,101.70 ,101.69 ,101.20 ,101.00 ,100.00 ,
C99.500 ,98.100 ,97.400 /
DATA (ETA(5,J) ,J=1,10)/
C0.5719 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8600 ,
C0.8394 ,0.8084 ,0.8058 /
DATA (PR(6,J) ,J=1,10)/
C1.0000 ,2.8550 ,4.2970 ,5.6130 ,6.9360 ,7.6220 ,8.5460 ,
C9.1340 ,9.9250 ,10.2190 /
DATA (WAC(6,J) ,J=1,10)/
C108.12 ,108.11 ,108.10 ,108.09 ,107.60 ,107.10 ,106.70 ,
C106.00 ,104.50 ,104.00 /
DATA (ETA(6,J) ,J=1,10)/
C0.5702 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8600 ,
C0.8394 ,0.8084 ,0.8041 /
DATA (PR(7,J) ,J=1,10)/
C1.0000 ,3.2610 ,4.7590 ,6.1170 ,7.4540 ,8.3080 ,9.2180 ,

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C9.6380 ,10.513 ,10.996 /
DATA (WAC(7,J) ,J=1,10)/
C114.52 ,114.51 ,114.50 ,114.49 ,114.48 ,114.30 ,113.60 ,
C113.30 ,112.60 ,112.40 /
DATA (ETA(7,J) ,J=1,10)/
C0.5599 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8394 ,0.8497 ,
C0.8394 ,0.8084 ,0.7981 /
DATA (PR(8,J) ,J=1,10)/
C1.0000 ,1.6860 ,3.8490 ,5.4660 ,6.8660 ,8.3710 ,8.9660 ,
C9.8830 ,10.912 ,11.815 /
DATA (WAC(8,J) ,J=1,10)/
C122.93 ,122.92 ,122.91 ,122.90 ,122.89 ,122.88 ,122.60 ,
C122.10 ,121.70 ,120.70 /
DATA (ETA(8,J) ,J=1,10)/
C0.5392 ,0.5702 ,0.6424 ,0.7250 ,0.7774 ,0.8084 ,0.8239 ,
C0.8394 ,0.8084 ,0.7774 /
DATA (PR(9,J) ,J=1,8 )/
C1.0000 ,4.3530 ,7.6220 ,10.219 ,11.059 ,11.899 ,13.159 ,
C13.656 /
DATA (WAC(9,J) ,J=1,8 )/
C139.82 ,139.81 ,139.80 ,139.79 ,139.78 ,139.50 ,139.30 ,
C139.00 /
DATA (ETA(9,J) ,J=1,8 )/
C0.4764 ,0.6011 ,0.7250 ,0.7774 ,0.7826 ,0.7774 ,0.7250 ,
C0.6992 /
DATA (PR(10,J) ,J=1,8 )/
C1.0000 ,3.7650 ,6.4810 ,9.1760 ,10.219 ,11.479,12.711 ,
C14.412 /
DATA (WAC(10,J) ,J=1,8 )/
C146.24 ,146.23 ,146.22 ,146.21 ,146.20 ,146.19 ,146.18 ,
C146.17 /
DATA (ETA(10,J) ,J=1,8 )/
C0.4661 ,0.5702 ,0.6424 ,0.7250 ,0.7508 ,0.7508 ,0.7250 ,
C0.6424 /
END

```

BLOCK DATA LTURB  
 THIS IS A GENERALIZED L.P. TURBINE MAP  
 COMMON/ LTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)  
 DATA N,NP/11,9\*15,12,9,4\*0/  
 DATA TFF/ 88.470, 102.80, 116.84, 129.33, 141.05, 145.73, 150.00,  
 1 153.35, 156.41, 159.78, 163.17, 4\*0./  
 DATA ( CN( 1,J),J=1,15)/  
 C0.3682, 0.5336, 0.7365, 0.9754, 1.2146, 1.4173, 1.6201, 1.7673,  
 C2.0247, 2.2827, 2.4665, 2.6137, 2.8166, 2.9456, 3.3138/  
 DATA ( DH( 1,J),J=1,15)/  
 C0.0010, 0.0026, 0.0035, 0.0044, 0.0051, 0.0056, 0.0059, 0.0061,  
 C0.0062, 0.0061, 0.0057, 0.0053, 0.0044, 0.0035, 0.0001/  
 DATA (ETA( 1,J),J=1,15)/  
 C0.7120, 0.7300, 0.7472, 0.7300, 0.7140, 0.7008, 0.6850, 0.6730,  
 C0.6452, 0.6200, 0.6000, 0.5750, 0.5310, 0.5000, 0.3850/  
 DATA ( CN( 2,J),J=1,15)/  
 C0.3682, 0.5518, 0.7919, 1.0672, 1.2882, 1.4446, 1.6937, 1.8954,  
 C2.0619, 2.2273, 2.3747, 2.6229, 2.8720, 3.0555, 3.3138/  
 DATA ( DH( 2,J),J=1,15)/  
 C0.0026, 0.0039, 0.0054, 0.0069, 0.0080, 0.0087, 0.0096, 0.0101,  
 C0.0104, 0.0107, 0.0108, 0.0106, 0.0101, 0.0094, 0.0077/  
 DATA (ETA( 2,J),J=1,15)/  
 C0.8000, 0.8100, 0.8200, 0.8300, 0.8300, 0.8298, 0.8100, 0.8000,  
 C0.7850, 0.7600, 0.7450, 0.7000, 0.6800, 0.6450, 0.5900/  
 DATA ( CN( 3,J),J=1,15)/  
 C0.3682, 0.5911, 0.8655, 1.0764, 1.2519, 1.4354, 1.6201, 1.8409,  
 C2.0247, 2.2455, 2.4302, 2.5956, 2.7791, 3.0555, 3.3138/  
 DATA ( DH( 3,J),J=1,15)/  
 C0.0031, 0.0051, 0.0071, 0.0087, 0.0099, 0.0111, 0.0122, 0.0134,  
 C0.0143, 0.0152, 0.0157, 0.0162, 0.0166, 0.0167, 0.0164/  
 DATA (ETA( 3,J),J=1,15)/  
 C0.8800, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,  
 C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/  
 DATA ( CN( 4,J),J=1,15)/  
 C0.3682, 0.4237, 0.6810, 0.8837, 1.1047, 1.2882, 1.5090, 1.7482,  
 C2.0429, 2.2091, 2.3747, 2.6047, 2.8720, 3.1291, 3.3138/  
 DATA ( DH( 4,J),J=1,15)/  
 C0.0033, 0.0038, 0.0061, 0.0078, 0.0096, 0.0110, 0.0126, 0.0141,  
 C0.0159, 0.0166, 0.0174, 0.0183, 0.0191, 0.0195, 0.0197/  
 DATA (ETA( 4,J),J=1,15)/  
 C0.7995, 0.8000, 0.8400, 0.8600, 0.8680, 0.8730, 0.8800, 0.8830,  
 C0.8830, 0.8830, 0.8800, 0.8740, 0.8600, 0.8350, 0.8200/  
 DATA ( CN( 5,J),J=1,15)/  
 C0.3682, 0.5065, 0.7365, 0.9754, 1.2882, 1.5647, 1.7301, 1.9690,  
 C2.0983, 2.2637, 2.4332, 2.6691, 2.9456, 3.1846, 3.3138/  
 DATA ( DH( 5,J),J=1,15)/  
 C0.0036, 0.0049, 0.0071, 0.0092, 0.0119, 0.0141, 0.0155, 0.0172,  
 C0.0181, 0.0192, 0.0202, 0.0214, 0.0226, 0.0235, 0.0239/  
 DATA (ETA( 5,J),J=1,15)/  
 C0.7750, 0.8000, 0.8480, 0.8600, 0.8750, 0.8900, 0.8912, 0.8940,  
 C0.8955, 0.8970, 0.8961, 0.8900, 0.8790, 0.8671, 0.8600/  
 DATA ( CN( 6,J),J=1,15)/  
 C0.3682, 0.6164, 0.8372, 1.1047, 1.2882, 1.5283, 1.7482, 1.9509,  
 C2.2133, 2.4302, 2.6510, 2.8619, 3.1384, 3.2584, 3.3138/  
 DATA ( DH( 6,J),J=1,15)/

C0.0038, 0.0064, 0.0087, 0.0113, 0.0130, 0.0152, 0.0171, 0.0187,  
 C0.0209, 0.0226, 0.0244, 0.0259, 0.0286, 0.0303, 0.0319/  
 DATA (ETA( 6,J),J=1,15)/  
 C0.7600, 0.8000, 0.8450, 0.8600, 0.8730, 0.8900, 0.8950, 0.9000,  
 C0.9005, 0.9010, 0.9004, 0.9000, 0.8980, 0.8800, 0.8735/  
 DATA ( CN( 7,J),J=1,15)/  
 C0.3682, 0.7728, 1.0129, 1.2659, 1.4729, 1.6785, 1.8409, 2.0247,  
 C2.1901, 2.3000, 2.3929, 2.5038, 2.5583, 2.6137, 2.6319/  
 DATA ( DH( 7,J),J=1,15)/  
 C0.0044, 0.0089, 0.0115, 0.0141, 0.0162, 0.0181, 0.0197, 0.0216,  
 C0.0235, 0.0250, 0.0265, 0.0284, 0.0296, 0.0314, 0.0329/  
 DATA (ETA( 7,J),J=1,15)/  
 C0.7310, 0.8000, 0.8300, 0.8600, 0.8750, 0.8900, 0.8930, 0.8975,  
 C0.8999, 0.9000, 0.8980, 0.8937, 0.8900, 0.8799, 0.8710/  
 DATA ( CN( 8,J),J=1,15)/  
 C0.3682, 0.6072, 0.7919, 0.9754, 1.2337, 1.4548, 1.6383, 1.8409,  
 C1.9509, 2.0801, 2.1537, 2.2091, 2.2637, 2.3009, 2.3051/  
 DATA ( DH( 8,J),J=1,15)/  
 C0.0048, 0.0078, 0.0102, 0.0124, 0.0153, 0.0177, 0.0201, 0.0226,  
 C0.0242, 0.0261, 0.0274, 0.0285, 0.0299, 0.0314, 0.0321/  
 DATA (ETA( 8,J),J=1,15)/  
 C0.7100, 0.7450, 0.7680, 0.8000, 0.8380, 0.8600, 0.8712, 0.8780,  
 C0.8800, 0.8775, 0.8760, 0.8722, 0.8660, 0.8600, 0.8480/  
 DATA ( CN( 9,J),J=1,15)/  
 C0.3682, 0.5518, 0.6629, 0.8282, 1.0129, 1.1691, 1.2337, 1.3809,  
 C1.5283, 1.6201, 1.7482, 1.8409, 1.8954, 1.9147, 1.9237/  
 DATA ( DH( 9,J),J=1,15)/  
 C0.0054, 0.0080, 0.0096, 0.0119, 0.0141, 0.0160, 0.0169, 0.0188,  
 C0.0209, 0.0223, 0.0244, 0.0263, 0.0279, 0.0289, 0.0303/  
 DATA (ETA( 9,J),J=1,15)/  
 C0.6780, 0.7000, 0.7125, 0.7350, 0.7690, 0.8000, 0.8060, 0.8225,  
 C0.8395, 0.8450, 0.8470, 0.8445, 0.8330, 0.8235, 0.8080/  
 DATA ( CN(10,J),J=1,12)/  
 C0.3682, 0.4782, 0.6447, 0.7546, 0.8655, 0.9754, 1.1047, 1.2015,  
 C1.2701, 1.3073, 1.3365, 1.3407/  
 DATA ( DH(10,J),J=1,12)/  
 C0.0061, 0.0078, 0.0104, 0.0122, 0.0139, 0.0157, 0.0181, 0.0201,  
 C0.0217, 0.0230, 0.0244, 0.0251/  
 DATA (ETA(10,J),J=1,12)/  
 C0.6380, 0.6550, 0.6700, 0.6850, 0.7000, 0.7110, 0.7180, 0.7180,  
 C0.7170, 0.7140, 0.7000, 0.6890/  
 DATA ( CN(11,J),J=1,9 )/  
 C0.3682, 0.4418, 0.5518, 0.6447, 0.7365, 0.8282, 0.8837, 0.9391,  
 C0.9715/  
 DATA ( DH(11,J),J=1,9 )/  
 C0.0069, 0.0086, 0.0106, 0.0123, 0.0141, 0.0159, 0.0172, 0.0186,  
 C0.0201/  
 DATA (ETA(11,J),J=1,9 )/  
 C0.6000, 0.6000, 0.6120, 0.6170, 0.6210, 0.6258, 0.6250, 0.6230,  
 C0.6089/  
 END

C

## BLOCK DATA ITURB

THIS IS A GENERALIZED I.P. TURBINE MAP

COMMON/ ITURE/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)

DATA N,NP/11,9\*15,12,9,4\*0/

DATA TFF/ 39.670, 82.236, 93.468, 103.464, 112.836, 116.580,  
1 120.000, 122.68, 125.12, 127.82, 130.536, 4\*0./

DATA ( CN( 1,J),J=1,15)/

C0.3522, 0.5104, 0.7044, 0.9330, 1.1618, 1.3556, 1.5497, 1.6905,

C1.9367, 2.1835, 2.3593, 2.5001, 2.6941, 2.8175, 3.1698/

DATA ( DH( 1,J),J=1,15)/

C0.0016, 0.0023, 0.0031, 0.0038, 0.0045, 0.0049, 0.0052, 0.0054,

C0.0055, 0.0054, 0.0051, 0.0047, 0.0038, 0.0031, 0.0001/

DATA (ETA( 1,J),J=1,15)/

C0.7120, 0.7300, 0.7472, 0.7300, 0.7140, 0.7000, 0.6850, 0.6730,

C0.6452, 0.6200, 0.6000, 0.5750, 0.5310, 0.5000, 0.3850/

DATA ( CN( 2,J),J=1,15)/

C0.3522, 0.5278, 0.7575, 1.0208, 1.2322, 1.3818, 1.6201, 1.8130,

C1.9723, 2.1305, 2.2715, 2.5089, 2.7471, 2.9227, 3.1698/

DATA ( DH( 2,J),J=1,15)/

C0.0023, 0.0035, 0.0047, 0.0061, 0.0070, 0.0076, 0.0084, 0.0089,

C0.0092, 0.0094, 0.0095, 0.0093, 0.0089, 0.0083, 0.0068/

DATA (ETA( 2,J),J=1,15)/

C0.8000, 0.8100, 0.8200, 0.8300, 0.8300, 0.8290, 0.8100, 0.8000,

C0.7800, 0.7600, 0.7450, 0.7000, 0.6800, 0.6450, 0.5900/

DATA ( CN( 2,J),J=1,15)/

C0.3522, 0.5654, 0.8279, 1.0296, 1.1975, 1.3730, 1.5497, 1.7609,

C1.9367, 2.1479, 2.3245, 2.4827, 2.6583, 2.9227, 3.1698/

DATA ( DH( 2,J),J=1,15)/

C0.0027, 0.0045, 0.0063, 0.0076, 0.0087, 0.0098, 0.0107, 0.0118,

C0.0126, 0.0134, 0.0139, 0.0142, 0.0146, 0.0147, 0.0145/

DATA (ETA( 2,J),J=1,15)/

C0.8000, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,

C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/

DATA ( CN( 3,J),J=1,15)/

C0.3522, 0.5654, 0.8279, 1.0296, 1.1975, 1.3730, 1.5497, 1.7609,

C1.9367, 2.1479, 2.3245, 2.4827, 2.6583, 2.9227, 3.1698/

DATA ( DH( 3,J),J=1,15)/

C0.0027, 0.0045, 0.0063, 0.0076, 0.0087, 0.0098, 0.0107, 0.0118,

C0.0126, 0.0134, 0.0139, 0.0142, 0.0146, 0.0147, 0.0145/

DATA (ETA( 3,J),J=1,15)/

C0.8000, 0.8300, 0.8600, 0.8630, 0.8670, 0.8700, 0.8720, 0.8720,

C0.8700, 0.8670, 0.8600, 0.8500, 0.8300, 0.8000, 0.7600/

DATA ( CN( 4,J),J=1,15)/

C0.3522, 0.4052, 0.6514, 0.8452, 1.0567, 1.2322, 1.4434, 1.6722,

C1.9540, 2.1131, 2.2715, 2.4915, 2.7471, 2.9931, 3.1698/

DATA ( DH( 4,J),J=1,15)/

C0.0029, 0.0034, 0.0054, 0.0069, 0.0084, 0.0097, 0.0111, 0.0124,

C0.0140, 0.0146, 0.0153, 0.0161, 0.0168, 0.0172, 0.0173/

DATA (ETA( 4,J),J=1,15)/

C0.7995, 0.8000, 0.8400, 0.8600, 0.8680, 0.8730, 0.8800, 0.8830,

C0.8835, 0.8830, 0.8800, 0.8740, 0.8600, 0.8350, 0.8200/

DATA ( CN( 5,J),J=1,15)/

C0.3522, 0.4844, 0.7044, 0.9330, 1.2322, 1.4967, 1.6548, 1.8834,

C2.0071, 2.1652, 2.3274, 2.5531, 2.8175, 3.0461, 3.1698/

DATA ( DH( 5,J),J=1,15)/

C0.0031, 0.0043, 0.0062, 0.0081, 0.0105, 0.0124, 0.0136, 0.0152,  
 C0.0159, 0.0169, 0.0178, 0.0189, 0.0199, 0.0207, 0.0210/  
 DATA (ETA( 5,J),J=1,15)/  
 C0.7750, 0.8000, 0.8480, 0.8600, 0.8750, 0.8900, 0.8912, 0.8940,  
 C0.8955, 0.8970, 0.8961, 0.8900, 0.8790, 0.8671, 0.8600/  
 DATA ( CN( 6,J),J=1,15)/  
 C0.3522, 0.5896, 0.8008, 1.0567, 1.2322, 1.4619, 1.6722, 1.8660,  
 C2.1171, 2.3245, 2.5357, 2.7375, 3.0019, 3.1167, 3.1698/  
 DATA ( DH( 6,J),J=1,15)/  
 C0.0034, 0.0057, 0.0076, 0.0100, 0.0114, 0.0134, 0.0150, 0.0165,  
 C0.0184, 0.0199, 0.0214, 0.0228, 0.0251, 0.0267, 0.0290/  
 DATA (ETA( 6,J),J=1,15)/  
 C0.7600, 0.8000, 0.8150, 0.8300, 0.8730, 0.8900, 0.8950, 0.9000,  
 C0.9005, 0.9010, 0.9004, 0.9000, 0.8900, 0.8800, 0.8710/  
 DATA ( CN( 7,J),J=1,15)/  
 C0.3522, 0.7392, 0.9689, 1.2109, 1.4089, 1.6056, 1.7609, 1.9367,  
 C2.0948, 2.2000, 2.2689, 2.3949, 2.4471, 2.5001, 2.5175/  
 DATA ( DH( 7,J),J=1,15)/  
 C0.0038, 0.0078, 0.0101, 0.0124, 0.0142, 0.0159, 0.0173, 0.0190,  
 C0.0207, 0.0220, 0.0233, 0.0250, 0.0261, 0.0276, 0.0290/  
 DATA (ETA( 7,J),J=1,15)/  
 C0.7310, 0.8000, 0.8300, 0.8600, 0.8750, 0.8900, 0.8930, 0.8975,  
 C0.8999, 0.9000, 0.8980, 0.8937, 0.8900, 0.8799, 0.8710/  
 DATA ( CN( 8,J),J=1,15)/  
 C0.3522, 0.5808, 0.7575, 0.9330, 1.1801, 1.3915, 1.5671, 1.7609,  
 C1.8660, 1.9897, 2.0601, 2.1131, 2.1652, 2.2009, 2.2048/  
 DATA ( DH( 8,J),J=1,15)/  
 C0.0042, 0.0069, 0.0090, 0.0109, 0.0135, 0.0156, 0.0177, 0.0199,  
 C0.0213, 0.0230, 0.0241, 0.0251, 0.0263, 0.0276, 0.0283/  
 DATA (ETA( 8,J),J=1,15)/  
 C0.7100, 0.7450, 0.7680, 0.8000, 0.8380, 0.8600, 0.8712, 0.8780,  
 C0.8800, 0.8775, 0.8760, 0.8722, 0.8660, 0.8600, 0.8480/  
 DATA ( CN( 9,J),J=1,15)/  
 C0.3522, 0.5278, 0.6340, 0.7422, 0.9689, 1.1183, 1.1801, 1.3209,  
 C1.4619, 1.5497, 1.6722, 1.7609, 1.8130, 1.8315, 1.8401/  
 DATA ( DH( 9,J),J=1,15)/  
 C0.0047, 0.0070, 0.0084, 0.0104, 0.0124, 0.0141, 0.0148, 0.0166,  
 C0.0184, 0.0196, 0.0214, 0.0232, 0.0245, 0.0255, 0.0267/  
 DATA (ETA( 9,J),J=1,15)/  
 C0.6780, 0.7000, 0.7125, 0.7350, 0.7690, 0.8000, 0.8060, 0.8225,  
 C0.8395, 0.8450, 0.8470, 0.8445, 0.8330, 0.8235, 0.8080/  
 DATA ( CN(10,J),J=1,12)/  
 C0.3522, 0.4574, 0.6167, 0.7218, 0.8279, 0.9330, 1.0567, 1.1493,  
 C1.2148, 1.2505, 1.2784, 1.2824/  
 DATA ( DH(10,J),J=1,12)/  
 C0.0054, 0.0069, 0.0092, 0.0107, 0.0123, 0.0138, 0.0159, 0.0177,  
 C0.0191, 0.0202, 0.0214, 0.0221/  
 DATA (ETA(10,J),J=1,12)/  
 C0.6380, 0.6550, 0.6700, 0.6850, 0.7000, 0.7110, 0.7180, 0.7180,  
 C0.7170, 0.7140, 0.7000, 0.6890/  
 DATA ( CN(11,J),J=1,9 )/  
 C0.3522, 0.4226, 0.5278, 0.6167, 0.7044, 0.7422, 0.8452, 0.8983,  
 C0.9293/  
 DATA ( DH(11,J),J=1,9 )/  
 C0.0061, 0.0075, 0.0093, 0.0108, 0.0124, 0.0140, 0.0151, 0.0164,

```
CO.0177/  
DATA (ETA(11,J),J=1,9 )/  
CO.6000, 0.6000, 0.6120, 0.6170, 0.6210, 0.6258, 0.6250, 0.6230,  
CO.6009/  
END
```

C

## BLOCK DATA HTURB

THIS IS A GENERALIZED H.P. TURBINE MAP

COMMON / HTURB/TFF(15),CN(15,15),DH(15,15),ETA(15,15),N,NP(15)

DATA N,NP/10,9\*15,12,5\*0/

DATA TFF/ 39.670, 42.990, 47.460, 48.610, 49.175, 49.600,  
1 50.000, 50.425, 50.920, 51.575, 5\*0./

DATA ( CN( 1,J),J=1,15)/

C0.1872, 0.3372, 0.5156, 0.7128, 0.9382, 1.1442, 1.3138, 1.5382,  
C1.7264, 1.9324, 2.1500, 2.4058, 2.5892, 2.7862, 2.9460/

DATA ( DH( 1,J),J=1,15)/

C0.0032, 0.0057, 0.0084, 0.0108, 0.0133, 0.0152, 0.0164, 0.0174,  
C0.0179, 0.0176, 0.0167, 0.0144, 0.0120, 0.0082, 0.0034/

DATA (ETA( 1,J),J=1,15)/

C0.6219, 0.7078, 0.7868, 0.8090, 0.8090, 0.7963, 0.7779, 0.7422,  
C0.7078, 0.7635, 0.6058, 0.5309, 0.4773, 0.4045, 0.3034/

DATA ( CN( 2,J),J=1,15)/

C0.1872, 0.3942, 0.5814, 0.7128, 0.8442, 0.9804, 1.1068, 1.2754,  
C1.4450, 1.7068, 1.9696, 2.2706, 2.6970, 3.0960, 3.3774/

DATA ( DH( 2,J),J=1,15)/

C0.0036, 0.0080, 0.0113, 0.0136, 0.0156, 0.0175, 0.0192, 0.0212,  
C0.0228, 0.0248, 0.0260, 0.0261, 0.0241, 0.0188, 0.0128/

DATA (ETA( 2,J),J=1,15)/

C0.6068, 0.7078, 0.8090, 0.8292, 0.8363, 0.8393, 0.8368, 0.8302,  
C0.8254, 0.8090, 0.7696, 0.7078, 0.6066, 0.5056, 0.4197/

DATA ( CN( 3,J),J=1,15)/

C0.1872, 0.4362, 0.6568, 0.8726, 1.0696, 1.2382, 1.4638, 1.6882,  
C1.9696, 2.2138, 2.5520, 2.8650, 3.0392, 3.2348, 3.3774/

DATA ( DH( 3,J),J=1,15)/

C0.0046, 0.0100, 0.0144, 0.0184, 0.0216, 0.0240, 0.0268, 0.0292,  
C0.0316, 0.0331, 0.0344, 0.0346, 0.0340, 0.0324, 0.0312/

DATA (ETA( 3,J),J=1,15)/

C0.5764, 0.7078, 0.8090, 0.8494, 0.8543, 0.8515, 0.8494, 0.8409,  
C0.8262, 0.8090, 0.7579, 0.7078, 0.6652, 0.6068, 0.5865/

DATA ( CN( 4,J),J=1,15)/

C0.1872, 0.2550, 0.4784, 0.6942, 0.9148, 1.1442, 1.3862, 1.5618,  
C1.8010, 1.9794, 2.2794, 2.5138, 2.8334, 3.1422, 3.3774/

DATA ( DH( 4,J),J=1,15)/

C0.0052, 0.0068, 0.0120, 0.0164, 0.0204, 0.0244, 0.0280, 0.0304,  
C0.0336, 0.0356, 0.0388, 0.0412, 0.0441, 0.0472, 0.0494/

DATA (ETA( 4,J),J=1,15)/

C0.5643, 0.6068, 0.7078, 0.8090, 0.8494, 0.8596, 0.8596, 0.8575,  
C0.8535, 0.9494, 0.8363, 0.8262, 0.8090, 0.7797, 0.7584/

DATA ( CN( 5,J),J=1,15)/

C0.1872, 0.5000, 0.5254, 0.7500, 0.9754, 1.2754, 1.4824, 1.7638,  
C2.9450, 2.3362, 2.6450, 2.8700, 3.0754, 3.1520, 3.1618/

DATA ( DH( 5,J),J=1,15)/

C0.0056, 0.0088, 0.0144, 0.0192, 0.0236, 0.0288, 0.0321, 0.0360,  
C0.0400, 0.0444, 0.0496, 0.0540, 0.0596, 0.0640, 0.0661/

DATA (ETA( 5,J),J=1,15)/

C0.5562, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8696, 0.8662,  
C0.8615, 0.8559, 0.8520, 0.8494, 0.8494, 0.8532, 0.8570/

DATA ( CN( 6,J),J=1,15)/

C0.1872, 0.3568, 0.6196, 0.8628, 1.0932, 1.2852, 1.5010, 1.6882,  
C1.9138, 2.1246, 2.2706, 2.4226, 2.4950, 2.5372, 2.5558/

DATA ( DH( 6,J),J=1,15)/



C0.0068, 0.0120, 0.0192, 0.0252, 0.0300, 0.0340, 0.0384, 0.0421,  
 C0.0472, 0.0524, 0.0564, 0.0668, 0.0640, 0.0668, 0.0698/  
 DATA (ETA( 6,J),J=1,15)/  
 C0.5309, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8819, 0.8899,  
 C0.8940, 0.8969, 0.8975, 0.8937, 0.8968, 0.8937, 0.8896/  
 DATA ( CN( 7,J),J=1,15)/  
 C0.1872, 0.4314, 0.6844, 0.9568, 1.2010, 1.3834, 1.5108, 1.6186,  
 C1.7450, 1.8618, 1.9558, 2.0000, 2.0450, 2.0824, 2.1010/  
 DATA ( DH( 7,J),J=1,15)/  
 C0.0080, 0.0164, 0.0236, 0.0308, 0.0372, 0.0416, 0.0448, 0.0476,  
 C0.0510, 0.0544, 0.0576, 0.0600, 0.0624, 0.0660, 0.0700/  
 DATA (ETA( 7,J),J=1,15)/  
 C0.5062, 0.6068, 0.7078, 0.8090, 0.8494, 0.8697, 0.8797, 0.8899,  
 C0.8954, 0.9000, 0.9010, 0.9000, 0.8980, 0.8925, 0.8793/  
 DATA ( CN( 8,J),J=1,15)/  
 C0.1872, 0.4834, 0.7314, 0.8814, 1.0226, 1.1442, 1.2804, 1.3696,  
 C1.4638, 1.5950, 1.6746, 1.7450, 1.8010, 1.8156, 1.8196/  
 DATA ( DH( 8,J),J=1,15)/  
 C0.0088, 0.0196, 0.0272, 0.0316, 0.0356, 0.0392, 0.0432, 0.0460,  
 C0.0488, 0.0528, 0.0560, 0.0596, 0.0640, 0.0664, 0.0693/  
 DATA (ETA( 8,J),J=1,15)/  
 C0.5051, 0.6068, 0.7078, 0.7665, 0.8090, 0.8292, 0.8494, 0.8596,  
 C0.8697, 0.8808, 0.8848, 0.8848, 0.8788, 0.8697, 0.8590/  
 DATA ( CN( 9,J),J=1,15)/  
 C0.1872, 0.3372, 0.5344, 0.6754, 0.8068, 0.9196, 1.0128, 1.1254,  
 C1.2196, 1.3138, 1.3696, 1.4068, 1.4450, 1.4638, 1.4676/  
 DATA ( DH( 9,J),J=1,15)/  
 C0.0093, 0.0159, 0.0232, 0.0284, 0.0330, 0.0368, 0.0400, 0.0442,  
 C0.0480, 0.0524, 0.0556, 0.0580, 0.0612, 0.0648, 0.0668/  
 DATA (ETA( 9,J),J=1,15)/  
 C0.4909, 0.5380, 0.6068, 0.6573, 0.7078, 0.7463, 0.7776, 0.8090,  
 C0.8191, 0.8302, 0.8347, 0.8363, 0.8322, 0.8241, 0.8090/  
 DATA ( CN(10,J),J=1,12)/  
 C0.1872, 0.2814, 0.3804, 0.4686, 0.5628, 0.6382, 0.6892, 0.7362,  
 C0.7696, 0.8068, 0.8254, 0.8304/  
 DATA ( DH(10,J),J=1,12)/  
 C0.0132, 0.0180, 0.0228, 0.0268, 0.0314, 0.0352, 0.0380, 0.0412,  
 C0.0440, 0.0476, 0.0504, 0.0530/  
 DATA (ETA(10,J),J=1,12)/  
 C0.4257, 0.4747, 0.5056, 0.5359, 0.5683, 0.5941, 0.6168, 0.6178,  
 C0.6240, 0.6310, 0.6265, 0.6118/  
 END

SECTION VIII  
SAMPLE INPUT LISTING

The first section of data cards is the set of variables selected for output (controlled output). The second section is the Namelist input for running the desired points. The Namelist input consists of the following points: design point at sea-level static, a setup case for sea-level static afterburning, a sea-level static full afterburning point, several points in a subsonic power hook, a setup case for supersonic afterburning, and the supersonic full afterburning point.

T35  
P55  
T24  
P24  
W40  
W50  
W624  
FAR24  
T25  
P25  
DPDUC  
ETAC  
AM25  
AM55  
ETAF  
ETAI  
ETAC  
ETATHC  
ETATIF  
ETATLC  
T6  
P6  
PS6  
AM6  
V6  
W66  
T7  
WFA  
FAR7  
ETAA  
DPAFT  
PS8  
AM8  
V8  
PS9  
AM9  
V9  
PS28  
AM28  
V28  
PS20  
AM20  
V20  
BVFA55  
HPEXT  
WFT  
WGT  
VA  
FRD  
CVMN07  
VJM  
CVQNM07  
VJD  
FGM  
FGP

A55

A25

A6

A8

A9

A28

A29

THEEND

\$DATAIN ITITLE=1,IDES=1,IDUMP=1,IAMTP=0,MODE=3,  
IGASMY=2,TMCD=1,PS55=2.0,AM55=0.00,AM6=0.24,MPEXT=0.,NCZFLT=0,  
ITRYS=200.,TOLALL=.005,ZFDS=0.8333,ZIDS=0.8143,ZCDS=0.8143,  
PCNFDS=100.,ETAFDS=0.839,PCNIDS=100.,ETAIDS=0.853,PCNCPS=100.,  
ETACDS=0.853,DPCDS=0.047,DPODS=0.05,DTCCDS=1250.,ETA8CS=0.9875,  
PRFDS=2.2,PRIDS=3.435,PRCDS=7.435,MAFDS=754.,MAIDS=10.,  
TFHPDS=50.0,CNHPDS=2.0,ETHPDS=0.881,TFIPDS=120.,CNIPDS=2.2,ETIPDS=.881,  
TFLPDS=130.0,CNLPDS=2.3,ETLPDS=0.917,DPAFDS=0.0,CVMNOZ=0.980,  
DELF6=1.0,DELFN=1.0,DELSFC=1.0,PCBLF=0.0,PCBLI=0.0,PCBLC=0.0,  
T4DS=2860.,AM=0.0,ALTP=0.03

SLS TRISPL DESIGN POINT

\$DATAIN MODE=0,T4=2860.,ITITLE=13

SETUP FOR SLS AFTERBURNING

\$DATAIN T4=2860.,IAFTBN=1,T7=3700.,ETAA=0.875,ITITLE=13

SLS FULL AFTERBURNING

\$DATAIN MODE=3,PCNF=99.0,AM=0.75,ALTP=25000.,ITITLE=13

SUPERSONIC POWER HOOK

\$DATAIN PCNF=90.0

\$DATAIN PCNF=80.0

\$DATAIN PCNF=70.0

\$DATAIN MODE=1,PCNC=100.,AM=1.0,ALTP=50000.,ITITLE=13

SETUP FOR SUPERSONIC AFTERBURNING

\$DATAIN PCNC=100.,IAFTBN=1,T7=3700.,ETAA=0.85,ITITLE=13

SUPERSONIC FULL AFTERBURNING

\$DATAIN IEVD=13

## SECTION IX

### SAMPLE OUTPUT LISTING

The following are typical computer printouts, the first point is the design point and includes a page of correction (or scaling) factors and a page of values of variables in common. The other operating points consist of a page of output for each point. Not included for these points is a common dump, which follows each output page and is very similar to the common dump following the design point correction factors.

# POINT

FAN DESIGN	PRFCF=	.30001200E+01	ETAFCF=	.95341519E+00	WAFCF=	.58999571E+00	T2DS=	.51866820E+03
I.P. COMPRESSOR DESIGN	PRICF=	.41531922E+01	ETAI CF=	.97287007E+00	WAI CF=	.18585113E+00	T21DS=	.67476966E+03
H.P. COMPRESSOR DESIGN	PRCCF=	.34787092E+00	ETACCF=	.99186872E+00	WACCF=	.19857858E+00	T22DS=	.10024498E+04
COMBUSTOR DESIGN	W3CDS=	.16833805E+02	ETABCF=	.98750000E+00	CTCOCF=	.89568366E+00		
H.P. TURBINE DESIGN	CNHPCF=	.10695794E+01	TFHPCF=	.30748997E+01	ETHPCF=	.97888889E+00	DHHP CF=	.67280169E+00
I.P. TURBINE DESIGN	CNIPCF=	.10950379E+01	TFIPCF=	.38446973E+01	EIIPCF=	.97888889E+00	DHIP CF=	.14359988E+01
L.P. TURBINE DESIGN	CNLP CF=	.10820381E+01	TFLP CF=	.25251660E+01	ETLP CF=	.10394036E+01	DHLP CF=	.31564692E+01
DUCT DESIGN	W423DS=	.25046259E+04						
TURBINE/DUCT AREA DESIGN	A55=	.18480091E+01	AM55=	.65168846E+00	A25=	.64758021E+01	AM25=	.25239827E+00
AFTERBURNER ENTRANCE DESIGN AREA A6		11.620						
AFTERBURNER DESIGN	W6CDS=	.51899812E+04						
NOZZLE DESIGN	A8=	.46147586E+01	AM8=	.10000080E+01	A9=	.46825091E+01	AM9=	.11367079E+01
MAIN SHOCK OUTSIDE C-D NOZZLE			FG= 17454.88		FN= 17454.88		SFC=	.52412

# DIVERGED AFTER 1 LOOPS

MMON	.833300E+00	.100000E+03	.814300E+00	.100000E+03	.814300E+00	.100000E+03	.814300E+00	.100000E+03	.206000E+04	3
.100000E+03	.100000E+03	.100000E+03	.206000E+04	0.	.100000E+01	.100000E+01	.100000E+01	.100000E+01	.100000E+01	
.833300E+00	.100000E+03	.220000E+01	.839000E+00	.354000E+03	.300012E+01	.354000E+03	.300012E+01	.354000E+03	.300012E+01	
.214300E+00	.100000E+03	.343500E+01	.653000E+00	.108000E+03	.425319E+01	.108000E+03	.425319E+01	.108000E+03	.425319E+01	
.814300E+00	.100000E+03	.343500E+01	.853000E+00	.108000E+03	.347871E+01	.108000E+03	.347871E+01	.108000E+03	.347871E+01	
.286000E+04	0.	.125000E+04	.987500E+00	.108338E+02	.470000E+01	.108338E+02	.470000E+01	.108338E+02	.470000E+01	
.500000E+02	.200000E+01	.881000E+00	.307490E+01	.106958E+01	.978889E+00	.106958E+01	.978889E+00	.106958E+01	.978889E+00	
.120000E+03	.220000E+01	.881000E+00	.384470E+01	.109504E+01	.978889E+00	.109504E+01	.978889E+00	.109504E+01	.978889E+00	
.130000E+03	.230000E+01	.517000E+00	.252517E+01	.108204E+01	.103940E+01	.108204E+01	.103940E+01	.108204E+01	.103940E+01	
0.	0.	0.	0.	.290463E+04	.500000E+01	.290463E+04	.500000E+01	.290463E+04	.500000E+01	
0.	0.	0.	0.	.518998E+04	0.	.518998E+04	0.	.518998E+04	0.	
.184001E+01	.647580E+01	.116201E+02	.116201E+02	.461476E+01	.468251E+01	.461476E+01	.468251E+01	.461476E+01	.468251E+01	
.200300E+01	.651698E+00	0.	.950000E+00	0.	0.	0.	0.	0.	0.	
.518670E+03	.100000E+01	.123918E+03	.159103E+01	.518668E+03	.100000E+01	.518668E+03	.100000E+01	.518668E+03	.100000E+01	
.674770E+03	.220000E+01	.161385E+03	.160010E+01	.100245E+04	.755700E+01	.100245E+04	.755700E+01	.100245E+04	.755700E+01	
.146442E+04	.259583E+02	.359655E+03	.162430E+01	.286000E+04	.247383E+02	.286000E+04	.247383E+02	.286000E+04	.247383E+02	
.247750E+04	.119953E+02	.659137E+03	.183511E+01	.221324E+04	.687363E+01	.221324E+04	.687363E+01	.221324E+04	.687363E+01	
.179758E+04	.262345E+01	.460833E+03	.184603E+01	0.	0.	0.	0.	0.	0.	
.100000E+01	.220000E+01	.839000E+00	.600004E+03	.354000E+03	.600004E+03	.354000E+03	.600004E+03	.354000E+03	.600004E+03	
.100000E+01	.343500E+01	.853000E+00	.301279E+03	.108000E+03	.987500E+00	.108000E+03	.987500E+00	.108000E+03	.987500E+00	
.100000E+01	.343500E+01	.853000E+00	.100000E+03	.108000E+03	0.	.108000E+03	0.	.108000E+03	0.	
.200000E+01	.881000E+00	.403681E+01	.115453E+03	0.	0.	.115453E+03	0.	0.	0.	
.200000E+01	.881000E+00	.315920E+01	.782690E+02	0.	0.	.782690E+02	0.	0.	0.	
.230000E+01	.917000E+00	.542121E+01	.119985E+03	0.	0.	.119985E+03	0.	0.	0.	
.110541E+03	.235298E+01	.110541E+03	.235298E+01	.110541E+03	.235298E+01	.110541E+03	.235298E+01	.110541E+03	.235298E+01	
0.	.100000E+01	.833000E+00	.100000E+03	.814300E+00	.100000E+03	.814300E+00	.100000E+03	.814300E+00	.100000E+03	
.254122E+01	.500000E+02	.128000E+03	.130300E+03	0.	0.	.130300E+03	0.	0.	0.	
0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
.100000E+01	.354000E+03	.108000E+03	.108000E+03	.161385E+03	.160810E+01	.161385E+03	.160810E+01	.161385E+03	.160810E+01	
.241496E+03	.359655E+03	.674770E+03	.220000E+01	.674770E+03	.209000E+01	.674770E+03	.209000E+01	.674770E+03	.209000E+01	
.674770E+03	.220000E+01	.161385E+03	.160010E+01	0.	0.	0.	0.	0.	0.	
.674770E+03	.209000E+01	.161385E+03	.160362E+01	0.	0.	0.	0.	0.	0.	
0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
.246800E+03	0.	.246000E+03	0.	0.	.500000E+01	0.	.500000E+01	0.	.500000E+01	
0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
.179758E+04	.262345E+01	.460833E+03	.184603E+01	.674770E+03	.209000E+01	.674770E+03	.209000E+01	.674770E+03	.209000E+01	
.254122E+01	.110541E+03	.235298E+01	0.	.246000E+03	0.	.246000E+03	0.	.246000E+03	0.	
.104745E+04	.222336E+01	.254240E+03	.170844E+01	.104745E+04	.222336E+01	.104745E+04	.222336E+01	.104745E+04	.222336E+01	
.104745E+04	.222336E+01	.254240E+03	.170844E+01	0.	0.	.254240E+03	0.	.254240E+03	0.	
.356541E+03	0.	.356541E+03	.717858E+02	.213745E+01	.375176E+03	.213745E+01	.375176E+03	.213745E+01	.375176E+03	
.213757E+01	.375176E+03	.240000E+00	.103628E+04	.880629E+03	.100000E+01	.880629E+03	.100000E+01	.880629E+03	.100000E+01	
.880181E+03	.118643E+01	.14567E+04	.100000E+01	.157512E+04	.174549E+05	.157512E+04	.174549E+05	.157512E+04	.174549E+05	
0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
.174549E+05	0.	.254122E+01	.356541E+03	.717858E+02	.174549E+05	.717858E+02	.174549E+05	.717858E+02	.174549E+05	

SETUP FOR SLS AFTER TRAINING

**OUTPUT**

000.0 = 25

ALTP=

6

**T4= 2860.49**

**ETAR= 1.0000**

PCNF	.100000E+03
PCNI	.100000E+03
PCNC	.100000E+03
Y2	.100000E+03
Y3	.518680E+03
BL08	.16442E+04
WA3	.18083E+03
TFHP	.500000E+02
TFIP	.120000E+03
TFPL	.130000E+03
T24	.674770E+03
T25	.674770E+03
ETAF	.839000E+00
T6	.164745E+04
T7	.164745E+04
AM8	.100000E+01
AM28	
HPEXT	
VJM	.157275E+04
A25	.647500E+01

CNF	.100000E+01
CNI	.100000E+01
CNC	.100000E+01
F2	.100000E+01
P3	.259583E+02
BLH?	
0.	
WFB	.254122E+01
CNMP	.280000E+01
CNIP	.220000E+01
CNLP	.230000E+01
F24	.209000E+01
P25	.209000E+01
ETAI	.853000E+00
P6	.222358E+01
WFA	
0.	
V8	.144567E+04
V28	
0.	
WFT	.254122E+01
CVN02	
0.	
A6	.116201E+02

0.	ZF	.833300E+00
0.	ZI	.814300E+00
0.	ZC	.814300E+00
0.	T21	.674770E+03
0.	BLF	.0
0.	OLIF	.0
0.	MG4	.110541E+03
0.	DHTCHP	.403681E-01
0.	DHTCIP	.315528E-01
0.	DHTCLP	.542121E-01
0.	MAD	.246008E+03
0.	OPUC	.500800E-01
0.	ETAC	.853808E+00
0.	PS6	.213765E+01
0.	FARY	.717858E-02
0.	FS9	.160035E+01
0.	PS29	.0
0.	WGT	.356541E+03
0.	VJC	.0
0.	A8	.461476E+01

PRF	.220000E+01
PRI	.363500E+01
PRC	.343500E+01
T21	.220000E+01
BLI	0.
ALLP	0.
FAR4	.235298E-01
SHTC	.115453E+03
DHTI	.782698E+02
DHTF	.119985E+03
MFO	0.
ETAD	0.
ETATHP	.881000E+00
AM6	.239993E+08
ETAA	0.
AM9	.113625E+01
AM29	0.
VA	0.
FGM	.174286E+05
A9	.468251E+01

MAFC	.600000E+03
MAIC	.301279E+03
MACC	.100000E+03
T22	.100245E+04
BLC	0.
T4	.286000E+04
ETAB	.987500E+00
T45	.247750E+04
T5	.221324E+04
T55	.179750E+04
H24	.246000E+03
AM25	.252415E+00
ETAT IP	.881000E+00
V6	.375160E+03
DPAFT	0.
V9	.160405E+04
V29	0.
FRD	0.
FGP	.346999E+01
A28	0.

[illegible]

MAIN SHOCK OUTSIDE C-D NOZZLE

FG- 17432.11

**FN:- 17432.11**

**SFC = .5241**



# SLS FULL AFTERBURNING

NOZZLE DESIGN	A6=	.11176959E+02	AM8=	.10000000E+01	A9=	.1119071E+02	AM9=	.10347625E+01
OUTPUT	AM=	0.000	ALTP=	0.	T4=	2860.00	T7=	3700.00
PCNF	CMF		CHF		ZF		WAF	
.100000E+03	.100000E+01		.833300E+00		.220000E+01		WAF	ETAR= 1.0000E+01
PCNY	CHY		ZI		PRI		WAF	WAF
.100000E+03	.100000E+01		.614300E+00		.343500E+01		WAF	.354000E+03
PCNC	CNC		ZC		PRC		WAF	.108000E+03
.100000E+03	.100000E+01		.814300E+00		.343500E+01		WAF	.108000E+03
T2	P2		Z2		P2		WAF	.108000E+03
.518660E+03	.100000E+01		.674770E+03		.220000E+01		WAF	.755700E+01
T3	P3		BLF		BLI		WAF	BLDU
.146442E+04	.259543E+02		.9LIP		.8LIP		WAF	0.
BLDP	BLDP		0.		0.		WAF	0.
0.	0.		0.		0.		WAF	0.
MA3	MA3		MA4		MA4		WAF	0.
.168000E+03	.254122E+01		.110541E+03		.235290E-01		WAF	.247363E+02
TFHP	CHHP		CHICHP		DHTC		WAF	OPCON
.500000E+02	.200000E+01		.403661E-01		.115453E+03		WAF	.470000E-01
TFIP	CHIP		CHICIP		DHTI		WAF	P45
.120000E+03	.220000E+01		.315920E-01		.702630E+02		WAF	.119953E+02
TFEL	CHLP		CHICLP		DHTF		WAF	P5
.130000E+03	.230000E+01		.542321E-01		.115985E+03		WAF	.687363E+01
T24	P24		MAO		MFD		WAF	P55
.874770E+03	.209000E+01		.246000E+03		0.		WAF	.262349E+01
T25	P25		DPDUC		0.		WAF	FAR24
.874770E+03	.209000E+01		.500000E-01		0.		WAF	0.
ETAF	ETAI		ETAC		ETATHP		WAF	AM55
.839000E+00	.253000E+00		.853000E+00		.261000E+00		WAF	.652110E+00
T6	P6		PS6		AM6		WAF	ETATIP
.104745E+04	.222353E+01		.213765E+01		.239991E+00		WAF	.917000E+00
T7	P7		FAQ7		ETAA		WAF	MG5
.370000E+04	.183777E+02		.605055E-01		.875000E+00		WAF	.356541E+03
AM8	VB		PS9		AM9		WAF	PSR
.100000E+01	.256563E+04		.100000E+01		.103476E+01		WAF	.104245E+01
AM28	V28		PS29		AM29		WAF	PS28
0.	0.		0.		0.		WAF	0.
WEXT	WFT		MG7		VA		WAF	0.
0.	.214189E+02		.375419E+03		0.		WAF	0.
WJM	CYH02		VJD		0.		WAF	0.
.269251E+04	0.		0.		0.		WAF	0.
A25	AS		AM		0.		WAF	0.
.647536E+01	.114201E+02		.211740E+02		.111907E+02		WAF	0.
MAIN SHOCK OUTSIDE C-D NOZZLE	FM=	31417.20	FM=	31417.20	FM=	31417.20	SFC=	2.45433

**ETAR= 1.0000**

F	6372E+03
I	7156E+02
C	6722E+02
Z	9396E+01
QU	
	9504E+02
CON	
	7366E-01
S	
	4009E+01
	9020E+01
S	
	4206E+01
R24	
55	
	2441E+00
ATLP	
	9003E+00
6	
	7867E+03
8	
	5100E+00
20	
PASS	
	4037E+01
THNOZ	
	0000E+00
S	
	4001E+01
9	

**SFC= .76626**

**PUT PUT**

AME	.750
PCNF	.90000E+02
PCNI	.91566E+02
PCNC	.93275E+02
T2	.97094E+03
T3	.129033E+04
BL02	
WA3	
	.53997E+02
TFHP	.500143E+02
TFPI	.120217E+03
TFPL	.129702E+03
T24	.60344E+03
T25	.66344E+03
ETAF	.94300E+00
T6	.91670E+03
T7	.9C16.9E+03
AM0	.10000E+01
AM1	.66755E+01
MPXT	
VJM	.14683E+04
AS5	
WZLE	

	ALTP= 25000.
CNF	
.937461E+00	
CNI	
.968273E+00	
CNC	
.994281E+00	
P2	
.539056E+00	
F3	
.124320E+02	
SLHP	
0.	
MFB	
.105668E+01	
GNNP	
.193525E+01	
CNIP	
.215719E+01	
CNLP	
.224916E+01	
P26	
.102718E+01	
P25	
.102718E+01	
EYAI	
.860803E+00	
P6	
.100301E+01	
MFA	
0.	
V0	
.124264E+04	
V29	
0.	
MFT	
.105868E+01	
GVONCZ	
0.	
A6	
.116201E+02	
0.	
9640.79	

74= 2505.19	PRF	0.	2505.19	PRF	0.	2505.19	PRF
200920	PRI	0.	200920	PRI	0.	200920	PRI
332622	FNC	0.	332622	FNC	0.	332622	FNC
335911	421	0.	335911	421	0.	335911	421
106311	BLI	0.	106311	BLI	0.	106311	BLI
0.	BLLP	0.	0.	BLLP	0.	0.	BLLP
0.	FAR4	0.	0.	FAR4	0.	0.	FAR4
196066	DMTC	0.	196066	DMTC	0.	196066	DMTC
011300	DMT1	0.	011300	DMT1	0.	011300	DMT1
675700	DMTF	0.	675700	DMTF	0.	675700	DMTF
101622	WFD	0.	101622	WFD	0.	101622	WFD
0.	ETAD	0.	0.	ETAD	0.	0.	ETAD
0.	ETATHR	0.	0.	ETATHR	0.	0.	ETATHR
600400	AMG	0.	600400	AMG	0.	600400	AMG
239333	ETAA	0.	239333	ETAA	0.	239333	ETAA
0.	AM9	0.	0.	AM9	0.	0.	AM9
113655	AM29	0.	113655	AM29	0.	113655	AM29
0.	VA	0.	0.	VA	0.	0.	VA
762422	FGM	0.	762422	FGM	0.	762422	FGM
050433	A9	0.	050433	A9	0.	050433	A9
0.	460253	0.	0.	460253	0.	0.	460253
FN= 522			FN= 522			FN= 522	

EE+01	WAFG	562351E+03
EE+01	WAIC	289366E+03
EE+01	WACC	936508E+02
EE+01	1-2	087570E+03
EE+01	BLC	0.
EE+01	T4	259519E+04
EE+01	ETAB	987500E+00
EE+03	T45	236052E+04
EE+02	T5	132572E+04
EE+03	T55	156240E+04
EE+03	W024	132288E+03
EE+00	AM25	261722E+00
EE+00	ETATIP	379322E+00
EE+00	V6	358319E+03
EE+00	CPAFI	0.
EE+01	V0	149032E+04
EE+01	V29	0.
EE+03	FAD	441460E+04
EE+04	FG3	133648E+04
EE+01	A26	0.
EE+00		6.20

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ETAR= 1.0000
WAF
.186295E+03
WAL
.540067E+02
WAC
.539972E+02
PEZ
.360276E+01
BLDU
0.
.115277E+02
DPCOM
.474593E-01
P45
.550674E+01
P5
.320299E+01
P55
.125485E+01
FAR24
0.
AM55
.614054E+00
ETATLP
.915050E+00
WG6
.187344E+03
PS8
.575798E+00
PS28
0.
BYPASS
.244948E+01
CVHNOZ
.980000E+00
A55
.184001E+01
A29
0.
SFC= .72926

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WAZN SHOCK CUTSIDE C-D NO 271E

FN= 5226.20

SFC= .72926

## SUBSIC FQWER HOOK

[illegible]

SEASONING POWER HOOK

[illegible]

FAIRY SHOCK OUTSIDE C-0 NO 7713



AMZLE DESIGN	A0=	.11057560E+02	AMB=	.10000000E+01	A9=	.13339090E+02	T7=	3700.00	WAF	ETAR=	.9445
PCNF	1.6U0	ALTP=	50000.	ZF	2650.40	WAF	WAF	WAF	WAF	WAF	WAF
.685761E+02	.795330E+00	.645492E+00	.163708E+01	.477555E+03	.157100E+03	.397357E+02	.255610E+03	.397357E+02	.397357E+02	.397357E+02	.397357E+02
PCNI	.926054E+02	.672474E+00	.622463E+00	.295360E+01	.255610E+03	.397357E+02	.255610E+03	.397357E+02	.397357E+02	.397357E+02	.397357E+02
PCMC	.180000E+03	.970105E+00	.772439E+00	.318060E+01	.956252E+02	.397357E+02	.956252E+02	.397357E+02	.397357E+02	.397357E+02	.397357E+02
T2	.643321E+03	.420975E+00	.760192E+03	.101658E+01	.106518E+04	.397357E+02	.106518E+04	.397357E+02	.397357E+02	.397357E+02	.397357E+02
T3	.152572E+04	.955024E+01	.0.	.0.	.0.	.397357E+02	.0.	.397357E+02	.397357E+02	.397357E+02	.397357E+02
BL0B	.0.	BLHP	.0.	BLIP	.0.	.397357E+02	.0.	.397357E+02	.397357E+02	.397357E+02	.397357E+02
WA3	.397237E+02	.892199E+00	.406199E+02	.225608E-01	.987500E+00	.397237E+02	.987500E+00	.397237E+02	.397237E+02	.397237E+02	.397237E+02
TFHP	.500093E+02	.199906E+01	.405606E-01	.116074E+03	.247514E+04	.397237E+02	.247514E+04	.397237E+02	.397237E+02	.397237E+02	.397237E+02
TFHP	.120519E+03	.203825E+01	.296827E-01	.734843E+02	.222672E+04	.397237E+02	.222672E+04	.397237E+02	.397237E+02	.397237E+02	.397237E+02
TFELP	.126723E+03	.233108E+01	.489704E-01	.109147E+03	.184906E+04	.397237E+02	.184906E+04	.397237E+02	.397237E+02	.397237E+02	.397237E+02
T24	.760192E+03	.960880E+00	.117364E+03	.0.	.117364E+03	.397237E+02	.117364E+03	.397237E+02	.397237E+02	.397237E+02	.397237E+02
T25	.760192E+03	.960880E+00	.547941E-01	.0.	.280798E+00	.397237E+02	.280798E+00	.397237E+02	.397237E+02	.397237E+02	.397237E+02
ETAF	.827705E+00	.278268E+00	.355916E+00	.880399E+00	.876926E+00	.397237E+02	.876926E+00	.397237E+02	.397237E+02	.397237E+02	.397237E+02
T6	.06105E+03	.992617E+00	.250412E+00	.239714E+00	.377156E+03	.397237E+02	.377156E+03	.397237E+02	.397237E+02	.397237E+02	.397237E+02
T7	.370000E+04	.857464E+01	.602501E+01	.ETAA	.858000E+00	.397237E+02	.858000E+00	.397237E+02	.397237E+02	.397237E+02	.397237E+02
W8	.200000E+01	.266588E+04	.114456E+00	.197244E+01	.455923E+04	.397237E+02	.455923E+04	.397237E+02	.397237E+02	.397237E+02	.397237E+02
AM20	.0.	W2B	.0.	PS29	.0.	.397237E+02	.0.	.397237E+02	.397237E+02	.397237E+02	.397237E+02
HPERT	.0.	WFT	.0.	W6Y	.0.	.397237E+02	.0.	.397237E+02	.397237E+02	.397237E+02	.397237E+02
WJW	.445885E+04	.947844E+01	.166571E+03	.174329E+04	.851214E+04	.397237E+02	.851214E+04	.397237E+02	.397237E+02	.397237E+02	.397237E+02
A25	.647508E+01	.116281E+02	.110576E+02	.193391E+02	.193391E+02	.397237E+02	.193391E+02	.397237E+02	.397237E+02	.397237E+02	.397237E+02
NOZZLE	FC=	23139.10	FN=	14610.84	SFC=	2.33239					

## REFERENCES

1. John S. McKinney, Simulation of Turbofan Engine, Parts I & II, AFAPL-TR-67-125, Air Force Aero Propulsion Laboratory, Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, November 1967
2. Laurence H. Fishbach and Robert W. Koenig, Geneng II - A Program for Calculating Design and Off-Design Performance of Two- and Three-Spool Turbofans with as Many as Three Nozzles, NASA TN D-6553, NASA-Lewis Research Center, Cleveland, Ohio, February 1972